

OPERATING INSTRUCTIONS

ALLIS-CHALMERS FUEL CELL POWER SYSTEM

(SYSTEM NUMBER 5R)

Prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

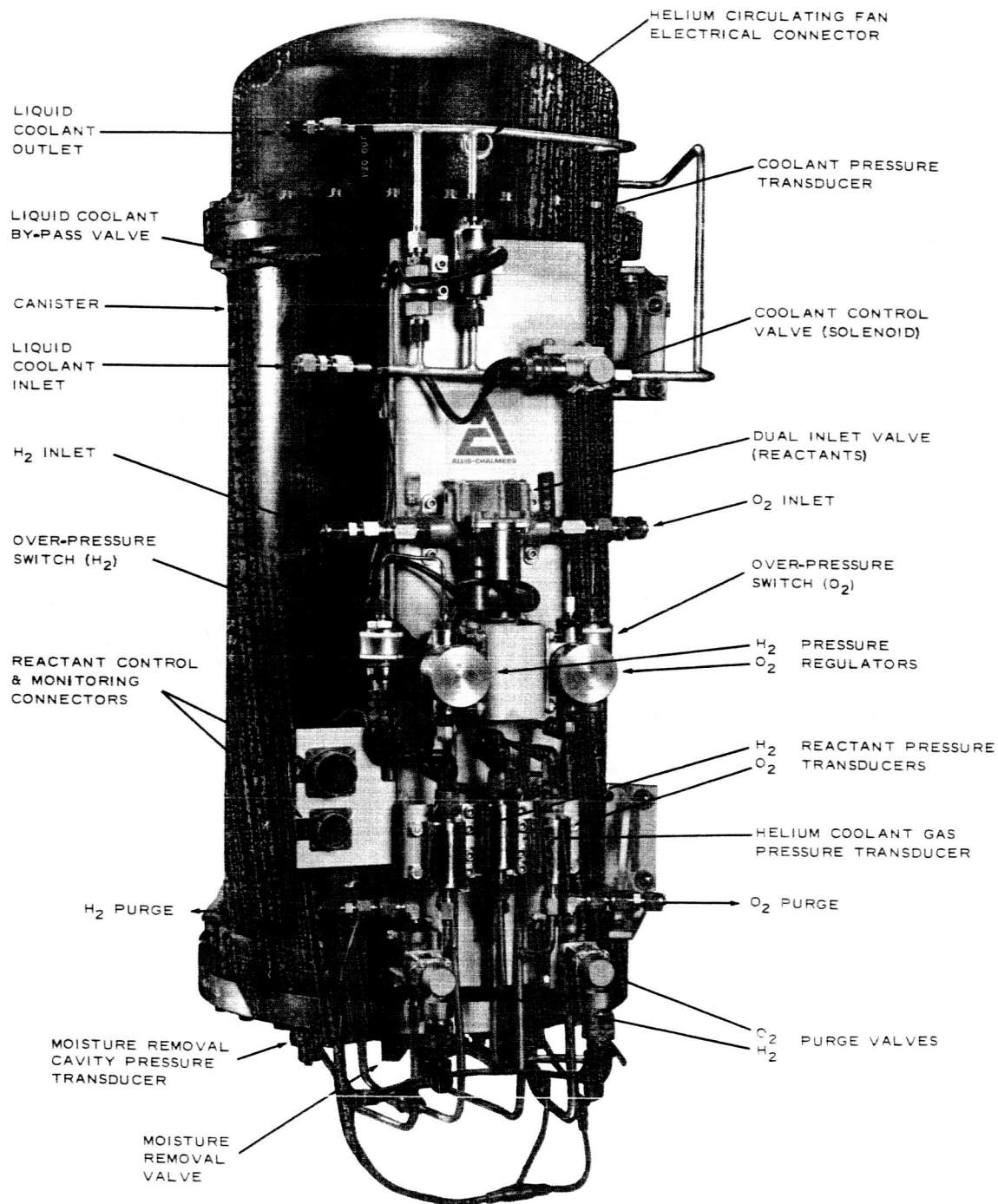
MARSHALL SPACE FLIGHT CENTER

Huntsville, Alabama

ALLIS-CHALMERS  
RESEARCH DIVISION

NASA Fuel Cell Systems Program  
Milwaukee, Wisconsin

June 6, 1966



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## SECTION I

### SYSTEM DESCRIPTION

#### 1.1 GENERAL DESCRIPTION OF OPEN LOOP FUEL CELL SYSTEM

The fuel cell system is an electro-chemical device in which the controlled temperature reaction of two input gases, hydrogen and oxygen, in the presence of a suitable electrolyte and electrodes, produces a direct current electrical output. The by-products of the reaction are heat and water which must be removed from the fuel cell. As the gases are consumed in the reaction, minute amounts of gas impurities (inerts) present in the reactant gases accumulate in the cell cavities. To maintain proper system performance, it is necessary to remove these impurities by periodically purging the cell cavities.

The basic operating voltage of the system is fixed by the number of cells, and the combination of electrical connections between the cells. Within the design limits, the load current will follow the demand, provided adequate amounts of reactant gases are supplied to the fuel cell.

This fuel cell is designated as an "asbestos capillary matrix" type in which the potassium hydroxide (KOH) electrolyte is held in a porous asbestos matrix without any circulation.

The by-product water is withdrawn from the fuel cell as vapor. It is produced at the hydrogen electrode, and diffuses into the moisture removal cavity through a KOH impregnated asbestos capillary matrix. The rate of water removal is controlled by maintaining the moisture removal cavity at a pressure equivalent to a predetermined electrolyte concentration. The quantity of water produced is proportional to the current produced by the fuel cell.

In the open loop system the moisture vapor, controlled by the moisture removal valve, is vented to a vacuum source.

Approximately one-third of the by-product heat is removed from the stack by evaporation of the product water. This helps promote good cell and stack temperature control. The majority of the by-product heat is removed from the fuel cell stack by recirculating helium gas over the finned edges of the cell plates. The heated helium gas then passes through a gas-to-liquid heat exchanger where the excess heat is rejected to the liquid coolant after which it is recirculated. The liquid coolant is cooled externally of the fuel cell system. The helium gas is recirculated constantly at a fixed rate by fans and the amount of heat removed from the fuel cell is controlled by controlling the flow of the liquid coolant through the heat exchanger.

## 1.2

### PERFORMANCE CAPABILITY GOALS

The performance capability goals of this fuel cell system are tabulated below:

Nominal Power Rating	2000 watts
Overload	2700 watts (load profile)
Voltage Regulation	27 to 32.5 volts (22 volts min.)
Minimum Voltage (overload)	22 volts
Cell Operating Temperature	195 $\pm$ 5°F
Maximum Reactant Consumption at 2000 watt load	Hydrogen - 0.22 pounds per hour Oxygen - 1.75 pounds per hour
Reactant Inlet Pressure	175-400 psia
Reactant Inlet Temperature	0 to 150°F
Reactant Purity	0.999 min. 0.99998 hydrogen 0.99995 oxygen Desired
Thermal efficiency (min).	55%
Endurance	1250 Qualification

Liquid Coolant	60% methyl alcohol, 40% water by weight, plus corrosion inhibitors
Liquid Coolant Supply	120 pounds per hour at 60°F
Liquid Coolant Pressure Drop (max.)	15.0 psid

### 1.3 SYSTEM DIAGRAMS

- 1.3.1 Figure 1-1 illustrates schematically the complete functional fuel cell open-loop system. The basic input requirements to the open-loop cell assembly are:

Hydrogen Gas  
Oxygen Gas  
Auxiliary Power Supply for Start-up  
Liquid Coolant  
Vacuum  
Helium Gas

The products removed from the fuel cell assembly are:

Direct Current Electricity  
Water  
Heat

The specifications for each of the input and output items are detailed in Section 2.1 Fuel Cell Assembly and Section III - INTEGRATED SYSTEM OPERATIONS.

- 1.3.2 Figure 1-2A, 1-2B and 1-2C (Drawing No. 49-500-125-404) is the Fuel Cell System Schematic illustrating SYSTEM OPERATIONS. A description of the subsystems is found in Section 4.

# SCHEMATIC - FUEL CELL POWER SYSTEM

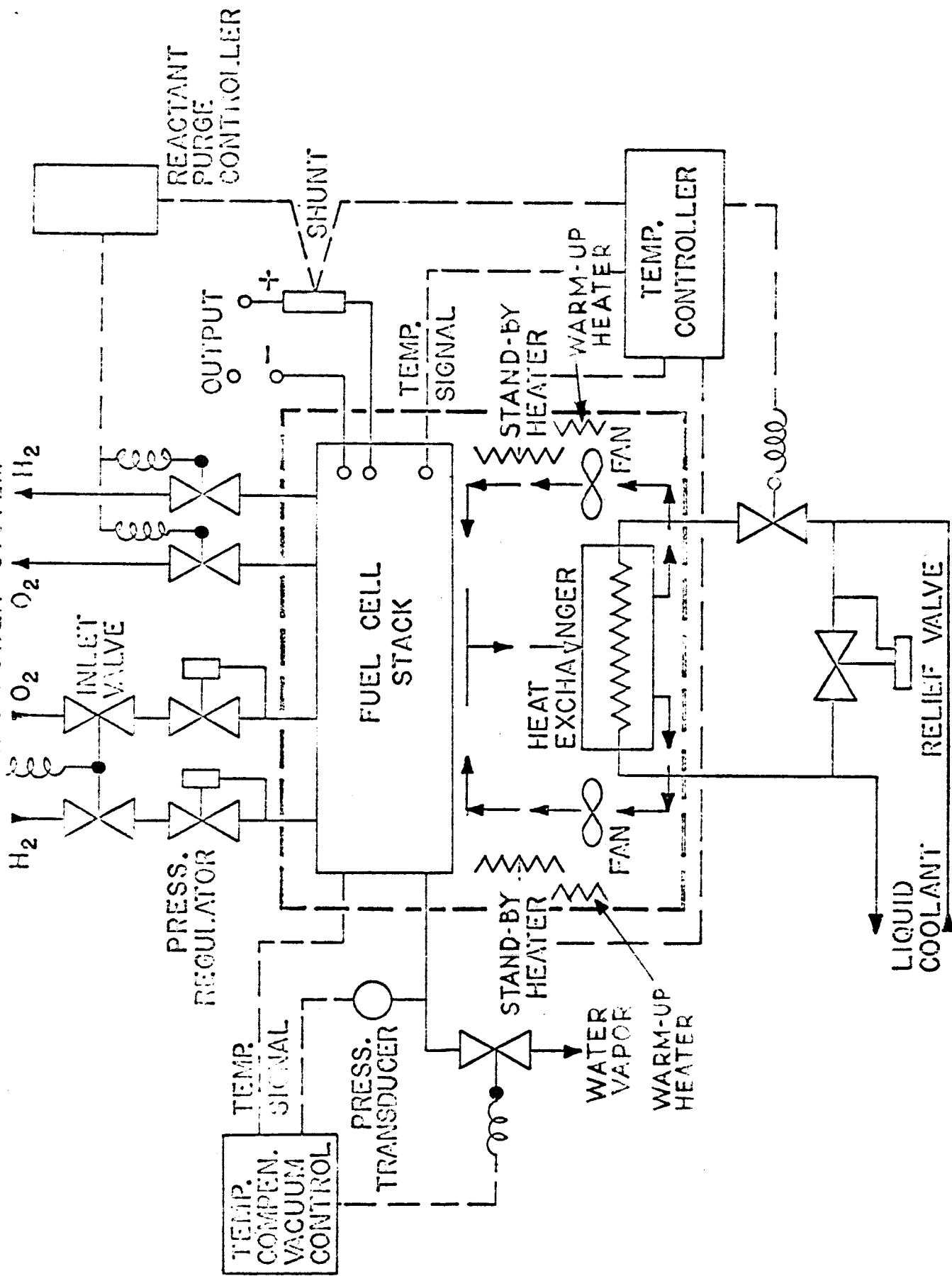
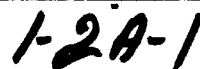


FIG. 1-1





- |   |  |   |  |
|---|--|---|--|
| LAMM DENTAL, PROPERTY OF<br>LAMM CHAMBERS MFG. CO.<br>534. M. 0000  |  | NAME<br>INTERFACE SCHEMATIC<br>915 MSP (C.A.)         |  |
| ORDER # (Customer Order #)<br>_____   |  | DATE<br>_____   |  |
| PLACE DEL. <input type="checkbox"/><br>PLACE DEL. <input type="checkbox"/><br>PLACE DEL. <input type="checkbox"/><br>UNUSUAL <input type="checkbox"/> |  | <input checked="" type="checkbox"/> SURFACE<br>FINISH |  |
| FIGURE 1-2A   |  | FIGURE 1-2A   |  |
| SCALE<br>1" = 1/16"   |  | PART NO.<br>49-500-25-404                             |  |

1-2A-2



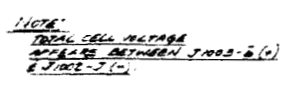


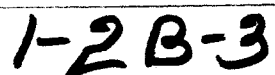
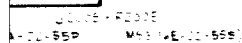
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SIGNAL TO MAIN -ENTER

S/G. TO OPEN MOISTURE REMOVAL V

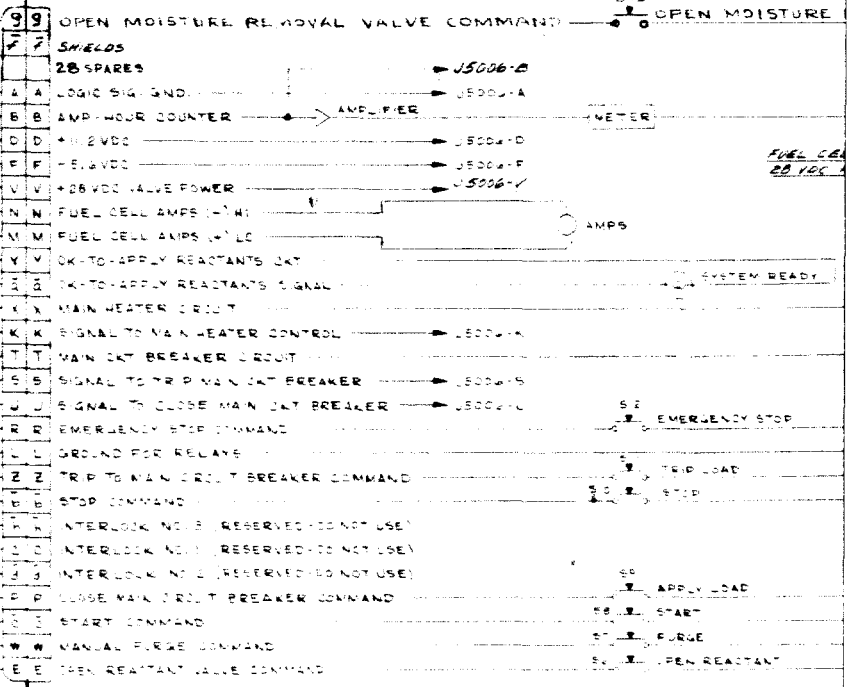
FUEL CELL ASSEMBLY





REF. CABLE ASSY  
DWG 49-100-500-501

TO SHIELDS



M22 WE-10-55PZ P5003 J5006 (M53-4E-12-5552)

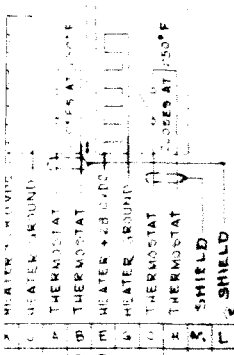
REF. CABLE ASSY  
DWG 49-100-588-500

PL011A-1 A +28.00 FUEL CELL PWR  
PL001B-1 B FUEL CELL GROUND  
PL001C-1 C +28.00 AUX HEATER PWR  
PL001D-1 D +28.00 AUX VALVE PWR  
PL001E-1 E HEATER GROUND  
(M53-4E-4-5P) P5004 J5004 (M53-4E-14-5P)

28.00VDC POWER TRIP  
TERMINALS

UNIT BOX TERMINAL A  
NASA LOAD CONNECTOR

10 WATT MAIN HEATERS



M53-00A-28-00P

J5003

P5003 (M53-4E-12-5552)

SHUNT  
CABINET

REF DWG. 49-200-432-501

J5004 P5004 (M53-4E-14-5P)

4 SPARES

TERMINAL

TERMINAL

TERMINAL

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ELECTRICAL

MONITORING

CONTROL

SYSTEM

TERMINAL

TERMINAL

TERMINAL

TERMINAL

TERMINAL

TERMINAL

TERMINAL

LOGIC SIGNAL GROUND

AMP-HOUR COUNTER

+11.2VDC

-5.0VDC

+28.0VDC VALVE PWR

SIGNAL TO TRIP MAIN CKT BRKR

SIGNAL TO CLOSE MAIN CKT BRKR

SIGNAL TO MAIN HEATER CONTROL

SIGNAL TO MAIN HEATER CONTROL

SIGNAL TO MAIN HEATER CONTROL

SIGNAL TO MAIN HEATER CONTROL

1-20-74





GROUND TO THERMISTOR

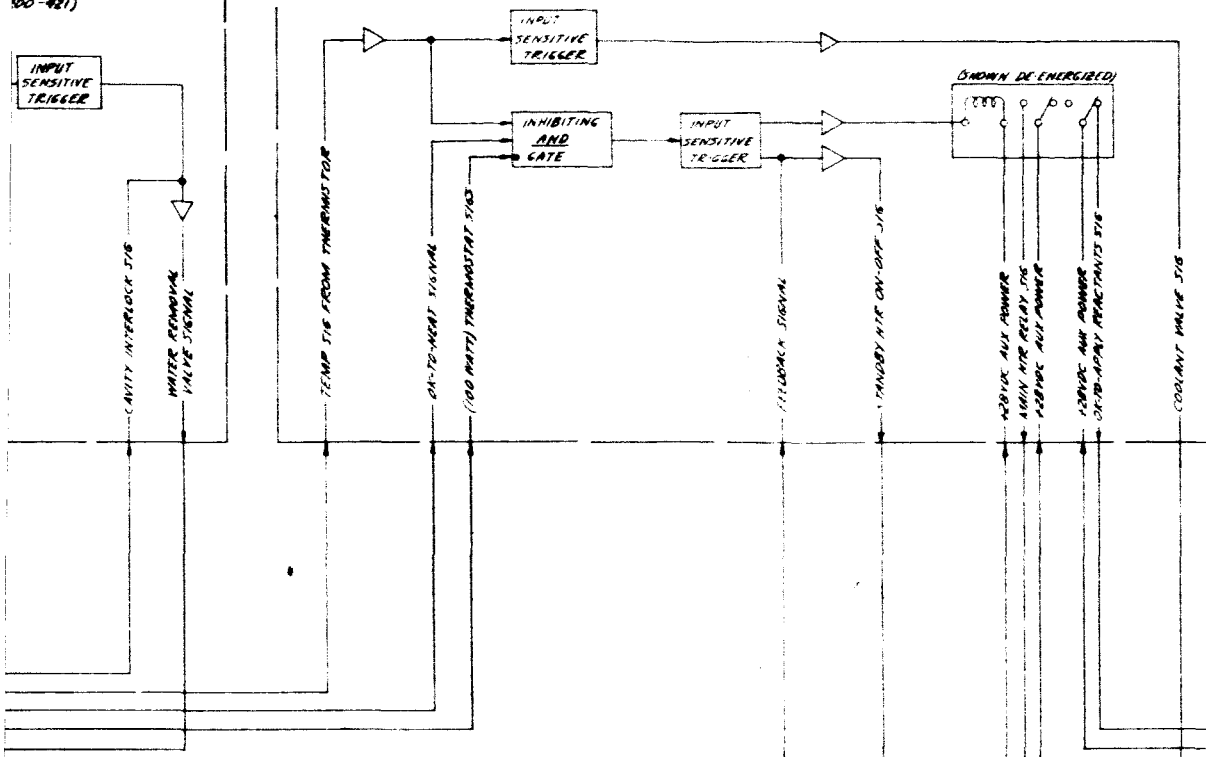
1-2 C-1



# CIVITY CONTROL (100-481)

## TEMPERATURE CONTROL

(REF DWG # 43-400-232)



TO GROUND  
TO 112 V.L  
TO GROUND  
TO 56 V.D

TO GROUND  
TO 56 V.D

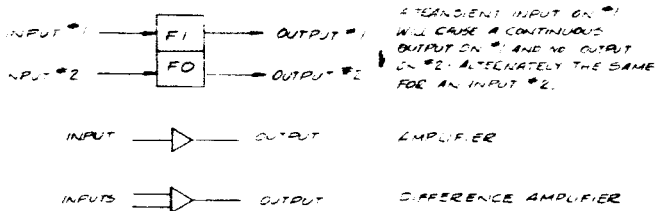
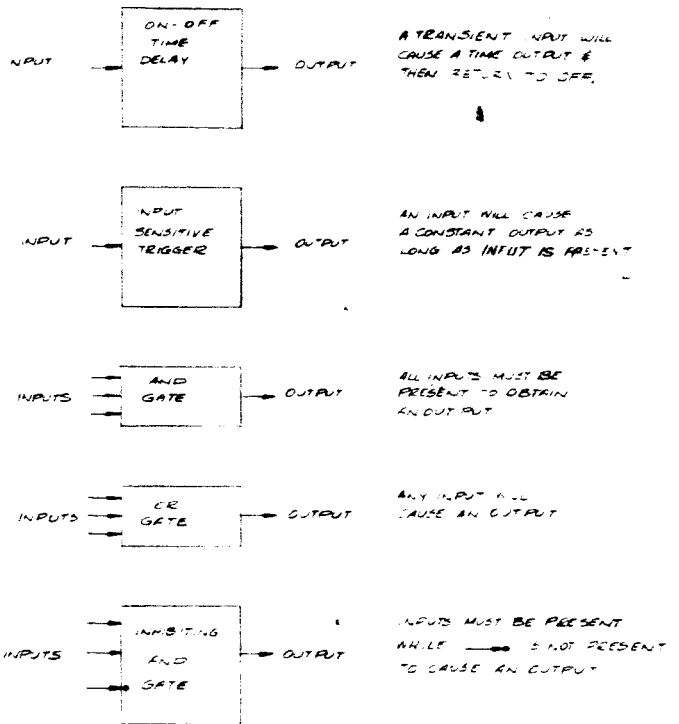
120VDC TO COOLANT VALVE  
5V TO COOLANT VALVE

J2004

1200-2



# SYMBOLS USED



R OK TO APPLY REACTANTS READY SIGNAL  
 I OK TO APPLY REACTANTS CIRCUIT

X MAIN HEATER CIRCUIT  
 K SIGNAL TO MAIN HEATER CONTROL  
 V +20 VDC AUX VALVE POWER (READOUT)  
 M MANUAL PURGE COMMAND  
 R EMERGENCY STOP COMMAND  
 B AMP HOUR READOUT

U CLOSE MAIN CIRCUIT BREAKER SIGNAL  
 T CLOSE MAIN CIRCUIT BREAKER CIRCUIT  
 S TOP MAIN CIRCUIT BREAKER CIRCUIT  
 C OPEN REACTANT VALVE COMMAND  
 P CLOSE MAIN CRY BEAKER COMMAND  
 V EMERGENCY STOP COMMAND  
 E TOP MAIN CRY BEAKER COMMAND  
 I EMERGENCY STOP COMMAND  
 I STOP COMMAND  
 E START COMMAND

M FUEL CELL SHUNT (H) LO (AMPS READOUT)  
 N FUEL CELL SHUNT (H) HI (AMPS READOUT)  
 A J086 SIGNAL GROUND  
 D +11.2 VDC (READOUT)  
 E GROUND FOR OPEN REACTANT VALVE COMMAND CIRCUIT  
 F -5.6 VDC (READOUT)

SPARES

C GROUND FOR EMERGENCY STOP CIRCUIT  
 A GROUND FOR EMERGENCY STOP CIRCUIT  
 F SPARE  
 F SHIELDS

J2005

01 3-22-66  
 011 3-22-66 REL. ED.  
 0153418

REDUCED REPRODUCTION  
 OF 49-500-125-404  
 SHEET 3 OF 3

CONFIDENTIAL PROPERTY OF ALLAN-CHAMBERS MFG. CO.		NAME: <b>LOGICAL SYSTEM SR</b>	
DATE: <b>3-22-66</b>		VERSION: <b>1.0</b>	
DESIGNED BY: <b>W. H. H.</b>		CHECKED BY: <b>W. H. H.</b>	
DRAWN BY: <b>W. H. H.</b>		DATE: <b>3-22-66</b>	
SCALE: <b>1:1 (NPG)</b>		SHEET: <b>3 OF 3</b>	
FIG. 1-2C		1-200-11	

## SECTION II

### INSTALLATION INSTRUCTIONS

#### 2.1 UNPACKING, INSPECTION AND ASSEMBLY

##### 2.1.1 Shipping Containers

The fuel cell module including the reactant control valve assembly will be shipped within a wooden container as shown in Figure 2-1 (drawing No. 49-400-230-401). The fuel cell module has been pressurized with helium to a pressure of approximately 5 psig and appropriate care should be exercised in handling the unit. The shipping container should be placed in the proper vertical attitude to facilitate cell removal.

The fuel cell module should be supported from the eyelets provided in the upper clamp-ring shim, and then it may be unbolted from the shipping container and moved to the test area.

The other items supplied by Allis-Chalmers for the assembly of Fuel Cell System #5R will be boxed in wood and/or cardboard containers suitably protected against moisture and dirt.

##### 2.1.2 Inspection Prior to Assembly

All major components have been cleaned and packaged at the Allis-Chalmers facility, however, prior to assembly all components should be examined for cleanliness.

##### 2.1.3 Assembly of the Fuel Cell System

Fuel Cell System #5R as delivered consists of five major subassemblies to which interconnections must be made in the field. Refer to Figure 2-2A (Drawing No. 49-400-230-405 and Figure 2-2B (Drawing No. 49-200-431-504.

These figures physically locate each point of interconnection, identify the interconnecting cable, show the connector designation and specify the customer tie-in points. The major subassemblies are:

- Item 2 Fuel Cell Module whose Major Subassemblies are Stack the Fuel Cell and the Reactant Control Subsystem.
- Item 4 Electrical Monitoring and Control Subsystem.
- Item 5 Current Shunt Cabinet
- Item 6 Allis-Chalmers Auxiliary Relay Box (For reference only)
- Item 7 Allis-Chalmers Control Panel (For reference only)

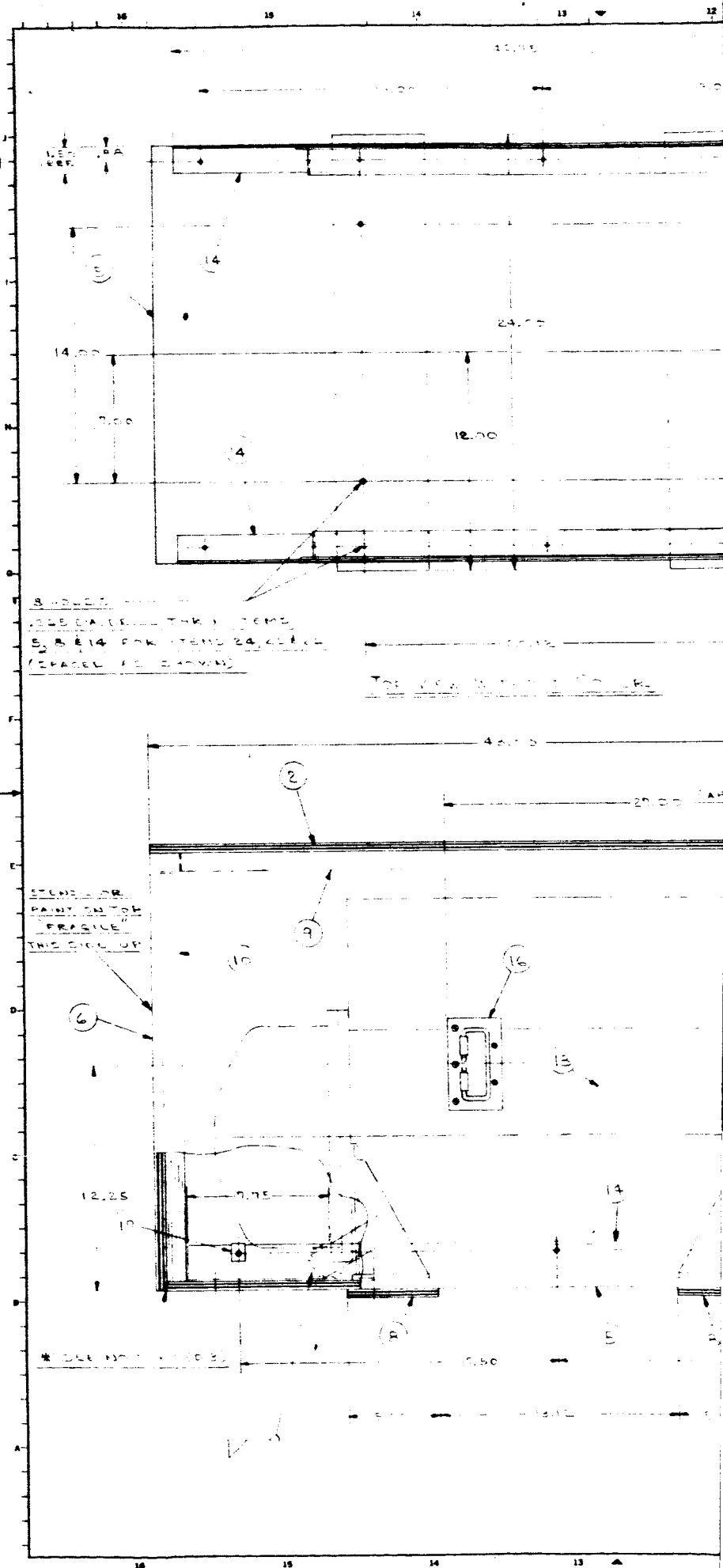
Allis-Chalmers has furnished interconnecting electrical cables between the fuel cell module and the EMCS, the module and the shunt cabinet and a "Y" cable from the EMCS to the shunt cabinet and from the EMCS to the fuel cell module.

In addition, Allis-Chalmers has furnished all mating connectors for the remaining electrical connectors and all pneumatic/hydraulic fittings for use on the A-C furnished equipments.

#### 2.1.4 Equipment Furnished by the Marshall Space Flight Center

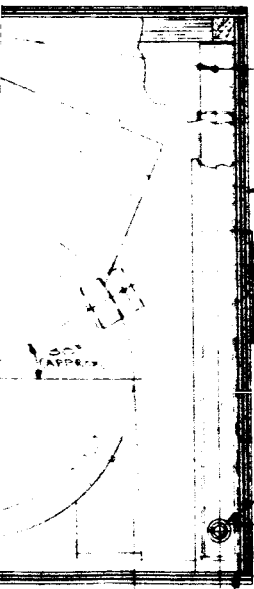
MSFC will furnish the following major items for the test setup of System #5R.

- A. All operational controls including switches and indicators. Figure 2-3 contains a listing of the recommended controls and indicators for operation of the test setup. Figure 2-4A (Dwg. No. 49-200-550-501),  
Figure 2-4B (Dwg. No. 49-400-318-401),  
Figure 2-4C (Dwg. No. 49-400-314-401),  
Figure 2-5 (Dwg. No. 49-200-523-501),  
Figure 2-6 (Dwg. No. 49-300-506-501),  
Figure 2-7 (Dwg. No. 49-300-520-501). &  
Figure 3-4 (Dwg. No. 49-400-349-402)  
contain additional information on the method of control and indicator hookup.
- B. All tubing, valves, gases, and flowmeters required to bring the reactant gases, helium gas and the liquid coolant to the fuel cell interface and to handle the by-product water. Figure 1-2A (Dwg. No. 49-500-125-404).





ITEM	DESCRIPTION	MATERIAL	PART NO	QTY
1	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-501	1
2	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-502	1
3	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-503	1
4	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-504	1
5	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-505	1
6	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-506	1
7	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-507	1
8	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-508	1
9	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-509	1
10	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-510	1
11	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-511	1
12	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-512	1
13	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-513	1
14	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-514	1
15	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-515	1
16	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-516	1
17	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-517	1
18	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-518	1
19	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-519	1
20	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-520	1
21	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-521	1
22	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-522	1
23	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-523	1
24	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-524	1
25	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-525	1
26	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-526	1
27	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-527	1
28	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-528	1
29	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-529	1
30	20" X 40" X 1/2" ALUM. PLATE	ALUMINUM	49-400-230-530	1



NOTES:

1. GLUE & NAIL ALL WOOD JOINTS, EXCEPT THOSE INDICATED THOSE ITEM 8 TO ITEM 5 & SKIDS TO FRAME.
2. PAINT BOX WITH GRAY OR BLUE ENAMEL.
3. USE ANGLE FRAMES AS DRILLING TEMPLATE FOR HOLES IN PANELS.

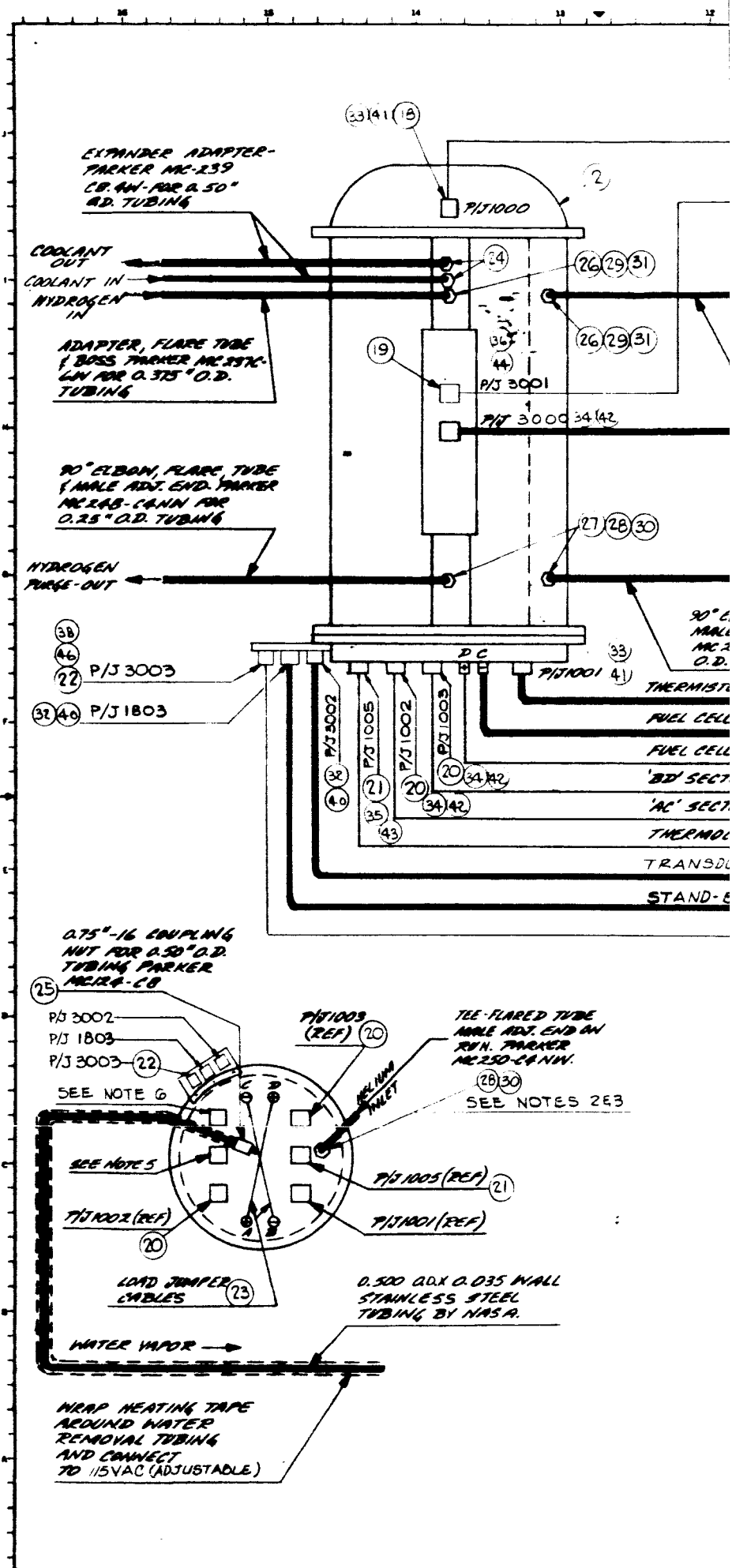
1. 1/2" DIA. HOLE 1/2" DEEP IN ITEM 12 FROM ITEM 15 CLEARANCE 1/4" PLACE

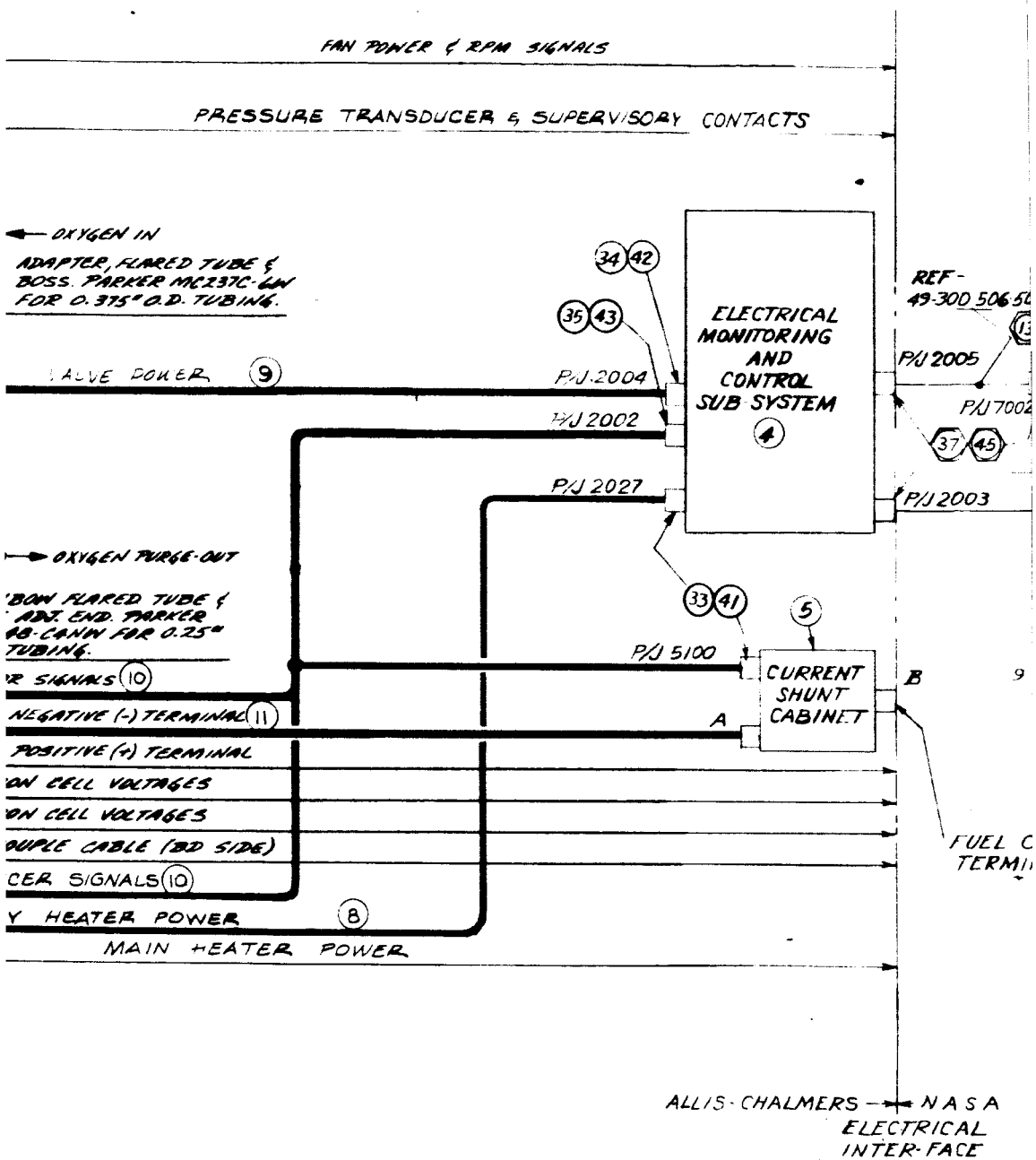
SEE NOTE HT (P. 3)

CONFIDENTIAL - PROPERTY OF ALLIS-CHALMERS MFG. CO. WEST ALLES WORKS		NAME <u>SHIPPING CONTAINER</u> <u>FUEL CELL</u>	
UNLESS OTHERWISE SPECIFIED		MATERIAL	
1. PLACE DEC. 2. PLACE DEC. 3. PLACE DEC. ANGULAR	✓ MACHINED SURFACE TEXTURE	R WT F	
20" X 40" X 1/2" ALUM. PLATE	SIMILAR TO	PART NO 49-400-230-501	
DR. 1 CH. 10 AP. 10	SCALE 3"=12"	SHEET	

49-400-230-501 01

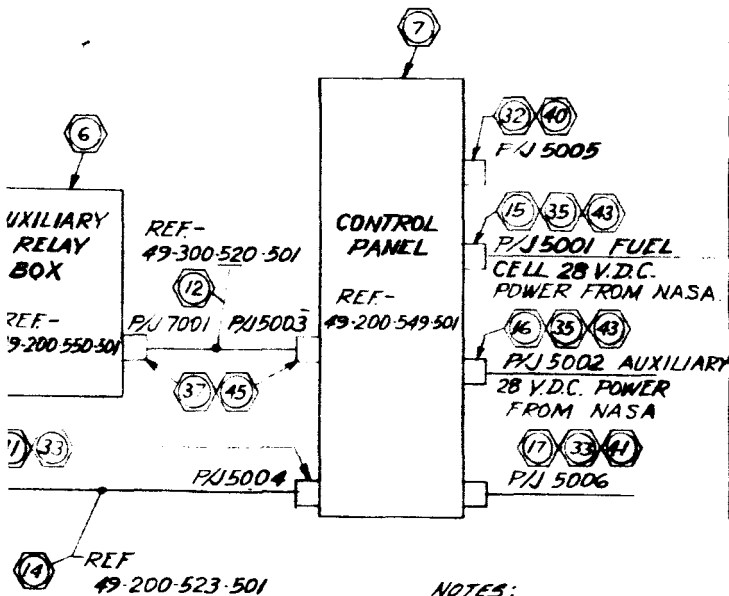
FIGURE 2-1-3







31 3-1-66 RCM  
ELECTRICAL INTERFACE  
- ANGELI TO INDICATE THAT  
THE ALLIS CHALMERS AUX.  
RELAY BOX, CONTROL PANEL  
AND ASSOCIATED CABLES,  
ALL INDICATED THUS  
ANY NOT TO BE SHIPPED.



#### NOTES:

1. WRAP FUEL CELL INSULATION WITH ALUMINUM FOL TYPE BEFORE PERFORMING ANY VACUUM CHAMBER TESTS.
2. PRIOR TO SHIPPING,  $H_2$ ,  $O_2$  &  $H_2O$  CAVITIES (CANISTER SYSTEM) TO BE PRESSURIZED TO 5 PSI WITH HELIUM GAS AFTER EVACUATION OR PURGING OTHER GASES FROM SYSTEM. (\*)
3. AFTER CANISTER COOLANT SYSTEM HAS BEEN PRESSURIZED, CRIMP (SEAL) 0.25" O.D. TUBING HELIUM FILL LINE LOCATED AT BOTTOM MOUNTING PLATE.
4. HEAVY CABLE LINES ARE A-C SUPPLIED ITEMS. LIGHT CABLE LINES ARE NASA OR TEST LAB SUPPLIED.
5. HAS CONNECTOR J1004, THERMOELECTRIC CABLE ON ORIGINAL SYSTEM NO. 5 DEFINITION. (FOR INFORMATION ONLY)
6. HAS CONNECTOR J1006, MAIN HEATER CABLE ON ORIGINAL SYSTEM NO. 5 DEFINITION. (FOR INFORMATION ONLY)
7. TUBING ADAPTERS SHOWN ARE THE MATING PARTS REQUIRED TO CONNECT TO THE FUEL CELL AND MRS.

& FOR INTERFACE SCHEMATIC SEE DRG. 49-500-125-404

(\*) CAUTION - PRESSURE SHOULD BE INCREASED OR DECREASED GRADUALLY TO INSURE MAX. PRESSURE DIFFER. BETWEEN  $H_2$  AND  $O_2$  PRESSURE GAUGE IS LESS THAN 2 PSI.

#### PARTS LIST 49-200-431-504

CONFIDENTIAL - PROPERTY OF ALLIS-CHALMERS MFG. CO. 3341 - RESEARCH		NAME FUEL CELL SYSTEM 5R	
UNLESS OTHERWISE SPECIFIED:		MATERIAL	
1. PLACE DEC.	✓ MACHINED SURFACE TEXTURE	FIGURE 2-2A	
2. PLACE DEC.			
3. PLACE DEC.			
ANGULAR			
DR. 1-24-66	SIMILAR TO		
CH. SPANNA 2/24	SCALE	PART NO.	
AP. 2/24/66		49-400-260-405	

0-260-405 02

2-2A-2

REQ. ITEM	DESCRIPTION	PART NUMBER
1	Fuel Cell System #5R PIX	49-400-260-405
1	Fuel Cell Assembly	49-500-105-510
1	Interface Schematic	49-500-125-404
1	Electrical Monitoring and Control Subsystem (EMCS)	49-200-585-502
1	Current Shunt Cabinet	49-200-432-501
1	Auxiliary Relay Box	49-200-550-501
1	Control Panel	49-200-549-501
1	Cable, Stand-by Heater Power	49-200-522-501
1	Cable, ROSS to EMCS Valves	49-300-513-501
1	Cable, EMCS to Thermistors, Shunt Cab. and Transducers	49-100-706-501
1	Cable, -28V D-C Stack	49-100-588-502
1	Cable, Aux. Relay Box to Control Panel	49-300-520-501
1	Cable, EMCS to Aux. Relay Box	49-300-506-501
1	Cable, EMCS to Control Panel	49-200-523-501
1	Electrical Connector (P 5001)	MS316E-18-3S(SR)
1	Electrical Connector (P5002)	MS316E-18-3Sx (SR)
1	Electrical Connector (P5006)	MS316E-14-19 SW (SR)
1	Electrical Connector (P1000) Bendix	PT06A-14-19S(SR)

DO NOT SHIP - USE FOR  
TEST AT ALLIS-CHALMERS ONLY

DO NOT SHIP - USE FOR  
TEST AT ALLIS-  
CHALMERS ONLY

1	19	Electrical Connector (P3001)	MS3106E-20-41S (SR)
2	20	Electrical Connector (P1002-P1003)	MS3106E-16-26S (SR)
1	21	Electrical Connector (P1005)	BL06-14-18S-TC2
1	22	Electrical Connector (P3003)	MS3106E-28-10S (SR)
2	23	Load Jumper Cable	49-100-588-501
2	24	Plug, Aluminum (Parker)	4-PNTX-D
1	25	Cap. Cres. (Parker)	8-FNTX-SS
2	26	Cap. Cres. (Parker)	6-FNTX-SS
2	27	Cap. Cres. (Parker)	4-FNTX-SS
3	28	Nut, Coupling Cres. (Parker)	PAH-124-C4W
2	29	Nut, Coupling Cres. (Parker)	PAH-124-C6W
3	30	Sleeve, Cres. (Parker)	PAH-125-C4
2	31	Sleeve, Cres. (Parker)	PAH-125-C6
3	32	Cap. Elec. Conn. Receptacle	MS3181-12N
7	33	Cap. Elec. Conn. Receptacle SHIP 4 ONLY (REMAINING 3 ON APPARATUS USED FOR TEST BUT NOT SHIPPED)	MS3181-14N
5	34	Cap. Elec. Conn. Receptacle	MS3181-16N
4	35	Cap. Elec. Conn. Receptacle SHIP 2 ONLY (REMAINING 2 ON APPARATUS USED FOR TEST BUT NOT SHIPPED)	MS3181-18N
1	36	Cap. Elec. Conn. Receptacle	MS3181-20N
4	37	Cap. Elec. Conn. Receptacle	MS3181-22N
1	38	Cap. Elec. Conn. Receptacle (Bendix)	10-70500-28
3	39		
3	40	Cap. Elec. Conn. Plug	MS3180-12C
7	41	Cap. Elec. Conn. Plug SHIP 4 ONLY (REMAINING 3 ON APPARATUS USED FOR TEST BUT NOT SHIPPED)	MS3180-14C

2-2A-2

[illegible]

## RECOMMENDED SYSTEM #5R CONTROLS AND MONITOR LIGHTS

<u>Support</u>	<u>Light Monitor</u>
1. Helium purge - valve control (switch)	x (open)
2. 115 Vac power (switch)	x (on)
3. 28 Vac auxiliary power (switch)	x (on)
4. Fan power (switch)	x (on)
5. WRS heater tape (switch)	x (on)
6. Vacuum pump - 1.5 psia (switch)	x (on)
7. Vacuum pump - reference vacuum (switch)	x (on)
8. Coolant pump (switch)	x (on)
9. Stack heater power (switch) 1600 watt should have circuit for thermostat	x (on)

Figure 2-3

REQ. ITEM	DESCRIPTION	MATERIAL	PART NUMBER
X 1	AUXILIARY RELAY BOX ASSEMBLY (PIX)		49-400-318-401
1	BOX	6061-T4 ALUM	49-300-555-001
1	COVER	6061-T4 ALUM	49-300-556-001
1	CHANNEL	6061-T4 ALUM	49-300-557-001
2	MOUNTING PLATE	6061-T4 ALUM	49-100-768-001
1	NAMEPLATE	.06 LAMICOID	49-100-770-001
1	INSULATOR BOARD	.031 GLASS EPONXG10	49-100-771-001
1	CIRCUIT BOARD	.062 GLASS EPONXG10	49-200-608-001
2	BUS BAR	BRASS	49-200-606-001
X 10	SCHEMATIC WIRING DIAGRAM		49-400-314-401
1	CONNECTOR		MS3114E22-55PY
1	"		MS3114E22-55PX
11	RELAY DPDT, 24 VDC, POTTER & BRUMFIELD SC11DB		49-200-550-013
30	TERMINAL, SOLDER, CAMBION TYPE 2051-1		49-200-550-014
22	" " VECTOR T-13		49-200-550-015
4	WIRING, INSULATED, BURNDY YEC-180		MS25311-180
4	" " BURNDY YEC-110		MS25311-110
11	SILICON RECTIFIER, MOTOROLA IN4003		49-200-550-018
8	MACHINE SCREW #4-40 x .375 FINE HD CR-BEC	CRCS	MS24693-74

2-4-A-1

6	20	NUT, HEX #4-40					NA5671C-4
100	21	LOCK WASHER, SPLIT, #4					MS3533B-78
22	22	MACHINE SCREW, PAN HD, CROSS-REC, #4-40 x .37					MS51957-15
44	23	MACHINE SCREW, FL-HD, #4-40 x .31					AN500D4-5
68	24	CLINCH NUT, #4-40, PENN ENGRG, SL4-40-2					49-200-550-024
50 FT.	25	WIRE, BLACK, UNSHIELDED, B-20 (19/32), ALPHA MSG/19, CAR 2					49-200-550-025
3 FT.	26	" GRAY, " " " " " " " 9					49-200-550-026
3 FT.	27	" BROWN, " " " " " " " 7					49-200-550-027
3 FT.	28	" GREEN, " " " " " " " 4					49-200-550-028
1 FT.	29	" WHITE & RED, " " " " " " " 12					49-200-550-029
1 FT.	30	" " BLUE, " " " " " " " 15					49-200-550-030
1 FT.	31	" " YELLOW, " " " " " " " 14					49-200-550-031
4	32	TUBULAR BRAID, .078 I.D., ALPHA # 2163, 10 IN. LG.					QQ-B-575
4	33	" " 172 I.D., " # 2168, 10 IN. LG.					QQ-B-575
4 FT.	34	TUBING, HEAT REACTIVE, (MSFC-276)				276-11BK012 DIA	49-200-550-034
4 FT.	35	" " "				276-11BK025 DIA	49-200-550-035
YES	36	SOLDER, SN-60				QQ-S-571 TYPE RA	49-200-550-036
YES	37	FLUX				MIL-F-14256 TYPE A	49-200-550-037
4 FT.	38	LACING, .000 FLAT NYLON BLACK, ALPHA LC-136				MIL-T-713A	49-200-550-038

014-21-66 DRA RCM  
 RECD 4/21/66 RECEIVED  
 ITEM 12 WAS 3 REQ.  
 ITEM 13 WAS 3 REQ.  
 ITEM 14 WAS 3 FT. REQ.  
 02 5-4-66 DRA RCM  
 03 5-4-66 DRA RCM

(49-200-431-504, -505  
 USED ON P/L: 49-200-495-501-502, -503

UNLESS OTHERWISE NOTED DIMENSIONS ARE IN INCHES & MACHINING TOLERANCES ARE:			ANGLES ±
DIMENSIONS	UP TO 6 INCL.	OVER 6 TO 24 INCL.	OVER 24
FRACTIONAL	±	±	±
DECIMAL	±	±	±

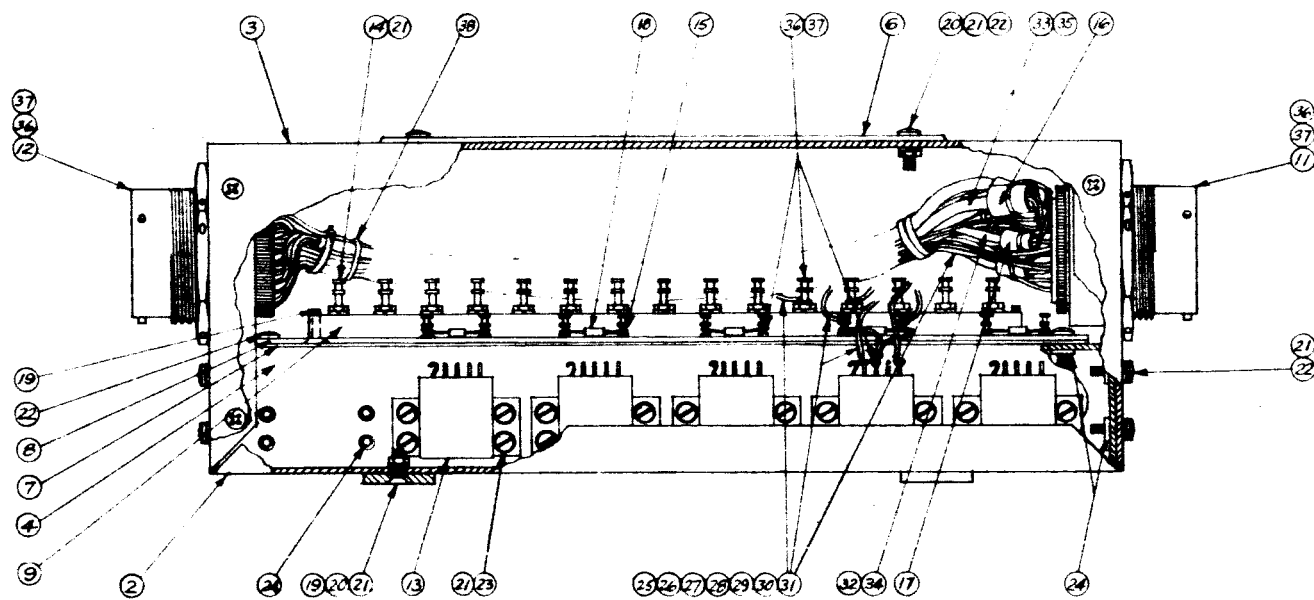
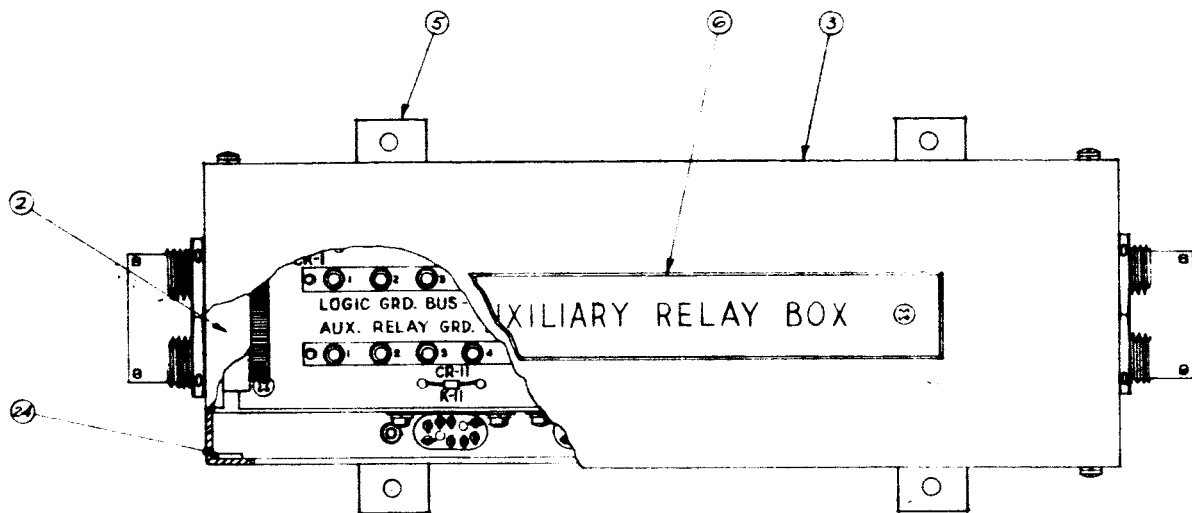
— Confidential — Property of  
 ALLIS-CHALMERS MFG. CO., Milwaukee, Wis.  
 RESEARCH 3341 DEPT.  
 DRN DSH 4/15-66  
 TRD  
 CH'D DSH 4-20-66  
 APP'D RCM 4-16-66

AUXILIARY  
 RELAY BOX  
 ASSEMBLY

P/L FIGURE 2-4A

SIMILAR TO: — SCALE: — 1/16  
 SHEET 1  
 OF 1  
 49-200-550-501

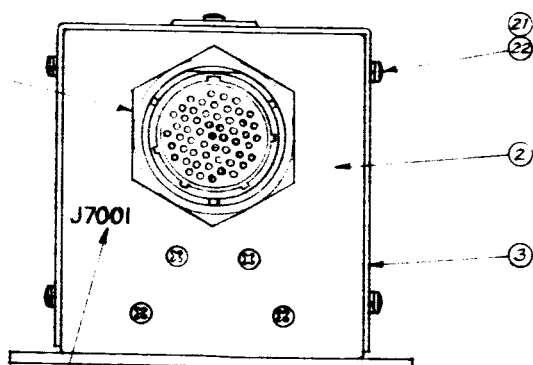




2-4-B-1

NOTES:

1. FOR SCHEMATIC WIRING DIAGRAM, SEE DRAWING 49-400-314-401.
2. SOLDER WHERE REQUIRED PER NPC-200-4.
3. HEAT SHRINK TUBING PER MSFC-PROC 273.
4. ROUTING, LACING, ETC. PER ELECTRICAL FABRICATION PROCEDURE MSFC-PROC-256.
5. CHECK FOR CONTINUITY PER SCHEMATIC.

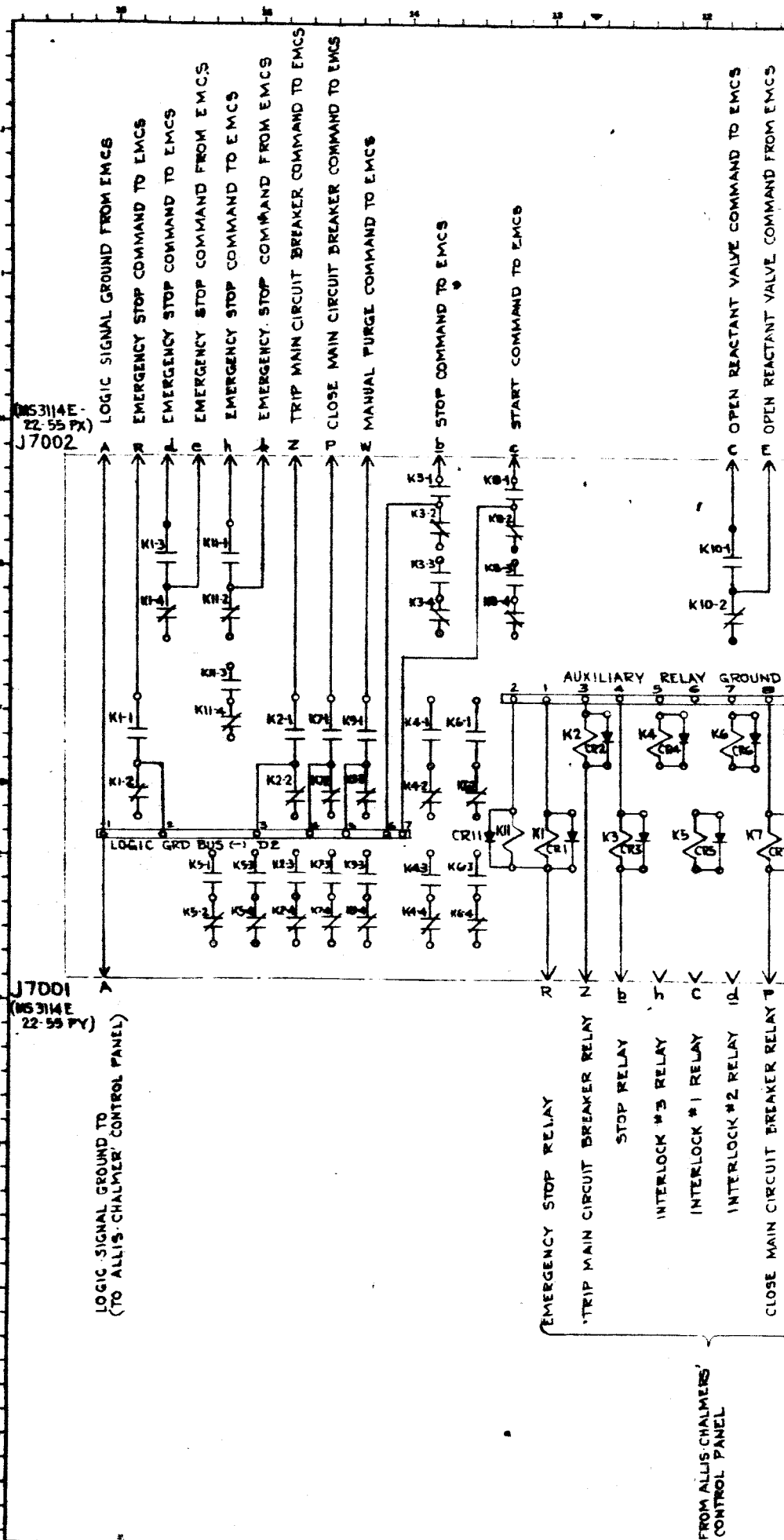


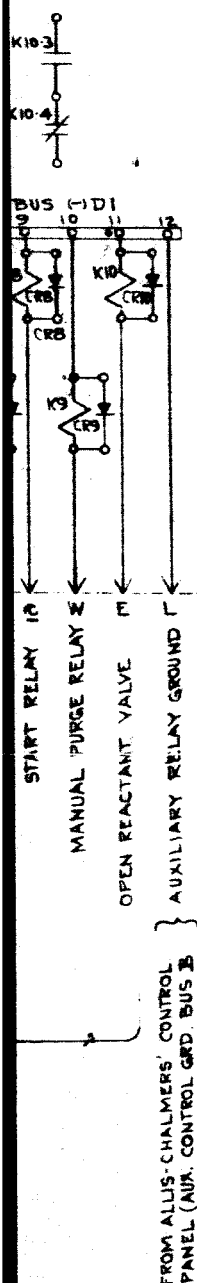
RUBBER STAMP CONNECTOR IDENTIFICATION NUMBERS, IN WHITE, .19 HIGH. "J7001" ON SAME END AS ITEM (1), "J7002" ON OPPOSITE END.

USED ON P/L 49-200-550-501

CONFIDENTIAL - PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b> RESEARCH 3341 PLANT		NAME <b>AUXILIARY RELAY BOX ASSEMBLY</b>	
UNLESS OTHERWISE SPECIFIED: 1 - PLACE DEC 6 2 - PLACE DEC 6 3 - PLACE DEC 6 ANGULAR A		DATE FIGURE 2-4B	
DR <b>DSN 4-84</b> CHCK <b>1/2/16</b> AP <b>RGM J-24</b>		<input checked="" type="checkbox"/> MACHINED SURFACE TEXTURE	PART NO. <b>49-400-318-401</b>
SCALE <b>FULL</b>		SHEET <b>1 of 1</b>	PRINTED IN U.S.A. FORM NO. 000-1

49-400-318-401 01





B	AMP. HOUR COUNTER (READOUT) FROM EMCS
D	+11.2 VOLT D.C. (READOUT) FROM EMCS
T	-3.6 VOLT D.C. (READOUT) FROM EMCS
K	+28 VOLT DC VALVE POWER (READOUT) FROM EMCS
Z	FUEL CELL AMPS (-) HI (READOUT) FROM EMCS
S	FUEL CELL AMPS (-) LOW (READOUT) FROM EMCS
K	OK TO APPLY REACTANTS CIRCUIT TO EMCS
P	OK TO APPLY REACTANTS READY SIGNAL FROM EMCS
X	MAIN HEATER CIRCUIT TO EMCS
X	SIGNAL TO MAIN HEATER CONTROL FROM EMCS
T	MAIN CIRCUIT BREAKER CIRCUIT TO EMCS
U	SIGNAL TO TRIP MAIN CIRCUIT BREAKER FROM EMCS
C	SIGNAL TO CLOSE MAIN CIRCUIT BREAKER FROM EMCS
→	SHIELD

TO ALLIS CHALMERS' CONTROL PANEL

FROM ALLIS CHALMERS' CONTROL PANEL

TO ALLIS CHALMERS' CONTROL PANEL

FROM ALLIS CHALMERS' CONTROL PANEL

TO ALLIS CHALMERS' CONTROL PANEL

FROM ALLIS CHALMERS' CONTROL PANEL

TO ALLIS CHALMERS' CONTROL PANEL

TO ALLIS CHALMERS' CONTROL PANEL

FROM ALLIS-CHALMERS' CONTROL PANEL

01	4-8-66 D.P.B.
MUST REM. VALUE CODE ATTACHED	
02	5-4-66 D.P.B.
MUST REM. VALUE CODE ATTACHED	



USED ON 49-200-550-501

CONFIDENTIAL - PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b> 3341 M1 WORKS		NAME <b>SCHEMATIC DIAGRAM          AUXILIARY RELAY BOX</b>	
UNLESS OTHERWISE SPECIFIED:		MATL.	
1. PLACE DEC. A 2. PLACE DEC. A 3. PLACE DEC. A ANGULAR A	<input checked="" type="checkbox"/> <b>MACHINED SURFACE TEXTURE</b>	<b>FIGURE 2-4C</b> WT.	
DR <b>GIT 4-8-66</b> CH <b>D.P.B. 4-8-66</b> AP <b>B.M. 4-8-66</b>	SIMILAR TO <b>NONE</b>	PART NO <b>49-400-314-401</b>	
SCALE SHEET		PART NO <b>49-400-314-401</b>	

100-314-401	02
-------------	----

NOTE 1

EACH WIRE MUST BE CHECKED FOR ELECTRICAL CONTINUITY PER WIRING DIAGRAM.

NOTE 2

EACH WIRE MUST BE CHECKED FOR INSULATION RESISTANCE TO EACH OTHER WIRE. THE INSULATION RESISTANCE SHALL EXCEED 50 MEGOHMS AT 500 VOLTS D.C.

NOTE 3

SOLDER AS PER NPC 2004

REQ	IT.	
<input checked="" type="checkbox"/>	1	CABLE, E
1	2	PLUG -
1	3	PLUG -
5	4	UNSHIELDED
1	5	.38 TUBING
2	6	BOOT STR
YES	7	ADHESIVE
YES	8	SOLDER
YES	9	SOLDER
2	10	.38 TUBING

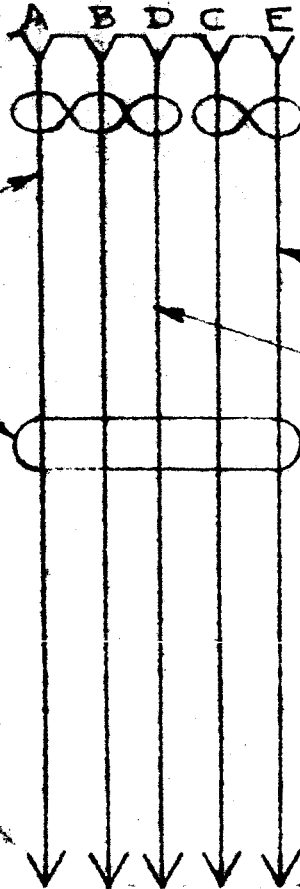
(10) (9) (8) (7) (6) (2)

P2003  
(MS 3116 E14-5S)

SHRINK AS PER  
MSFC-PROC-273

(4)

(5)



16" O  
OVERALL LE

(10) (9) (8) (7) (6) (3)

A B D C E  
P5004 (MS 3116 E14-5S)

(2) P2003 EN

(3) P5004 EN

49

DESCRIPTION	MATERIAL
MCS TO CONTROL PANEL	
WIRE B16(19/29) 600V- 18 FT LG.	49-150-00
, HEAT REACTIVE(MSFC-276) 18 FT LG.	276-31BK0
NIGHT THIN WALL 202 A0 42-3	
- RAYCLAD S-1006	
SN 60	QQ -S-5 TYPE RA
FLUX	MIL-F-142 TYPE A
HEAT REACTIVE(MSFC-276) 1.0 IN. LG	276-31BK0

#### NOTE 4.

RUBBER STAMP OR STENCIL C  
NUMBERS (P2003 AND P500  
CONNECTORS USING CHARAC  
SIMILARLY APPLY "PART N  
MIDWAY BETWEEN CONNECT

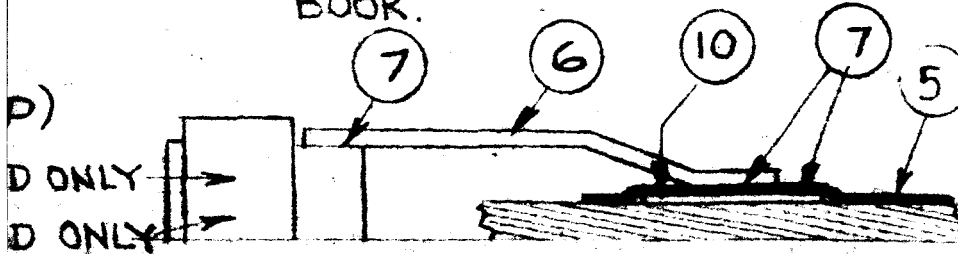
HELICALLY TWIST TWO LENGTHS OF ITEM  
TO FORM A MULTI-CONDUCTOR CABLE

HELICALLY TWIST THREE LENGTHS OF  
TO FORM A MULTI-CONDUCTOR CABLE

FINALLY HELICALLY TWIST THE AB  
ONE CABLE. LENGTH OF LAY 3.5 TO  
LAY TO BE OPPOSITE THAT OF INDIVI

#### NOTE 5

FOR ASSEMBLY PROCEDURE SEE  
RAYCLAD THERMOFIT INSTRUCTION  
BOOK.



DETAIL  
CABLE ENDS

200-523-501 02

2-5-2

IAL	PART NO.
	49-200-523-501
	MS 3116 E14-5S
	MS 3116 E14-5P
9-013	49-200-523-004
D38DIA	49-200-523-005
	49-250-014-004
	49-150-010-003
1	49-200-523-008
56	49-200-523-009
D38DIA	49-200-523-010

01	3-23-66 D.P.D. RGM
CAND 3-23-66 D REL ED 11/4 8/7/66	
ITEM 5 WAS .50 TUBING 276-31 BK OD 50 DIA. IT IS ADDED (A-4) DETAIL ADDED	
02	4-30-66 D.P.D. RGM
CAND 4-30-66 D REL ED 11/4 8/7/66	

CONNECTOR IDENTIFICATION  
 4) ADJACENT RESPECTIVE  
 TERS APPROXIMATELY .12 HIGH,  
 D. 49-200-523-501  
 ORS.

4 } FORM ONE TURN FOR  
 EACH 1.6 TO 2.2 INCHES  
 OF LENGTH, DIRECTION OF  
 T.4 } LAY TO BE THE SAME  
 FOR BOTH CABLES.

OVE TWO CABLES TO FORM  
 49 INCHES. DIRECTION OF  
 DUAL CABLES.

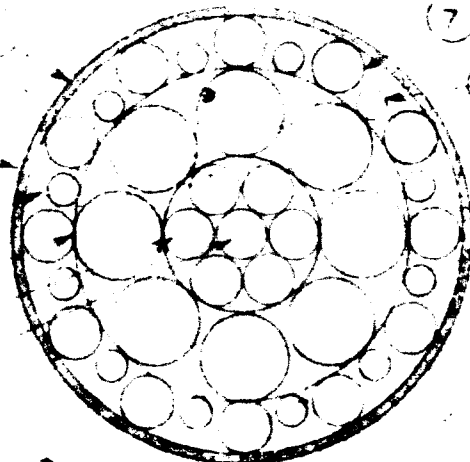
USED ON ASSY  
 49-200-495-501

CONFIDENTIAL—PROPERTY OF ALLIS-CHALMERS MFG. CO. MI PLANT		NAME CABLE, EMCS TO CONTROL PANEL P2003 TO P5004	
UNLESS OTHERWISE SPECIFIED:		MATERIAL	
1-PLACE DEC ±	<input checked="" type="checkbox"/> MACHINED SURFACE TEXTURE	FIGURE 2-5	
2-PLACE DEC ±			
3-PLACE DEC ±			
ANGULAR ±			
OR GJT 3-23-66	SIMILAR TO	WT	
OR S.P.D. 3-23-66			
OR RGM 3/24/66			
SCALE	SHEET	PART NO.	
NTS	—	49-200-523-501	



SHRINK AS PER  
MSEFC PROC 273

SECTION THRU  
CABLE



CORE TO CONSIST OF ONE LENGTH OF SINGLE CONDUCTOR  
SHIELDED (ITEM 7)

FIRST LAYER TO CONSIST OF -

54 LENGTHS OF SINGLE CONDUCTOR SHIELDED (ITEM 7)  
HELICALLY TWIST ABOUT CORE TO FORM ONE TURN FOR  
EACH 3.2 TO 4.5 INCHES OF LENGTH

SECOND LAYER TO CONSIST OF -

SEVEN LENGTHS OF TWO CONDUCTOR SHIELDED (ITEM 8)  
ONE LENGTH OF THREE CONDUCTOR SHIELDED (ITEM 9)

DIRECTION OF LAY OPPOSITE THAT OF FIRST LAYER  
LENGTH OF LAY - 7.0 TO 9.8 INCHES OF LENGTH

THIRD LAYER TO CONSIST OF -

12 LENGTHS OF SINGLE CONDUCTOR SHIELDED (ITEM 7)  
12 LENGTHS OF SINGLE CONDUCTOR UNSHIELDED (ITEM 4)

DIRECTION OF LAY OPPOSITE THAT OF SECOND LAYER  
LENGTH OF LAY - 9.1 TO 12.7 INCHES OF LENGTH

REQ	IT	DESCRIPTION
1	1	CABLE LISTS TO ANAL
2	2	PLUG
3	3	PLUG
4	4	UNSHIELDED WIRE B20
5	5	LOT BANG HEAT REACTIVE
6	6	BOOTS TO WITHIN 1/2 INCH
7	7	SHIELDED CABLE ONE B20
8	8	SHIELDED CABLE TWO B20
9	9	SHIELDED CABLE THREE B20
10	10	ADHESIVE - RAYCLAD
11	11	BOOT STRAIGHT TO WIRE
12	12	INSULATED UNIFORM
13	13	INSULATED UNIFORM
14	14	INSULATED UNIFORM
15	15	INSULATED UNIFORM
16	16	UNINSULATED UNIFORM
17	17	SOLDER SN 60
18	18	SOLDER FLUX
19	19	UNSHIELDED WIRE B20
20	20	15 TUBING HEAT REACT

SEE NOTE 5

P2005

(MS3116E 22-55 S)

17 18 12 10 11

20

15 16

9

INSTALLED  
WITHIN BOOT

5

8

7

STAGGER ALONG  
LENGTH OF CABLE  
WITHIN BOOT

13

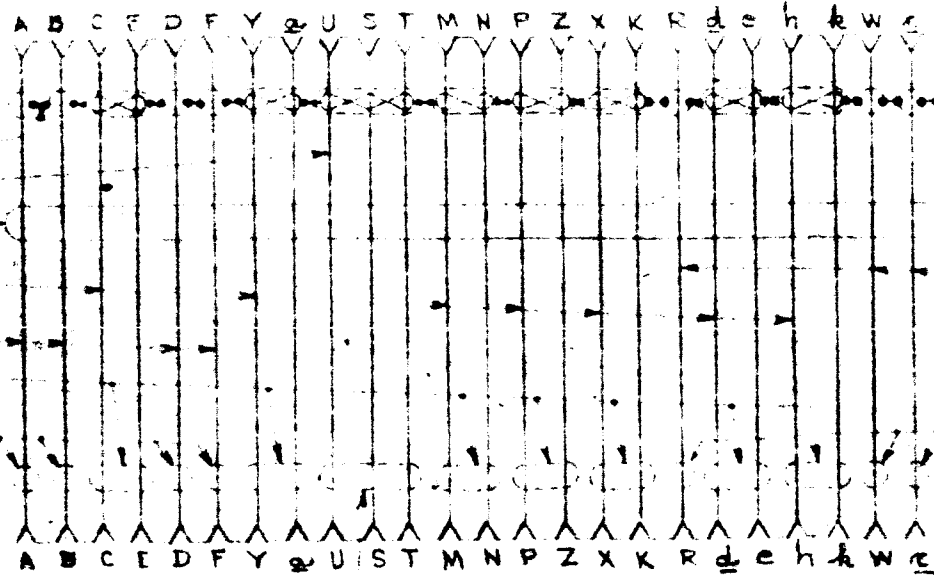
12

20 17 18 13 10 11

(MS3116E 22-55 SX)

P7002

SEE NOTE 5

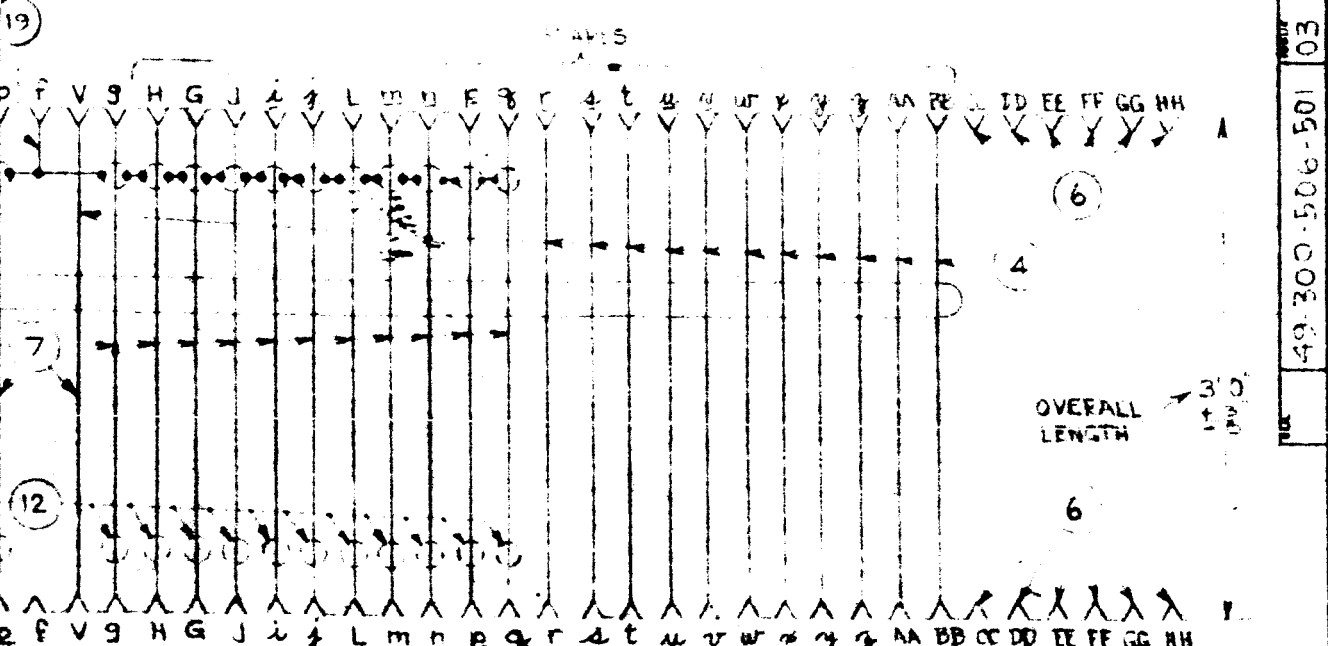


- NOTE 1. EACH WIRE MUST BE CHECKED FOR ELECTRICAL CONTINUITY PER W  
2. EACH WIRE MUST BE CHECKED FOR INSULATION RESISTANCE TO EACH  
THE INSULATION RESISTANCE SHALL EXCEED 50 MEGOHMS AT  
3. SOLDER AS PER NPC 200-4  
4. FOR ASSEMBLY PROCEDURE SEE RAYCLAD THERMOFIT INST  
5. RUBBER STAMP OR STENCIL CONNECTOR IDENTIFICATION NUMBERS (F  
RESPECTIVE CONNECTORS USING CHARACTERS APPROXIMATELY 1/16  
SIMILARLY APPLY "PART NO. 49-300-506-501 MIDWAY BETWEEN  
6. APPLY SHIELD TERMINATIONS (ITEMS 12 THRU 16 INCLUSIVE.) AS

MATERIAL		PART NO	2	AT P2005 END ONLY	1	3
LAY BOX		49300 506 501				
		MS31GE22 555				
		MS31GE22 55 3X				
9/32) 600V 48 IN LG	43 150 003 010	49 300 506 004				
MSFC 276)	42 IN LG 276 3 BK 150 DIA	49 300 506 005				
NS PLUG		49 150 011 001	(10)			
9/32)	48 IN LG 49 150 009 000	49 300 506 007				
9/32)	48 IN LG 49 150 009 006	49 300 506 008	(3)			
9/32)	48 IN LG 49 150 009 008	49 300 506 009	(2)			
1006		49 150 010 003	(3)			
202A074 3		49 250 014 007	(14)			
		MS 15311 110				
		MS 15311 150				
		MS 15311 100				
22		49 150 012 008				
MS 100 572		49 150 012 021				
	22 - S 571	49 300 506 017	(20)			
	TYPE RA	49 300 506 018				
	4256	49 300 506 013	(10)			
	TYPE A	49 300 506 020				
9/32) 600V 6 IN LG	49 150 003 026	49 300 506 013				
E (MSFC 276)	15 IN LG 276 3 BK 150 DIA	49 300 506 020				

ITEM 20 ADDED  
(F) DETAIL ADDED  
22 430 001 001  
ITEM 5 WAS 15 15 15  
NAT WAS 276 3 BK 150 DIA  
(25) 15 15 15  
15 15 15 15 15

DETAIL  
END OF CABLE



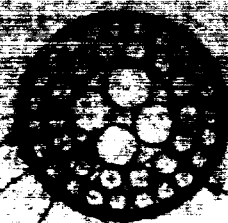
USED ON ASSY  
49-200 495-501

IRING DIAGRAM.  
OTHER WIRE  
500 VOLTS D.C.  
DUCTION BOOK.  
2005 AND P7002) ADJACENT  
HIGH.  
CONNECTORS.  
PER ABMA-PD-E-55.

CONFIDENTIAL - PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b> 3341 MI WORKS		NAME CABLE EMCS TO AUX RELAY BOX (P2005 TO P7002)	
UNLESS OTHERWISE SPECIFIED:		MATERIAL	
1. PLACE DEC.	<input checked="" type="checkbox"/> SURFACED <input type="checkbox"/> POLISHED <input type="checkbox"/> TUMBLING	<b>FIGURE 2-6</b>	
2. PLACE DEC.			
3. PLACE DEC.			
APPROVAL	DATE	PART NO <b>49-300-506-501</b>	
DATE 3-23-64	DATE 3-23-64	DATE 3-23-64	DATE 3-23-64
APPROVAL	DATE	DATE	DATE

2-6-2

49-300-506-501 03



SECTION THRU  
CABLE

CORE TO CONSIST OF:  
ONE LENGTH OF THREE CONDUCTOR  
SHIELDED CABLE (ITEM 9)  
THREE LENGTHS OF TWO CONDUCTOR  
SHIELDED CABLE (ITEM 8)  
HELICALLY TWIST TO FORM ONE TURN FOR  
EACH 4.4 TO 6.1 INCHES OF LENGTH.  
CENTER LAYER TO CONSIST OF:  
14 LENGTHS OF SINGLE CONDUCTOR  
SHIELDED (ITEM 7)  
DIRECTION OF LAY OPPOSITE THAT OF CORE.  
LENGTH OF LAY 6.5 TO 8.1 INCHES  
OUTER LAYER TO CONSIST OF:  
23 LENGTHS OF UNSHIELDED SINGLE  
CONDUCTOR WIRE (ITEMS 6 & 19)  
DIRECTION OF LAY TO BE OPPOSITE THAT  
OF CENTER LAYER  
LENGTH OF LAY 7.8 TO 11.0 INCHES

11 2 10 17 18 20

P7001 (M63116 E22-55 SY)

	A	B	D	F	N	M	Y	α	U	S	T	X	K	F	V	Z	b	h	c	d	P	W	E	R	L	G	H	J	e	g	α
14																															
12																															
13																															
8																															
9																															
7																															
15																															
16																															
INSTALL WITH IN																															
BOOT																															

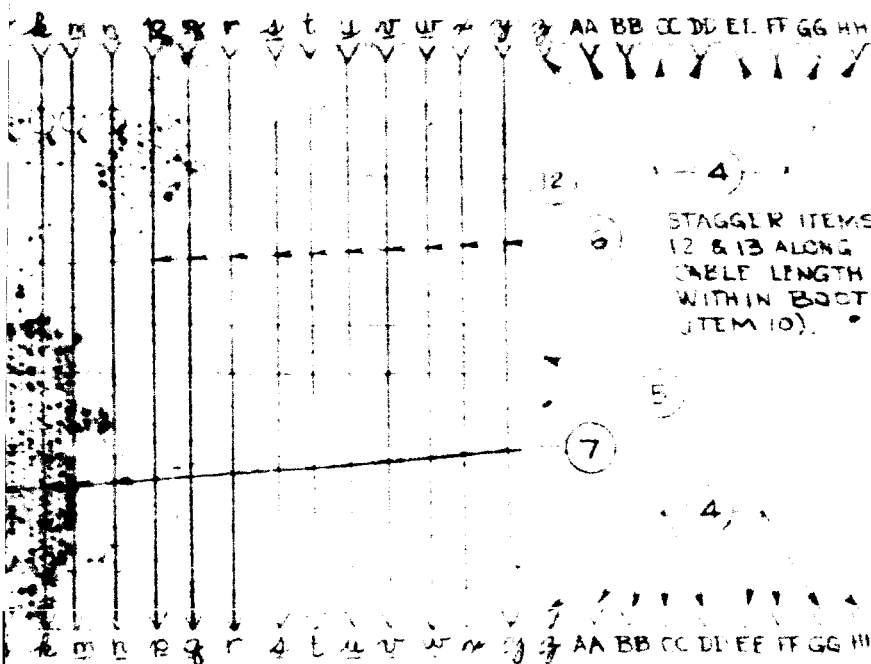
P5003 (M63116 E22-55 PZ)

20 3 10 11 17 18

- NOTE 1. EACH WIRE MUST BE CHECKED FOR ELECTRICAL CONTINUITY PER V.  
2. EACH WIRE MUST BE CHECKED FOR INSULATION RESISTANCE TO EARTH.  
THE INSULATION RESISTANCE SHALL EXCEED 50 MEGOHMS AT 50°C.  
3. SOLDER AS PER NRC 200-4  
4. FOR ASSEMBLY PROCEDURE SEE RAYOLAL THERMOFIT INSTR.

5. R. THIS CABLE IS TO BE USED FOR THE CABLE ASSEMBLY.  
6. APPLY SHIELD TERMINATIONS (ITEMS 12 THRU 16 INCLUSIVE)

	MATERIAL	PART NO.
CONTROL PANEL		49-300-520-001
		MS 25311-110
		MS 25311-150
THE PLUG		49-150-011-001
FC-276) 18 FT LG	276 31BK1D00DIA	49-300-520-005
32) COND. 18 FT LG	49-150-009-010	49-300-520-006
32) COND. 18 FT LG	49-150-009-090	49-300-520-007
32) COND. 18 FT LG	49-150-009-106	49-300-520-008
1/32) COND. 18 FT LG	49-150-009-138	49-300-520-009
12 AQ74-3		49-250-014-007
006		49-150-010-003
		MS 25311-110
		MS 25311-150
		MS 25311-160
		49-150-012-005
1G-YOC 676		49-150-012-019
	QB-S-571 TYPE BA	49-300-520-017
	MIL-F-14256 TYPE A	49-300-520-018
32) COND. 19 FT LG	49-150-009-026	49-300-520-019
15FC 276) 1.5 IN LG	276 31BK1D00DIA	49-300-520-020



SPARES

WING DIAGRAM  
OTHER V. PE  
0 VOLTS 11 C.

SECTION BOOK

1000  
0513

AS PER ABMA PD-E 13.

CONFIDENTIAL PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b>		NAME	
UNLESS OTHERWISE SPECIFIED		MATL	
1. PLACE DEC.	MACHINED SURFACE TEXTURE	<b>FIGURE 2-7</b>	
2. PLACE DEC.			
3. PLACE DEC.			
ANGULAR			
DR	SIMILAR TO		
CH	SCALE	SHEET	PART NO.
AP	NTS		<b>49-300-520-501</b>

provides additional information on the recommended test setup for the gases, liquid coolant, and the product water.

**CAUTION:** Care should be exercised in making all connections to prevent damage to electrical connectors and to insure leak tight tube connections. Refer to Paragraph 2.2 Initial Installation Check-out before making any external interface connections.

- C. A data acquisition system and direct readout equipment, which must also provide the excitation voltage for the five (5) Allis-Chalmers transducers to be monitored. A listing of the recommended functions and the method of monitoring or recording is contained in Figure 2-8.
- D. Temperature recording equipment will be required for fuel cell temperatures. Allis-Chalmers will furnish mating connectors for the fuel cell module instrumentation, but not the cable. All thermocouple connections will be made up for iron constantan couples. No reference junction compensation circuits are furnished.
- E. The 400-cycle three-phase power for the fuel cell coolant fans rated to supply at least 150 watts at 200 volts.
- F. The customer will furnish the electrical load bank for the fuel cell output capability testing and the load disconnect contactor.

# RECOMMENDED MONITOR FUNCTIONS FOR SYSTEM #5R

	Digitec	Panel Meter	VIV Recorder	Data Acquis.	Other
1. Total fuel cell voltage		x	x	x	
2. Total fuel cell current      Both on Panel Control		x	x	x	
3. Stack temperature (TC-8 and TC-9)			x	x	
4. Cavity pressure	x		x	x	
5. Individual cell volts	x			x	
**6. Fan current		x		x	
7. Heater current (28 vdc, 1600 watts)		x		x	
8. H <sub>2</sub> inlet pressure	x			x	
9. O <sub>2</sub> inlet pressure	x			x	
10. Canister pressure	x			x	
11. -5.0 vdc				x	
12. +11.2 vdc				x	
**13. System "ON" lapse time		x			
*14. Load "ON" lapse time		x			
**15. WRS heating tape temperature		x		x	
16. Stack thermocouples - inside (7 additional)				x	
17. Fan No. 1 RPM	x			x	
18. Fan No. 2 RPM	x			x	
**19. Fan Voltage				x	
20. Stack liquid coolant differential pressure				x	
*21. Stack liquid coolant outlet temperature				x	x
*22. Liquid coolant inlet temperature					
**23. Inlet liquid coolant flow					
**24. Liquid coolant-inlet pressure				x	
**25. Liquid coolant-outlet pressure				x	
**26. Helium inlet pressure				x	
**27. H <sub>2</sub> supply pressure		x		x	
**28. O <sub>2</sub> supply pressure		x		x	
29. Barometric pressure				x	

\* Function has changes of short duration requiring continuous recording if detailed information is required.

\*\* Sensors to be provided by NASA.

2.1.5 Supplementary Installation Information

Figure 2-9 (Connector Interface function description) contains additional interface information to aid in the assembly of the test setup.

2.2 INITIAL INSTALLATION CHECKOUT

2.2.1 Electrical Testing

The external interface connections should be verified as per Figure 1-2A, 1-2B, and 1-2C (Drawing No. 49-500-125-404) and Figure 2-2A (Drawing No. 49-400-260-405), NASA's interface should be checked prior to the initial application of any power or reactants.

Instrumentation calibration data for use during checkout and operating can also be found in Figure 2-9.

2.2.2 Leak Test

Prior to the initial operation of the fuel cell, the reactant gas system, the water removal cavity, and the canister enclosure must be subjected to a leak test. The following is the procedure for performing this test, but it should not be performed initially without a representative from Allis-Chalmers present. Figure 2-10 is a suggested schematic only.

## NASA INTERFACE CONNECTIONS

### J5001-NASA Control Panel

- A + 28 vdc input from fuel cell
- B - 28 vdc input from fuel cell and main ground

### J5002-NASA Control Panel

- A + 28 vdc input from auxiliary supply
- B - 28 vdc input from auxiliary supply and main ground.

### J5006-NASA Control Panel

- A Logic signal ground output
- D +11.2 vdc logic control voltage output - for monitor only
- F -5.6 vdc logic control voltage output - for monitor only
- V + 28 vdc valve power output - for monitor only
- K + 28 vdc output to main heater control - for 1,600 watt heater
- G - 28 vdc ground output to main heater control - for 1,600 watt heater
- S + 28 vdc output to trip main circuit breaker
- U + 28 vdc output to close main circuit breaker
- H - 28 vdc ground output to main circuit breaker

### Optional Monitor Signals

J5006-B      Amp-hour readout - NOTE: This signal is displayed on panel (Amp-Hour Counter)

### J3003 - 1,600 watt heater

- E + 28 vdc 29 amp heater power for first 800-watt heater
- G - 28 vdc ground heater power for first 800-watt heater
- C 800-watt heater thermostat N. O. for first
- H 800-watt heater thermostat N. O. 800-watt heater
- M Shield for Pins C and H, isolated from stack
- J + 28 vdc 29 amp heater power for second 800-watt heater
- K -28 vdc ground heater power for second 800-watt heater
- A 800-watt heater thermostat N. O. for second
- B 800-watt heater thermostat N. O. 800-watt heater
- L Shield for Pins A and B, isolated from stack



## NASA INTERFACE CONNECTIONS

### J1000- Fuel Cell Fans and RPM

- E Fan reluctance pickup "AC" side, output signal 150 millivolts peak-to peak at approximately 1300 cps representing 20,000
- F RPM to be fed into AC to DC converter for instrumentation readout.
- G Shield for Pin E and F.
- J Fan reluctance pickup "BC" side, output signal 150 millivolts peak-to peak at approximately 1300 cps representing 20,000
- H RPM to be fed into AC to DC converter for instrumentation readout.
- T Shield for Pin J and H
- A Phase "A" Fan power input for A-C Side, 200 Vac,
- B Phase "B" 400 cps, 3-phase at approximately
- C Phase "C" 0.15 amp. per phase
- D Shield for Pins A, B, and C carried through to motor, isolated from case ground.
  
- K Phase "A" Fan power input for B-D side, 200 Vac.
- L Phase "B" 400 cps, 3-phase at approximately
- M Phase "C" 0.15 amp. per phase
- N Shield for Pins K, L, and M carried through to motor, isolated from case ground.

# NASA INTERFACE CONNECTIONS

## CELL VOLTAGES - SYSTEM # 5R

Connectors J1002 and J1003 On The Fuel Cell

<u>Cell No.</u>	<u>Plus Pin</u>	<u>Minus Pin</u>
1	J1003b	J1002b
2	J1002b	J1002a
3	J1002a	J1003a
4	J1003a	J1003Z
5	J1003Z	J1002Z
6	J1002Z	J1002Y
7	J1002Y	J1003Y
8	J1003Y	J1003X
9	J1003X	J1002X
10	J1002X	J1002W
11	J1002W	J1003W
12	J1003W	J1003V
13	J1003V	J1002V
14	J1002V	J1002U
15	J1002U	J1003U
16	J1002U	J1003T
17	J1003T	J1002T
18	J1002T	J1002S
19	J1002S	J1003S
20	J1003S	J1003R
21	J1003R	J1002R
22	J1002R	J1002P
23	J1002P	J1003P
24	J1003P	J1003N
25	J1003N	J1002N
26	J1002N	J1002M
27	J1002M	J1003M
28	J1003M	J1003L
29	J1003L	J1002L
30	J1002L	J1002K
31	J1002K	J1003K
32	J1003K	J1003J
33	J1003J	J1002J

### Spare Pins

J1002 - A,B,C,D,E,F,G,H  
J1003 - A,B,C,D,E,F,G,H

Pin c left open for cable shield tie point  
Pin c left open for cable shield tie point

Figure 2-9 (cont.)  
(sheet 3 of 5)

## NASA INTERFACE CONNECTIONS

### Transducer Calibration Data

Note: All Transducers - 5000  $\pm$  100 ohms element-max  
current 10 ma

#### J3001 Fuel Cell Pressure Transducer

i	O <sub>2</sub> Inlet absolute pressure transducer (pin i transducer ground)
h	+ Excitation voltage
f	- Excitation Voltage and - signal output
g	+ Signal output, 0-50 psia $\pm$ 1% Sensitivity 0.02 volt/psia/volt excitation
a	H <sub>2</sub> Inlet absolute pressure transducer (pin a transducer ground)
Z	+ Excitation voltage
X	- Excitation voltage and - signal output
Y	+ Signal output, 0-50 psia $\pm$ 1% Sensitivity 0.02 volt/psia/volt excitation
e	Canister pressure transducer (pin e transducer ground)
d	+ Excitation voltage
b	- Excitation voltage and - signal output
C	+ Signal output, 0-50 psia $\pm$ 1% Sensitivity 0.02 volt/psia/volt excitation
J	Liquid coolant differential pressure transducer (pin J transducer ground)
H	+ Excitation voltage
F	- Excitation voltage and - signal output
G	+ Signal output, -10 to + 10 psid $\pm$ 1.25% Sensitivity 0.05 volt/psid/volt excitation
W	Water cavity readout pressure transducer (pin W transducer ground)
V	+ Excitation voltage
T	- Excitation voltage and - signal output
U	+ Signal output 0-25 psia $\pm$ 1% Sensitivity 0.04 volt/psia/volt excitation

#### LIQUID COOLANT VALVE SUPERVISORY CONTACTS

B	Monitor signal - coolant valve close
C	Input signal to coolant valve connections
D	Monitor signal - coolant valve open

## NASA INTERFACE CONNECTIONS

### OXYGEN PURGE VALVE SUPERVISORY CONTACTS

j Monitor Signal - O<sub>2</sub> Purge Valve Close  
k Input Signal to O<sub>2</sub> Purge Valve Connections  
m Monitor Signal - O<sub>2</sub> Purge Valve Open

### HYDROGEN PURGE VALVE SUPERVISORY CONTACTS

n Monitor Signal - H<sub>2</sub> Purge Valve Close  
p Input Signal to H<sub>2</sub> Purge Valve Connections  
q Monitor Signal - H<sub>2</sub> Purge Valve Open

### REACTANT INLET VALVE SUPERVISORY CONTACTS

k Monitor Signal - Reactant inlet valve close  
L Input signal to reactant inlet valve connection  
m Monitor Signal - Reactant inlet valve open

## J1005 FUEL CELL THERMOCOUPLE - INSIDE OF STACK

Pin	Polarity	Thermocouple No.	Location
A	Plus	1	In H <sub>2</sub> O Plate#45-Corner A (Cell #11)
B	Minus		
C	Plus	2	In H <sub>2</sub> O Plate#5 - Corner C (Cell #4)
D	Minus		
E	Plus	3	In H <sub>2</sub> O Plate #129-Corner D (Cell#33)
F	Minus		
G	Plus	4	Under Heat Exchanger
H	Minus		
J	Plus	5	Fan Outlet Return A-C Side
K	Minus		
L	Plus	6	Fan Outlet A-C Side
M	Minus		
N	Plus	7	Fan Outlet B-D Side
P	Minus		
R	Plus	8	In Transister holder between O <sub>2</sub> Plate #75 and H <sub>2</sub> Plate #72-Corner A (cell 19)
S	Minus		
T	Plus	9	In Transistor holder between O <sub>2</sub> Plate #67, and H <sub>2</sub> Plate#64-Corner D (cell 17)
U	Minus		

Note: Plus is iron - Yellow  
Minus is constantan - Red

# OPEN LOOP SYSTEM

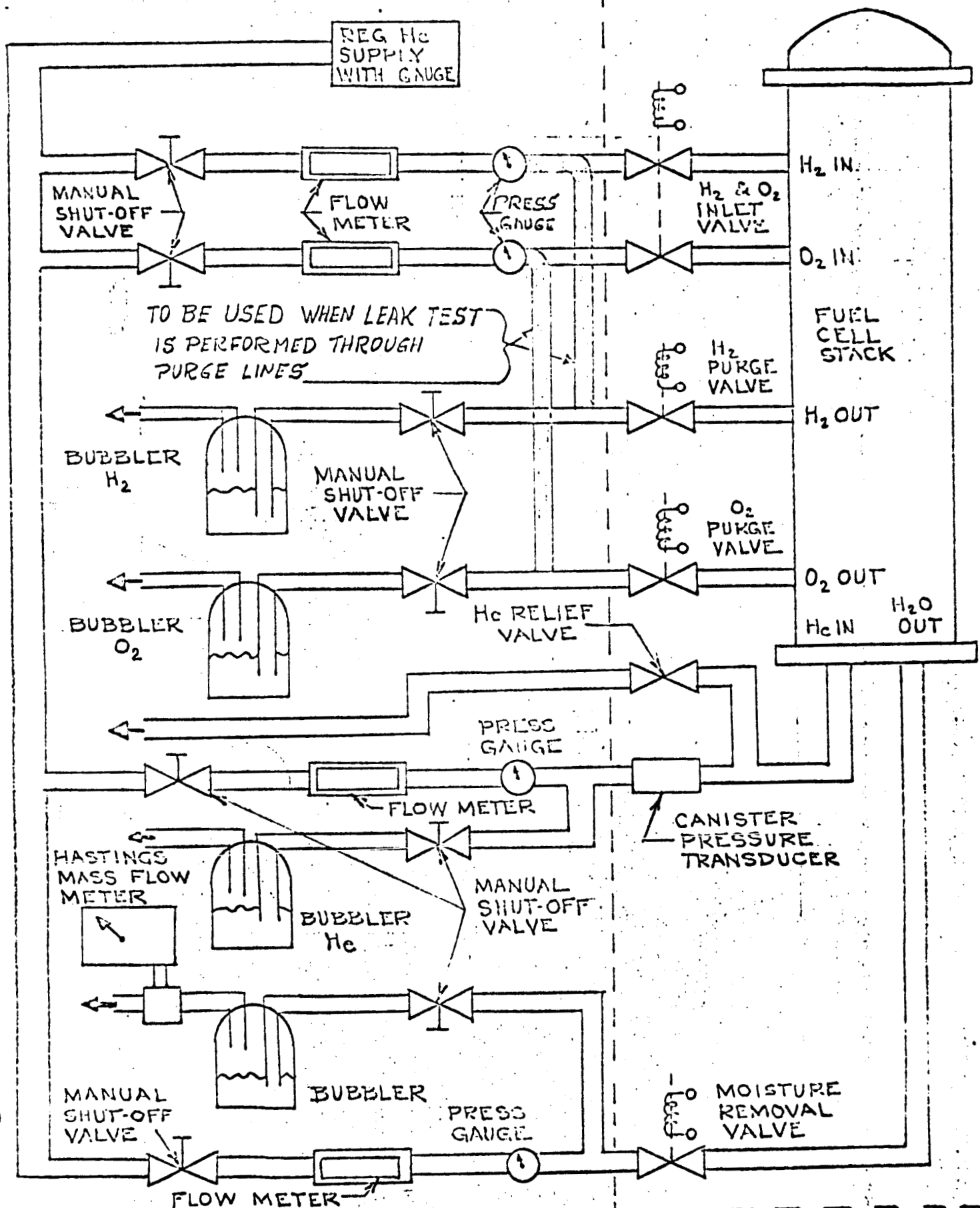


Figure 2-10

A regulated helium gas supply should be connected to each of the following lines on the fuel cell module.

- (1) Hydrogen gas inlet or hydrogen purge line
- (2) Oxygen gas inlet or oxygen purge line
- (3) Helium gas inlet
- (4) Moisture removal line (vapor vent)

The water vapor outlet from the fuel cell stack should be connected through a manual shut-off valve to a gas bubbler and a Hastings Mass Flowmeter for visual observation. The helium supply line to each of the above points should contain a manual shut-off valve and a pressure gauge downstream of the shut-off valve. A flow meter in each line is desirable but not mandatory. The hydrogen purge outlet and oxygen purge outlet should also discharge to the atmosphere through manual shut-off valves.

The detailed leak test procedure is as follows:

#### Cavity External Leak Check

- (1) Bleed the helium pressure within the fuel cell canister to 0 psig from the helium fill line.
- (2) Fill the reactant cavities with helium through the hydrogen and oxygen inlet lines or through the hydrogen and oxygen purge lines to 20 psig, and then depressurize to zero psig. Repeat.

CAUTION - The pressure should be increased or reduced gradually to insure that the maximum pressure difference between the hydrogen and oxygen pressure gauges is less than 2 psi.

- (3) Close the purge valves (only when reactant cavities are filled through oxygen and hydrogen inlet valves), open the moisture removal cavity valve and the vapor vent.
- (4) Apply helium pressure simultaneously to all three cavities until a pressure of 20 psig is reached, and allow to stand for ten (10) minutes. Observe the Caution note from Step 2.
- (5) Shut off the supply valves and record the three cavity pressures after allowing to stand for 15 minutes. A pressure drop is indicative of a leak into the canister interior or valve leakage. The purge valve leakage may be checked by connecting the purge outlet lines to a gas bubbler. Leakage into the canister may also be checked by putting a bubbler on the canister helium fill line.

#### Cavity Cross Leak Check

- (6) Open the helium supply valves and raise all cavity pressure back to an equalized pressure of approximately 20 psig and then close the hydrogen inlet or hydrogen purge valve and moisture removal valve.
- (7) Continue to raise the pressure on the oxygen cavity, through the oxygen inlet line or oxygen purge line until a pressure of 22 psig is reached. Close the manual oxygen inlet valve or oxygen purge valve. Allow to stand for 15 minutes and record the reactant cavity pressures. A simultaneous drop in the oxygen cavity pressure and rise in the hydrogen cavity pressure is indicative of a cross leak.

- (8) Open the moisture removal cavity valve and bleed the pressure gradually to zero (0) psig. Open the hydrogen and oxygen inlet valves and gradually pressurize both reactant gas cavities to thirty (30) psi differential across the water removal matrices. Leave the inlet or purge valves open. Allow to stand for 15 minutes and record the bubbles per minute observed in the vacuum vent line gas bubbler and record the Hastings Mass Flowmeter reading.
- (9) Gradually bleed down all helium pressure to zero (0) psig.

#### Canister Leak Check

- (10) Pressurize the fuel cell canister to 40 psi differential across the canister wall and open the purge lines and vacuum vent line from the vent graduate, to bubblers for visual observation. Disconnect the helium supply from the helium fill line thereby isolating the canister. Record canister pressure every half hour for twenty-four (24) hours along with room temperature. If at all possible, the room temperature should be held constant throughout the test.
- (11) Any observable pressure drops during the tests detailed in Steps 1 through 10 should be reported immediately to the Allis- Chalmers designated representative or site representative. Any flow rate reading in excess of 25 cc/hour observed during the leak test outlined in Step 8 should be reported immediately to Allis-Chalmers. If the pressure drop during the fuel cell canister leak test as detailed in Step 10 exceeds one (1) psi in the twenty-four (24) hour test period, notify Allis- Chalmers.



### 2.2.3 Heating Tape

To prevent the product water vapor from condensing in the water removal line, the line must be heated from the moisture removal valve at the bottom of the fuel cell module to the water recovery condenser. A section of heating tape is provided for this purpose. The tape is applied by removing its backing and sticking it directly to the water removal manifold and tubing. Cut the tape to the desired length and attach the input leads as per Figure 1-2A (Drawing No. 49-500-125-404). The A-C power input to the input leads on breaking tone is provided from a Variac (See Figure 3-4).

Upon initial startup, adjust the temperature of the water removal line to  $195 \pm 10^{\circ}\text{F}$  as follows:

- a. Attach a thermocouple to the tubing to read tubing temperature.
- b. Turn off 110 vac heater power "ON" after first assuring the Variac is set for minimum voltage.
- c. Gradually, raise the heating tape voltage by turning the Variac clockwise until the desired steady-state tubing temperature is reached.

## SECTION III

### INTEGRATED SYSTEM OPERATION

#### 3.1 GENERAL

This portion of the instruction manual covers the various operational modes of the complete fuel cell assembly, and includes the expected readings of all essential instrumentation. The description of the subsystems, and the thermocouple instrumentation are presented in Section IV of this manual.

#### 3.2 PRE-STARTUP CONDITIONING OF THE SYSTEM (See A-C control panel Figure 3-1 (49-400-317-401), Figure 3-2 (49-200-549-501), Figure 3-3 (49-500-167-401) and NASA Control Panel Figure 3-4 (49-400-349-402).

##### 3.2.1 Initial Conditions

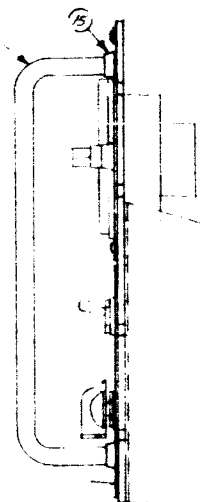
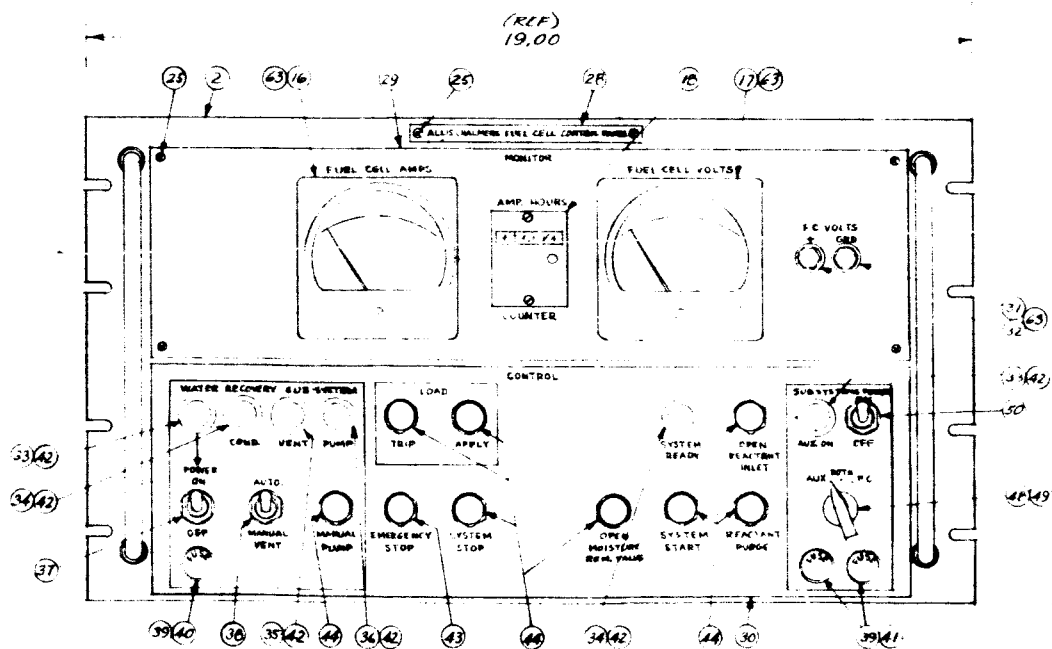
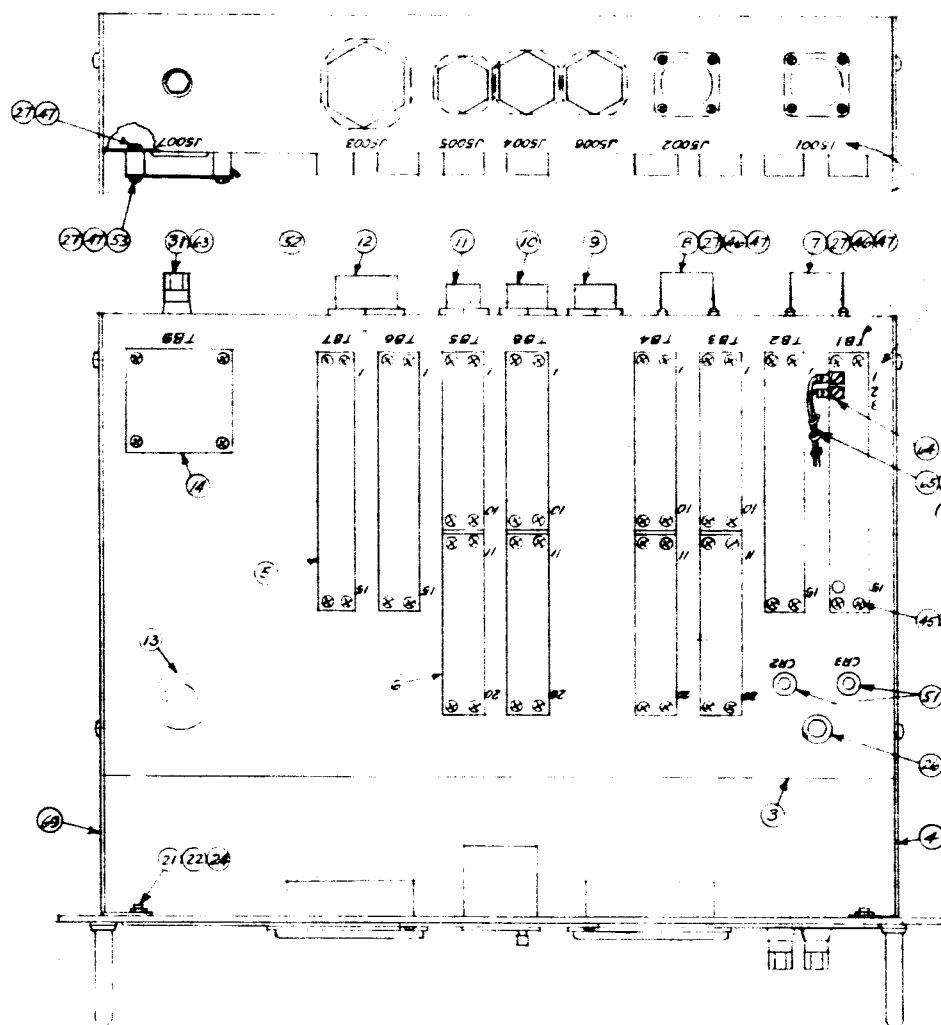
Pre-startup conditioning will be initiated with all valves closed, all switches off, and with system supplies as follows:

Hydrogen— per NASA specifications with minimum purity of 0.9998  
—Regulator closed off,

Oxygen — per NASA specification with minimum purity of 0.99995  
—Regulator closed off,

Helium — Commercial Grade  
—Regulator closed off,

Vacuum— normal vacuum of at least 1.5 psia applied to the moisture removal line.



NOTES:

1. FOR WIRING SCHEMATIC, SEE DWG. 49-500-167-401.
2. SOLDER WHERE REQUIRED. PER NRC-200-4.
3. HEAT SHRINK TUBING PER MSFC-PROC-273.
4. ROUTING, CRIMPING, LACING, ETC PER ELECTRICAL FABRICATION PROCEDURE MSFC-PROC-256.
5. CHECK FOR CONTINUITY PER SCHEMATIC.

01	4-16-66 B.P.D.	R6M
OPEN HOISTING BONE VALVE AND (1)		
HERE (1) (2) ADDED		
02	5-4-66 B.P.D.	R6M
ADDED (1) (2) ADDED		

USED ON P/L 49-200-549-501

COMMERCIAL PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b> RESEARCH 3341 WORKS		NAME <b>ALLIS-CHALMERS</b> <b>CONTROL PANEL ASSEMBLY PIX</b>	
UNLESS OTHERWISE SPECIFIED:		MATL.	
1- PLACE DEC. A 1- PLACE DEC. A 1- PLACE DEC. A ANGULAR	<input checked="" type="checkbox"/> MACHINED SURFACE TEXTURE	WT.	
DR 3-21-66 B.S.H. CH B.P.D. 4-16-66 AP R6M 4-16-66	SIMILAR TO SCALE HALF SIZE	SHEET 1 OF 1	
49-400-317-401 02		PART NO <b>49-400-317-401</b>	

REQ.	ITEM	DESCRIPTION	PART NUMBER
X	1	CONTROL PANEL ASSEMBLY PIX	49-400-317-401
1	2	PANEL	49-300-530-001
1	3	CHASSIS BASE	49-400-347-001
1	4	RIGHT HAND BRACKET	49-300-531-001
4	5	TERMINAL STRIP, CINCH-JONES #15-140Y	49-200-549-005
8	6	" " " #10-140Y	49-200-549-006
1	7	CONNECTOR, BENDIX	MS3102E18-3P
1	8	" "	MS3102E18-3PX
1	9	" "	MS3114E14-19SW
1	10	" "	MS3114E14-5S
1	11	" "	MS3114E12-10S
1	12	" "	MS3114E22-55SZ
1	13	GROMMET, .62 I.D., .125 O.D. RUBBER	49-200-549-013
1	14	AMPLIFIER CIRCUIT BOARD ASSEMBLY (P/L)	49-200-611-501
1	15	FERRULES	49-200-549-015
1	16	AMMETER, 0-100 AMP DC, 100 MV, WESTON MODEL 301, SQUARE.	49-200-549-016
1	17	VOLTMETER, 0-50 V DC, 1000 $\Omega$ /VOLT, WESTON MODEL 301, SQUARE.	49-200-549-017
1	18	IMPULSE COUNTER, 24V, 25 IMP./SEC, SODECO TYPE TCBBZGE.	49-200-549-018
X	19	SCHEMATIC WIRING DIAGRAM	49-500-167-401

2	20	HANDLE, PRECISION ALUM. FANDUCTS		49-200-549-020
10	21	MACHINE SCREW, #10-24 x .38, CROSS-REC, FLAT HD, CRES		MS24693-70
10	22	NUT, #10-24, HEX, CRES		MS35649-104
4	23	MACHINE SCREW, #10-24 x .38, CROSS-REC, PAN HD, CRES		MS51957-61
10	24	LOCKWASHER, #10, SPLIT TYPE, CRES		MS35338-81
6	25	MACHINE SCREW, #2-56 x .187, CROSS-REC, PAN HD, CRES		MS51957-2
1	26	GROMMET, .38 I.D., .62 O.D., RUBBER		49-200-549-026
16	27	MACHINE SCREW, #4-40 x .38, CROSS-REC, PAN HD, CRES		MS51957-15
1	28	NAMEPLATE		49-100-728-001
1	29	MONITOR SECTION PANEL		49-300-528-001
1	30	CONTROL " "		49-300-529-001
2	31	BINDING POST, BLACK, SUPERIOR DF30BC		49-200-549-031
1	32	" " RED " DF30RC		49-200-549-032
2	33	LENS, RED, PENNSYLVANIA TYPE B0110		49-200-549-033
2	34	" GREEN, " " B0112		49-200-549-034
1	35	" AMBER, " " B0111		49-200-549-035
1	36	" CLEAR, " " B0116		49-200-549-036
1	37	TOGGLE SWITCH, SPST, CUTLER-HAMMER B803, CIRC. ON-NONE-OFF		49-200-549-037
1	38	" " DPDT, " " B825, CIRC. ON-NONE-ON		49-200-549-038
3	39	FUSEHOLDER, BUSS TYPE HKP		49-200-549-039
1	40	FUSE, STD. TYPE 3AG, 32V, 4AMP, LITTEL FUSE SERIES 311000		49-200-549-040
2	41	" " " " 15AMP, " " "		49-200-549-041
6	42	HOUSING & LAMP COMBINATION, PENNSYLVANIA LAMP, 38, 50, 55, 60, 75, 100, 150, 200, 250, 300, 350, 400, 450, 500, 550, 600, 650, 700, 750, 800, 850, 900, 950, 1000, 1100, 1200, 1300, 1400, 1500, 1600, 1700, 1800, 1900, 2000, 2200, 2400, 2600, 2800, 3000, 3200, 3400, 3600, 3800, 4000, 4200, 4400, 4600, 4800, 5000, 5200, 5400, 5600, 5800, 6000, 6200, 6400, 6600, 6800, 7000, 7200, 7400, 7600, 7800, 8000, 8200, 8400, 8600, 8800, 9000, 9200, 9400, 9600, 9800, 10000		49-200-549-042







01 4-16-66 MA REM  
 02 4-2-66 MA REM  
 03 5-4-66 MA REM  
 04 5-7-66 REL. FA 0945-9

UNLESS OTHERWISE NOTED DIMENSIONS ARE IN INCHES & MACHINING TOLERANCES ARE			ANGLES ±	
DIMENSIONS	UP TO 8 INCL	OVER 8 TO 24 INCL	OVER 24	OVER 24
FRACTIONAL	±	±	±	±
DECIMAL	±	±	±	±

—Copyright — Property of  
 ALLIS-CHALMERS MFG. CO., Milwaukee, Wis.  
 RESEARCH 3341 DART

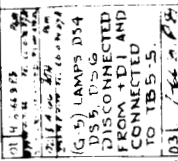
DR N DSH 4-16-66  
 CHS D.P.B. 4-16-66  
 TRD  
 APPRO RGM 4-21-66

# ALLIS-CHALMERS CONTROL PANEL ASSEMBLY P/L

FIGURE 3-2

SIMILAR TO SCALE 012

SHEET 2  
 OF 2  
 49-200-549-501





NASA  
115 VOLT  
AC (HOT)

S8 F8

S1

F1

DS1

TO NASA FCA  
VACUUM PUMP  
CONTROL

S2

F2

DS2

TO MOISTURE  
REMOVAL SYSTEM  
HEATER TAPE

NASA 115 VOLT AC (GROUND)

3 PHASE  
400 CYCLE  
200 VOLT  
FAN POWER

BA K1-1  
BB K1-2  
BC K1-3

FAN  
CURRENT

POWER

TO J1000  
TO J1000  
TO J1000  
TO J1000  
TO J1000  
TO J1000  
TO J1000

NASA  
(+) 28 VOLT  
D.C.

S10 F8

TO J3003-H  
WARM-UP  
HEATER  
THERMOSTAT S8

TO J3003-C

TO J3003-B  
WARM-UP  
HEATER  
THERMOSTAT S9

TO J3003-A

S9 FAN  
POWER

K1

DS6

TO J3000  
TO REACTANT  
TO J3001-K

VALVE  
CLOSED  
DS10

NASA  
(-) 28 VOLT D.C.

K4

NASA'S  
60 AMP  
POWER  
SUPPLY

(+) 28 VOLTS  
D.C.

S7 F7

K2-2

K2-1

TO J3003-K  
WARM-UP  
HEATER HR4  
(800 WATT)  
TO J3003-J

K2-4

K2-3

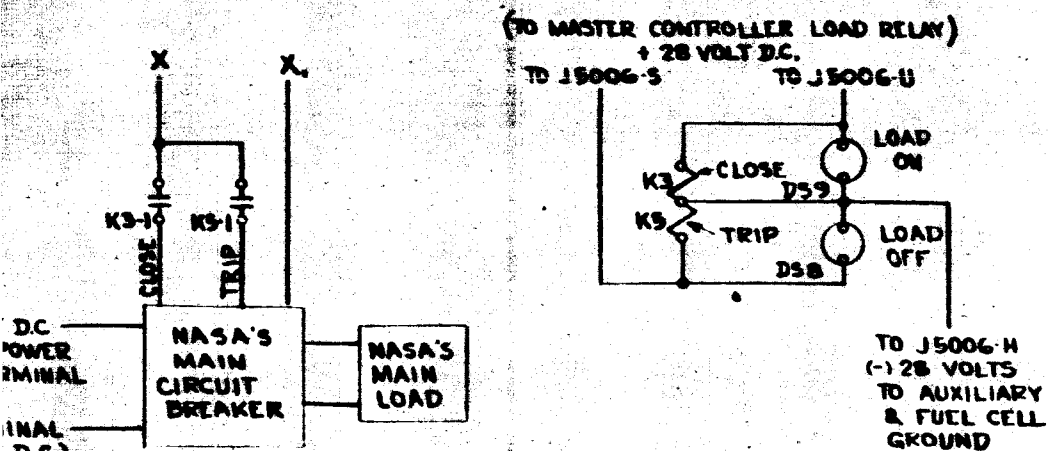
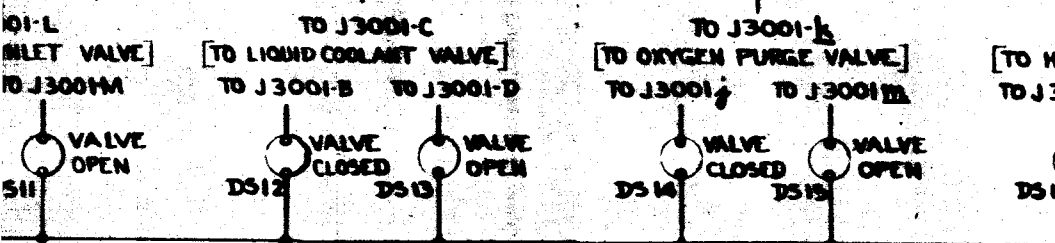
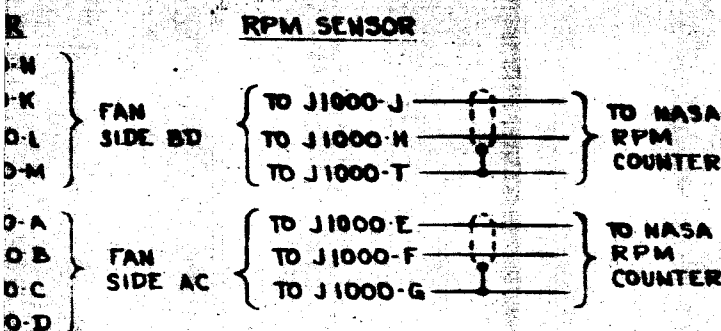
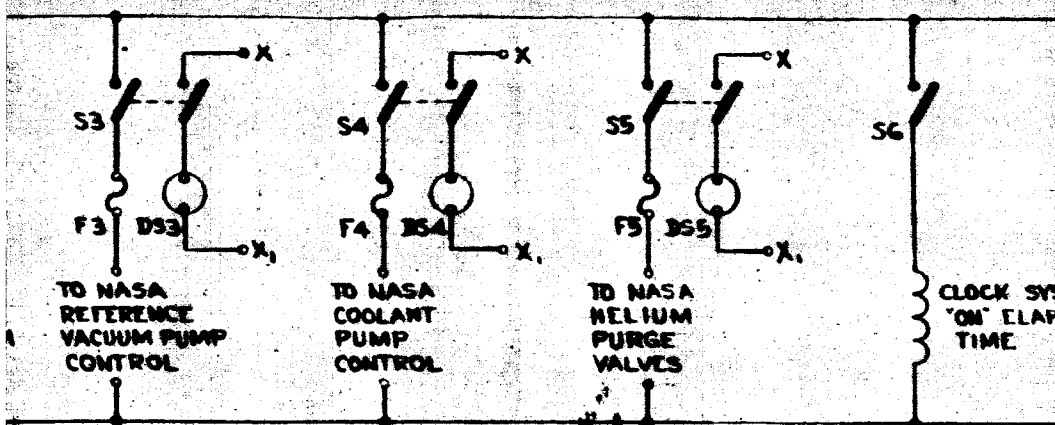
TO J3003-G  
WARM-UP  
HEATER HR3  
(800 WATT)  
TO J3003-E

(-) 28 VOLTS D.C.

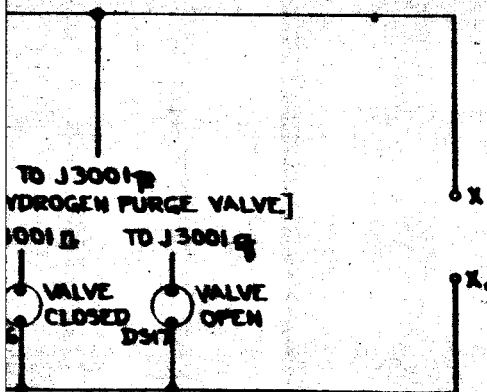
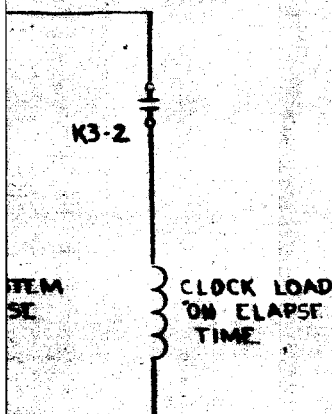
TO (+) 28 VOLT  
FUEL CELL  
OUTPUT TERMINAL

TO "B" TERMINAL  
(-) 28 VOLT  
OF SHUNT

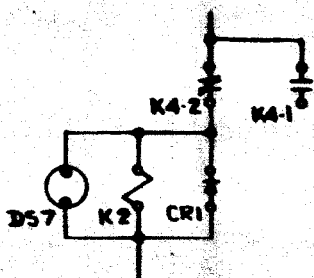
3-4-1



3-4-2



TO TEMPERATURE CONTROLLER RELAY J 5006-K +28 VOLTS D.C.



HIGHEST NO. USED

S-10

F-8

K-4

DS-17

REDUCED REPRODUCTION OF 49-400-349-402

CONFIDENTIAL - PROPERTY OF ALLIANCE-CHALLENGER REPR. CO.		NAME	
3341 MI PLANT		RECOMMENDED NASA AUXILIARY CONTROL	
UNLESS OTHERWISE SPECIFIED:		MATERIAL	
1 - PLANT USE &	✓		
2 - PLANT USE &			
3 - PLANT USE &			
APPROVAL & SIGNATURE			
DATE 6-3-66			
BY APP 6-3-66			
BY APP 6-3-66			
		FIG 3-4	

3-4-9

Liquid coolant supply available to the canister heat exchanger at a minimum rate of 150 pounds per hour .

110 volt, 60 cps power available at the NASA Test Control Panel

200 volt, 400 cps three-phase power available at the NASA Test Control Panel,

28 vdc auxiliary power available at the NASA Test Control Panel.

### 3.2.2 Cavity Helium Fill

Fill the three fuel cell cavities (hydrogen, oxygen, and water removal) with helium using the following procedure.

- a. Apply 28 vdc auxiliary power to the EMCS.
- b. Apply power to the helium purge (crossconnect) valves.
- c. Open the moisture removal (cavity) valve.
- d. Open the reactant inlet valve, or open purge valves if helium is supplied through purge lines.
- e. Open the helium supply valves to permit helium flow into the three cavities.
- f. Slowly raise the helium pressure to 5 psig. Allow 5 minutes for stabilization.
- g. Return all switches and valves (Steps b through f) to their original position.

### 3.2.3 Canister Helium Fill

To evacuate and fill the fuel cell canister with helium to 13 psig, proceed as follows:

- a. Open manual valving between the canister and the vacuum header, and evacuate the canister. Monitor the canister pressure.

- b. When the canister is completely evacuated, close the manual valve between the canister and the vacuum header.
- c. Raise the helium supply pressure to 30 psig.
- d. Crack open the manual valving between the helium supply and the canister, and gradually fill the canister to a pressure of approximately 13 psig (28 psia on the pressure transducer output).
- e. Repeat Steps a through d, then close the manual helium supply valve to the canister.

#### 3.2.4 Coolant Fans

Start the coolant fans, and check the fan voltage and current to determine if they are within acceptable limits.

#### 3.2.5 Liquid Coolant System

Start liquid coolant system, and check the coolant pressure and flow to insure it is within acceptable limits.

#### 3.2.6 Water Removal Heating Tape

Apply power to the WRS heating tape and check to insure the tubing rises to the proper temperature range.

### 3.3 SYSTEM STARTUP (See A-C control panel Figure 3-1 (49-400-317-401), Figure 3-2 (49-200-549-501), Figure 3-3 (49-500-167-401) and NASA Control Panel Figure 3-4 (49-400-349-402).

#### 3.3.1 Initial Conditions

After the conditions of the pre-startup are met, a normal startup will be initiated with the system supplies set as follows:



Hydrogen — Set the house regulator to deliver an outlet pressure of between 200 and 400 psia. The fuel cell hydrogen regulator is set to deliver a pressure output of approximately 37 psia over the full flow range (0.04 to 0.75 pounds per hour).

Oxygen — Set the house regulator to deliver a pressure of between 200 and 400 psia. The fuel cell oxygen regulator is set to deliver a pressure output of approximately 37 psia over the full flow range (0.32 to 6.0 pounds per hour).

Helium — Set the house regulator to deliver pressure of  $40 \pm 2$  psia, for the WRS pump.

Liquid Coolant — applied to the fuel cell as specified in Paragraph 3.2.1.

Vacuum — Applied as indicated in Paragraph 3.2.1

Auxiliary Power — Applied to EMCS

Fan Power — ON

WRS Heater Tape Power — ON

### 3.3.2 Startup Sequence

- a. Vent the moisture removal cavity into the vacuum system. Check the cavity pressure transducer to determine if the cavity has vented.
- b. Press the System Start switch. This applies power to stand-by heaters (50-watt each - 100-watt total) and turns ON ~~the~~ warm-up heaters (800 watt each - 1600 watt total) providing the 1600 watt switch is ON.

- c. Monitor the fuel cell stack warm-up until the operating temperature (185°F) is reached. The heater is controlled by a stack thermister to automatically cut-out at 185°F. The individual heaters are protected by over-temperature thermostats on the heater surfaces set for 250°F.

Note: For initial startup, allow the system to stabilize and monitor the performance of the temperature controller for one hour.

- d. After fuel cell has reached its operating temperature, the "System Ready Light" will come ON, and the reactant inlet valve can be opened to allow hydrogen and oxygen enter the reactant cavities.

Note: During initial checkouts and until sufficient testing has been conducted to demonstrate system operational response and stability during startup, it would be desirable to gradually buildup the reactant pressures in the hydrogen and oxygen cavities rather than applying full cavity pressures instantaneously. This may be accomplished by manually controlling pressure buildup using the reactant supply regulators.

1. With both the hydrogen and oxygen supply regulators shutoff, press the Open Reactant button and check that the reactant inlet valve is open (light indication).
  2. Slowly bring up the hydrogen and oxygen supply pressures, simultaneously using the supply regulators. Maintain hydrogen and oxygen pressures with 2 psi of each other during this procedure.
  3. When rated cell pressure (37 psia) is reached, the fuel cell regulators will maintain this pressure and the supply pressures can be set as indicated in Paragraph 3.3.1 "Initial Conditions"
- e. Check the hydrogen and oxygen pressure transducers. Both transducers should indicate  $37 \pm 1$  psia.

- f. Check that the water removal pipe is hot, approximately 190°F.
- g. Purge the reactant cavities three successive times to clean out the helium or any other inerts. This is accomplished by manually closing the reactant inlet supply valves feeding the reactant inlet valve (since reactant inlet valve cannot be closed manually after being open), then opening the hydrogen and oxygen purge valves. Then refill the reactant cavities. Repeat the purge cycle until a total of three purges have been accomplished. During these operations, be sure to follow the procedure in the note under Step d, on page 3-5.
- h. Check the individual cell voltages. Should any cell voltage be less than 1.0 volt, two additional manual purges are permitted to attempt bringing up the individual cell voltage.
- i. Monitor the Total Fuel Cell Voltage. The total open circuit voltage should read 35 volts, or more.
- j. Check the canister pressure to insure it is  $40 \pm 2$  psia. If it is not, fill to the required pressure.
- k. Check to insure that the WRS manual by-pass switch (Manual Vent-Auto) is in the "AUTO" position unless open loop tests are planned.

### 3.3.3

#### Switching Auxiliary Power To The Fuel Cell Output

After the fuel cell is warmed up to its operating temperature, the auxiliary power may be switched from external power to fuel cell power, or to Fuel Cell and Auxiliary power (both) in which case the EMCS will "Automatically Select" the source with the higher voltage.

3.4 OPERATION UNDER LOAD (See A-C control panel Figure 3-1 (49-400-317-401), Figure 3-2 (49-200-549-501), Figure 3-3 (49-500-167-401) and NASA Control Panel Figure 3-4 (49-400-349-402).

3.4.1 Initial Conditions

Initially, the system is at operating temperature with an open circuit voltage of 35 volts or more. Reactant supplies to the fuel cell are regulated to specified values, and all pressures, individual cell voltages, and temperatures are normal. The auxiliary dc power may be supplied from the fuel cell or from an external source, as desired, to meet specific test requirements. The RPM sensors must indicate that both coolant fans are running, and the liquid coolant pressure transducers must indicate that coolant is available.

CAUTION: The fuel cell should not be held in the hot, unloaded condition (total load less than 200 watts) for more than three hours at any one time. Refer to Paragraph 3.6.2 if longer standby periods are contemplated.

3.4.2 External Load Circuits

Since the external load circuits are being furnished by MSFC, their description does not form a part of this manual. A pilot relay must be provided to control the external load circuitry as the "Load" relay in the EMCS will only handle 1 ampere maximum.

3.4.3 Load Application

To apply load to the fuel cell system, push the "Apply Load" Button. The load "ON" signal will also initiate an automatic cavity purge cycle whenever it is actuated.

3.4.4 Normal Operating Parameters

After each incremental load change, the following operating parameters should be monitored to insure normal operation:

Module Temperature	—	190° - 200°F
Hydrogen Inlet Pressure	—	37 $\pm$ 1.0 psia
Oxygen Inlet Pressure	—	37 $\pm$ 1.0 psia
Moisture Removal Cavity Pres.	—	Varies automatically to maintain proper KOH concentration (See Figure 3-5)
Canister Helium Pressure	—	40 $\pm$ 2 psia
Load Current	—	Not to exceed 75 amperes continuous, and 150 amperes for 30 seconds.
Total Fuel Cell Voltage	—	See Figure 3-5
Reactant Gas Purge	—	
Frequency	—	Every 128 ampere-hours
Duration	—	6 seconds
Flow Rate:	—	H <sub>2</sub> 0.69 lb/hr.
		O <sub>2</sub> 5.47 lb/hr.

#### 3.4.5 Automatic Operation

The system will now operate automatically under load and during load changes as long as reactant gases are supplied and all critical parameters are under control. The individual cell voltages should be monitored at least every one-half hour. Should any individual cell voltage fall below 0.5 volts, it is recommended that a manual purge cycle of six seconds duration be initiated. If the individual cell voltage does not remain above 0.5 volts under load, a second manual purge cycle may be initiated. However, to avoid drying-out the cell, only two manual purges are recommended. If the cell voltage cannot be brought up by purging, a cell depressurization cycle should be performed as follows:

- a. Remove all load from the fuel cell system.

- b. Close the reactant inlet supply valves.
- c. Monitoring the reactant pressure transducers, apply a manual purge until the pressures drop to atmospheric.
- d. Close the purge valves, open the reactant inlet valve and reapply reactants to the fuel cell.
- e. Reapply a load "ON" signal to the EMCS.

NOTE: If after depressurization, the cell voltage fails to rise above 0.5 volt under load, shut down the fuel cell system and immediately notify Allis-Chalmers.

3.5 LOAD REMOVAL OPERATION (See A-C control panel Figure 3-1 (49-300-317-401), Figure 3-2 (49-200-549-501), Figure 3-3 (49-500-167-401) and NASA Control Panel Figure 3-4 (49-400-349-402). The "Drop" load or "Trip" load function can be initiated at any time during normal operation by pushing the "Trip Load" Button. This will result in the following:

- a. The external electrical load on the fuel cell system will be dropped.
- b. The moisture removal valve will close and remain closed until a load "ON" signal or a manual cavity valve signal is applied.

All other fuel cell system controls will remain operable and the external load may be reapplied by providing an "Apply Load" signal to the EMCS. Auxiliary DC power will continue to be supplied by the fuel cell if the system controls are so set.

As stated in Paragraph 3.4.1, the Load "OFF" condition should not be maintained for longer than 3 hours at any one time.

- 3.6 SYSTEM SHUTDOWN, STANDBY, AND STORAGE (See A-C control panel Figure 3-1 (49-400-317-401), Figure 3-2 (49-200-549-501), Figure 3-3 49-500-167-401) and NASA Control Panel Figure 3-4 (49-400-349-402).

3.6.1 System Shutdown

To shut down the fuel cell system, provide a system stop signal to the EMCS. The following actions will occur:

- a. The external electrical load on the fuel cell system will be dropped, if the fuel cell is in a load "ON" condition, and Step b Paragraph 3.5 will thereby take place.
- b. The reactant inlet valve will close, and shut off the supply of hydrogen and oxygen to the fuel cell.

CAUTION: Manually reopen the reactant inlet valve and reestablish the cavity pressures and maintain cavity pressure balance within 2 psi. The system can be held at open circuit for a short period (6 hours) in this manner or the cavities can be filled with helium in the case of prolonged shutdown.

- c. The System Stop signal also opens the circuit to the stack heater controller. The warmup and stand-by heaters cannot be energized until the System Start signal is again applied.

3.6.2 System Standby at Operating Temperature

After a System Stop signal has been applied and the actions indicated in Steps a through c above have occurred, it is possible to hold the system at approximately operating temperature by performing the following: Apply a "Start" signal which will close the heater circuit. The helium circulating fans must continue to operate. The fuel cell module may be held in this condition for a period not to exceed 6 hours. It will be necessary, however, to continuously monitor the fuel cell so that the following conditions are met.

- a. The fuel cell temperature as indicated by thermocouples should not exceed 200°F.
- b. All external auxiliary power must be maintained.
- c. Assure that water removal heater tape is energized and maintaining the water removal piping temperature. (This is to allow immediate operation when desired instead of requiring a warm-up period).

### 3.6.3 Prolonged Shutdown

For any period exceeding 6 hours, it will be necessary to complete system shutdown, and to flood all the cavities with helium according to the following:

- a. Shut down the helium circulating fans.
- b. Close any manual valves in the hydrogen and oxygen supply lines to the fuel cell module.
- c. Apply a manual purge signal until the hydrogen and oxygen cavities are at atmospheric pressure.
- d. Close the purge valves.
- e. Perform a cavity helium fill by following Steps b through g of Paragraph 3.2.2.
- f. Check the helium pressure in the fuel cell canister; if it is less than 13 psig follow Steps c and d, Paragraph 3.2.3 to bring the pressure back up.
- g. Switch off all power, and pumps which are used in the test setup.

The system is now secure and may be left inoperative for a prolonged period as long as helium pressure is maintained in the cavities and canister. It is recommended that these pressures be monitored once every hour for the first 24 hours and once every 8 hours thereafter. The



helium pressure should be brought back to nominal values when the following minimums are reached:

- a. The canister pressure drops to 8 psig.
- b. The three cavity pressures drop to 3 psig.

#### 3.6.4 System Storage

The fuel cell stack may be stored indefinitely in its shipping container in a protected area with an ambient temperature of 49°F or higher.

All other components may be stored in protected, but unheated, areas, provided adequate steps are taken to prevent condensation on the components.

### 3.7 ABNORMAL OPERATION AND EMERGENCY SHUTDOWN

Abnormal operation is defined as any operating condition in which one or more of the critical control parameters exceeds its normal control range. The critical control parameters, their normal range and action to be taken if the control band is exceeded, is presented briefly in this section.

#### 3.7.1 Stack Temperature

The normal control range is  $195^{\circ} \pm 5^{\circ}\text{F}$  as sensed by the thermister located between plates 80 and 83 on the fuel cell stack. A spare thermister is provided in the same block.

- a. Over-Temperature — Over-temperature may be detected by monitoring the thermocouples that form part of the test instrumentation. On an over-temperature indication, check for the following:

1. loss of liquid coolant,
2. loss of helium circulating fans,
3. loss of helium from the canister.

The fuel cell should be shut down immediately if a stack temperature of 220°F is reached. Below this temperature, the fuel cell may be operated up to 3 hours while the cause of the over-temperature is being investigated.

- b. Under-Temperature — Under-temperature operation is indicated by a gradual deterioration of cell performance under load, but represents no immediate safety or performance problem. It may be caused by overcooling by the liquid coolant system, or under-heating when operating at light loads in cold environments. To determine the cause, the liquid Coolant Valve should be closed. Monitor the stack temperature for effect. If the temperature begins to rise, a fault exists somewhere in the cooling system. If the temperature remains low, the fault is probably in the heaters or heater controls; this should be checked with the fuel cell system shutdown.

### 3.7.2 Total Fuel Cell Voltage

The normal fuel cell voltage will vary with load, as shown in Figure 3-5 and with fuel cell load life. The operator should be alert primarily for voltage changes during operation at constant load. Should the total voltage change 1/2 volt in 2 hours under a constant load condition, a complete survey of the individual cell voltages should be made.

A decreasing total cell voltage under constant load conditions may be caused by several factors such as:

- a. Accumulation of inerts in the cavities,
- b. Decrease in reactant supply,
- c. Improper operation of the water removal subsystem,
- d. Fuel cell stack temperature too high or too low. (See Paragraph 3.7.1)

Upon detection of an abnormal voltage output drop, the external fuel cell load should be removed, and the open circuit voltage on the individual cells should be checked.

- a. Accumulation of Inerts — If the voltage change is due to an accumulation of inerts within one or more of the individual cells, the total voltage will usually show a gradual decrease over a period of hours. The operator may perform two manual purge cycles to bring all the cell voltage to a minimum of 1.0 volts open circuit.
- b. Reactant Cavity Pressures — The cavity pressure regulators are set to maintain the hydrogen and oxygen cavities at  $37 \pm 1$  psia over the fuel flow range. Each cavity pressure is monitored by a pressure switch set to close the reactant inlet valve and to drop the main fuel cell load at a pressure of  $44.5 \pm 0.5$  psia.

Should the reactant inlet valve close during the operation, the load should be removed and the reactant cavity pressures should be checked immediately. If high reactant inlet pressure is indicated as the cause of the inlet valve closure then the cavities should be manually purged until atmospheric pressure is attained.

Close the manual valves in the reactant inlet lines and flood the cavities with helium to 5 psig. Then check the operation of the reactant gas pressure regulators before restarting the system.

- c. Moisture Removal - Moisture removal subsystem malfunctions

may result in a decrease in cell voltage either due to excessive wetting of the cell or excessive cell drying. If the moisture removal valve stays closed, the cell will accumulate too much moisture and the electrolyte will become diluted. To determine if this is the problem, reduce the load to reduce the probability of causing permanent damage and take readings of the cavity pressure and stack temperature. Compare these readings to the allowable operating range as shown on Figure 3-6. Should over wetting prove to be the cause of decrease in cell voltage, the moisture removal valve should be opened and the change in cavity pressure observed. If the malfunction appears to be on the cavity controller, the fuel cell should be shut down and Allis-Chalmers should be notified immediately.

Excessive cell drying can occur if the moisture removal valve is open too long. This condition can be corrected by a period of low load operation with the moisture removal valve closed. Excessive drying can be determined by observing the action of the moisture removal valve, monitoring the cavity pressure, and observing cell voltage for a period of 1 to 2 hours while operating at low load (20 amp). The information obtained should be compared with Figure 3-6. Should cell drying be indicated, the Allis-Chalmers Project Manager or his site representative should be contracted for a detailed corrective procedure.

### 3.7.3 Canister Pressure

Canister over-pressure protection is provided by a relief valve set to open at a pressure of 45 psia. Canister pressure is monitored by a pressure transducer. Since canister pressure is controlled manually,

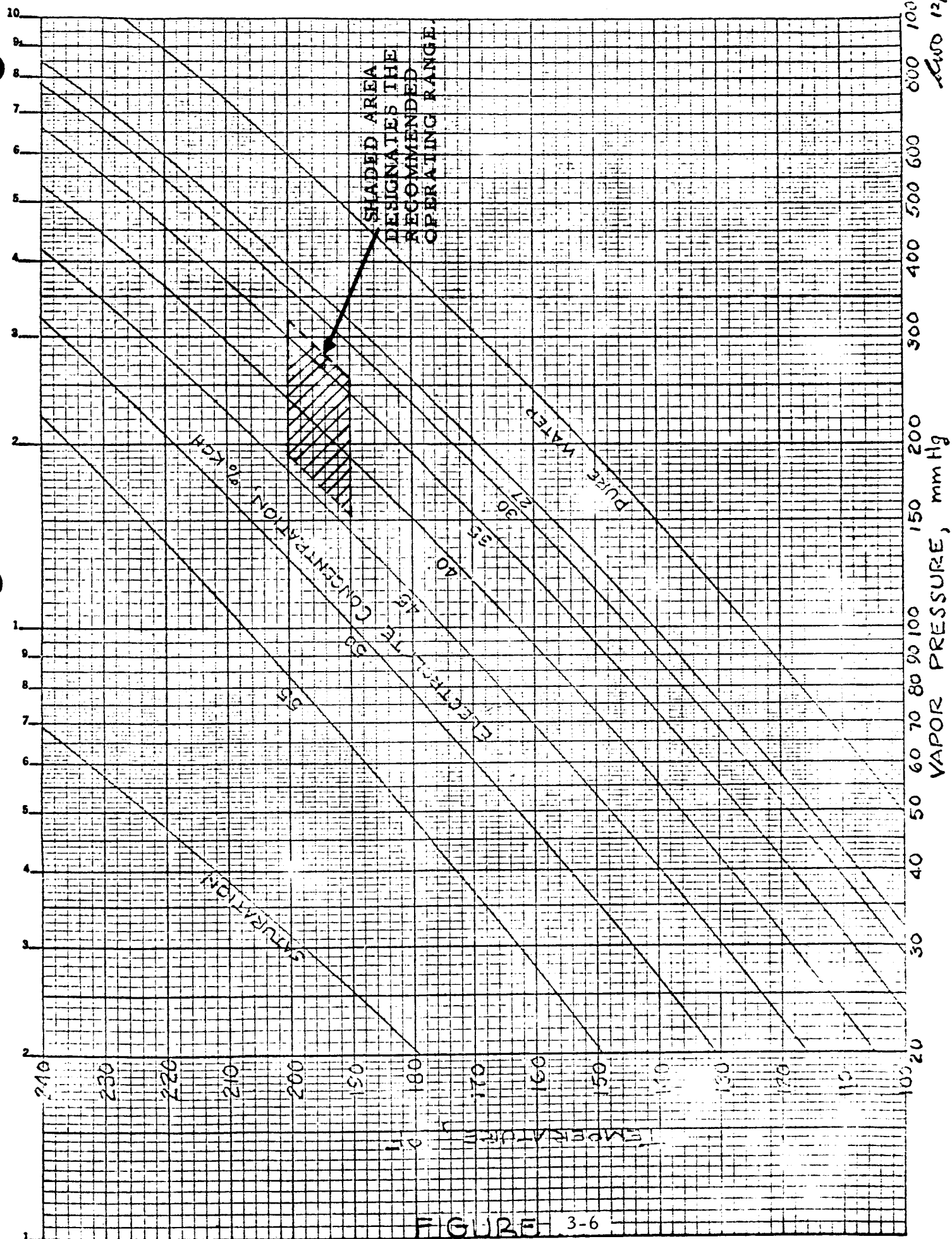


FIGURE 3-6

AWD 12/1/73

it should be monitored periodically and adjusted as necessary to maintain a pressure of  $40 \pm 1$  psia during normal operation.

#### 3.7.4 Loss of Auxiliary Power

If loss of either the 28 vdc or the auxiliary power 200 v, 400-cycle fan power is indicated, the fuel cell should be shut down immediately. Since the reactant inlet valve requires power to close, it is possible to lose auxiliary power and have the reactant inlet valve remain open. Therefore, the first step after loss of auxiliary power is to close the manual valves in the reactant supply lines, and to verify the position of the reactant inlet valve.

#### 3.7.5 Emergency Stop

An Emergency Stop signal will perform the following functions:

- a. Close the reactant inlet valve,
- b. Drop the external load,
- c. Interrupt the heater circuit,
- d. Close the moisture removal valve,

The Emergency Stop signal should be used to shut down the fuel cell system any time a malfunction is suspected or a safety problem exists. The Emergency Stop performs essentially the same functions as the System Stop, but it by-passes some of the control system interlocks; its use, therefore, should be limited to emergency conditions. The normal startup procedure should be utilized following an emergency stop.

## SECTION IV

### SUBSYSTEM DESCRIPTION

#### 4.1 FUEL CELL STACK SUBSYSTEM (FCSS)

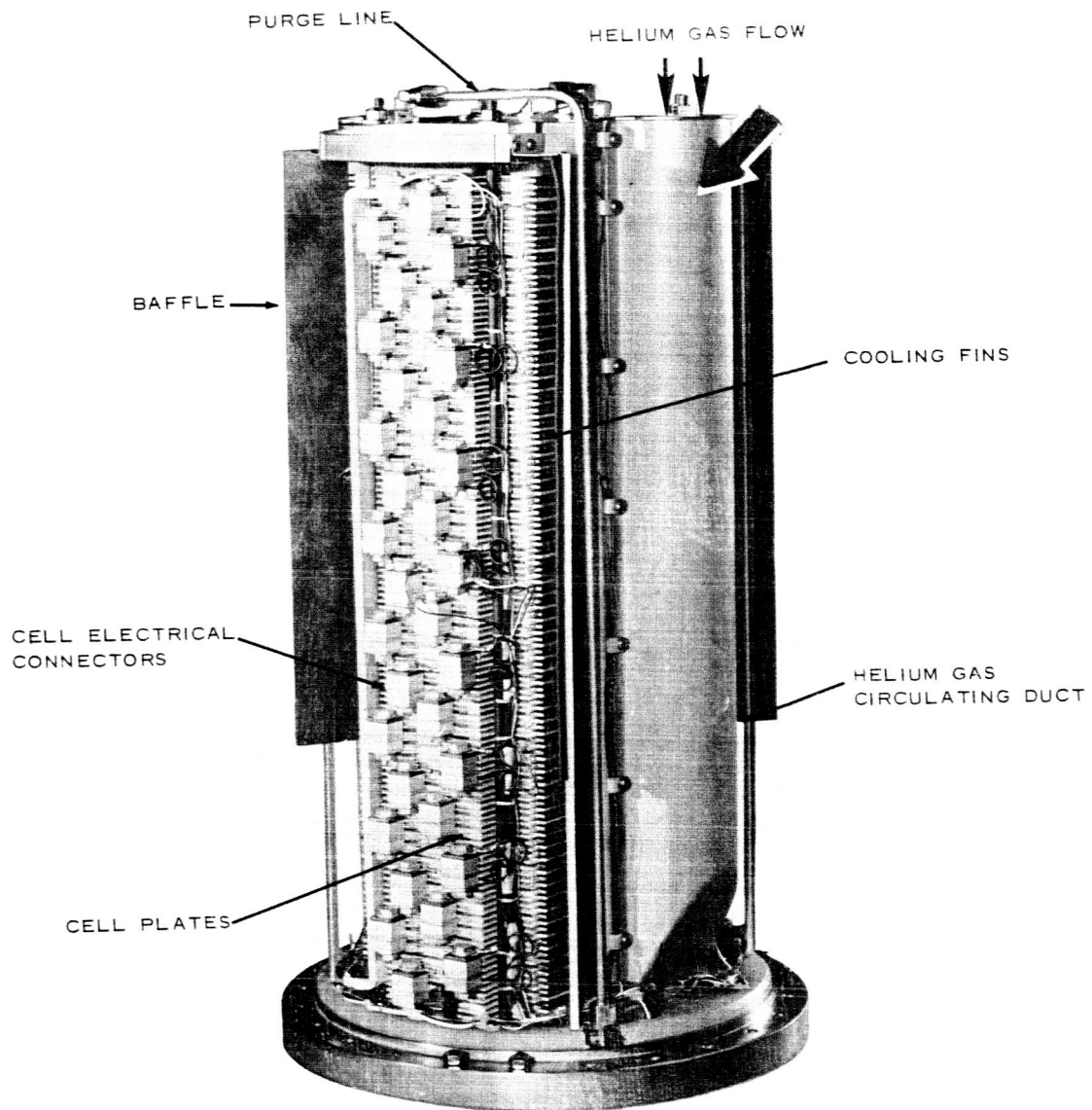
The Fuel Cell Stack Subsystem (FCSS) is the subassembly that converts chemical energy to electrical energy, producing heat and potable water as by-products of the reaction.

##### 4.1.1 Description

The FCSS consists of 33, two-cell parallel connected, sections. The 33 sections are electrically connected in series to produce a nominal 29 vdc output capable of supplying 2,000 watts. Each cell has an effective electrochemical reaction area of 0.20 feet<sup>2</sup>. An asbestos capillary matrix facing the hydrogen electrode provides for removal of the by-product water. The stack subassembly is shown in Figure 4-1. An exploded view of the cell construction is shown in Figure 4-2.

The construction consists of individual cells stacked in a manner that results in common oxygen and water plates for adjacent cells. The electrical connections are provided at both ends of the hydrogen and oxygen plates to minimize the voltage drop due to plate and connector resistance. In addition, electrical connections are silver or gold plated to minimize contact resistance.

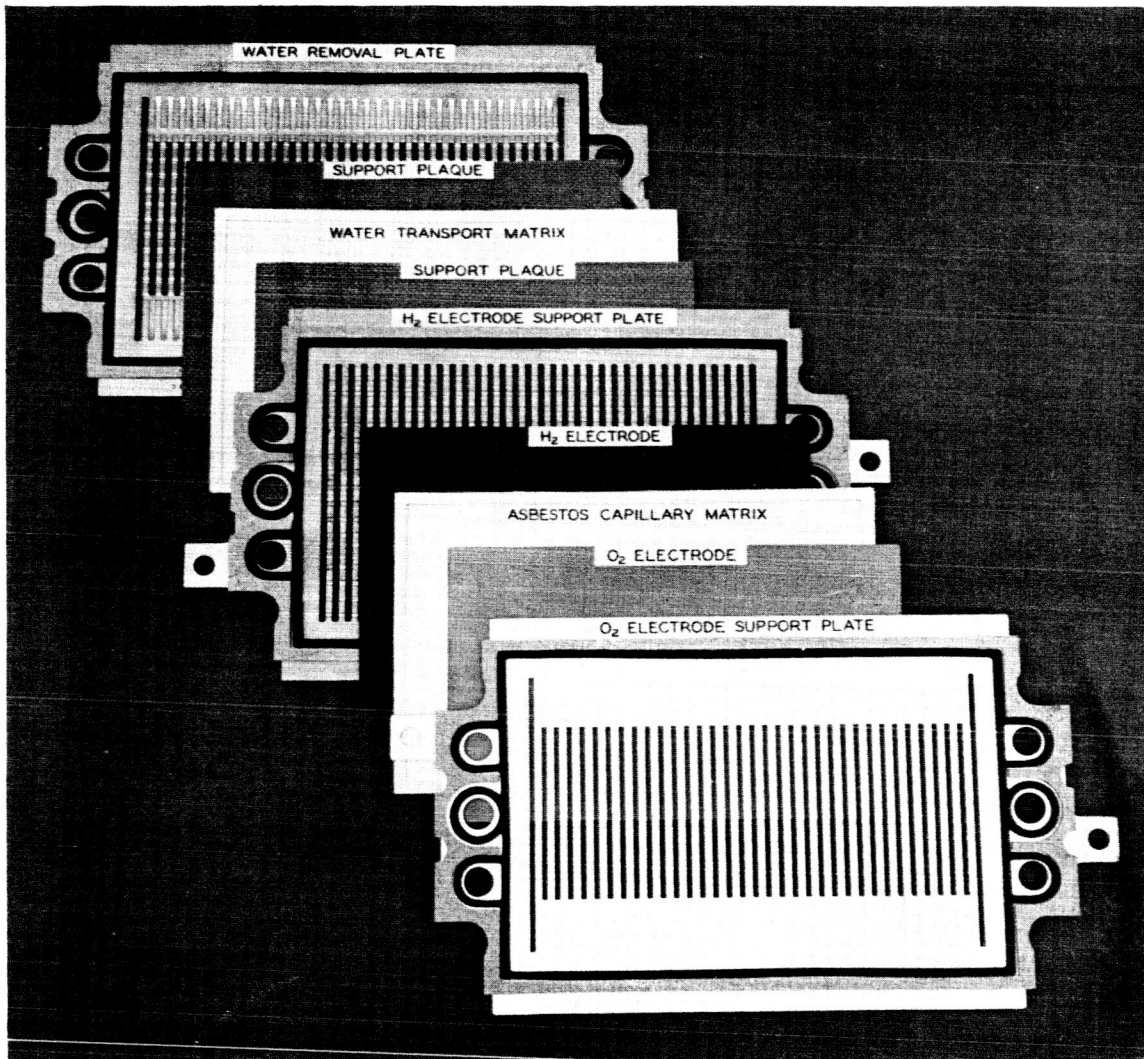
The reactants are supplied to the cell plates, and the by-product water is removed, through the main manifolds formed by holes and seals at the end of the plates. The reactants flow from the main manifold holes through electro-discharge machined (EDM) slots in the plates to grooved passages



**FUEL CELL STACK.** The stack, the power producing element of the system, consists of 33 two-cell sections. Each cell has an effective electro-chemical reaction area of 0.2 square foot. By-product water is removed from the stack in the vapor state using a simple and characteristically stable static moisture removal technique. Plastic ducts (arrow) direct coolant gas over the cell edges for thermal conditioning.

Figure 4-1





**CELL CONSTRUCTION.** The cell consists of two porous electrodes separated by an asbestos capillary matrix, which holds the aqueous potassium hydroxide (KOH) electrolyte. The electrode support plates provide passageways for distributing the reactants to the cell, and serve as current collectors and terminals for the electrodes. Product water is removed via the water transport matrix and the water removal plate as water vapor. When assembled, the plates extend slightly beyond the cell, thus serving as cooling fins for removing waste heat.

Figure 4-2

on the plate surfaces, providing uniform distribution of the reactants over the electrode surface. By-product water is removed through similar passages and slots.

Heat from the cell reactions is removed by conduction through the metal sections of the cell plates to cooling fins located on two sides of the plates, and with the by-product water vapor.

The cell plates are made of a high thermal conductivity magnesium alloy, and are electroless nickel plated; the oxygen plates have an additional plating of gold.

The seal for containing the reactants and by-product water vapor in the cells is formed by compressing the KOH impregnated asbestos capillary matrix between the plates. A elastomer seal is used to contain the liquid KOH and to contain the reactants and water vapor in the common manifold passageways. An insulating spacer material, 0.065 inch thick, surrounds and contains the elastomer seal. This spacer controls the compression on the cells, moisture removal matrix, and the elastomer seals.

The cells are sandwiched between thermal insulation and end plates and are held in compression by tie rods.

#### 4.2 REACTANT CONTROL AND CONDITIONING SUBSYSTEM (RCCS)

The Reactant Control and Conditioning Subsystem (RCCS) has several functions. It maintains constant hydrogen and oxygen (reactant) pressures within the Fuel Cell Stack Subassembly (FCSS), maintains reactant flow rates as demanded by the FCSS, provides a means of purging the FCSS of inert gas contaminants, and prevents over-pressurization of the FCSS. The reactants are supplied to the RCCS at pressures between

200 and 400 psia, at temperatures between 0 and 150°F, and with purity requirements as noted in Paragraph 3.2.1.

#### 4.2.1 Description

The RCCS consists of three solenoid valves: two pressure regulators, two absolute pressure switches, two absolute pressure transducers, ~~two reactant pre-heaters~~, and the associated plumbing. These components are functionally arranged as indicated schematically in Figure 4-3. The inlet solenoid valve isolates the Fuel Cell Stack from the reactant supply during non-operation. The pressure regulators maintain constant reactant pressures within the FCSS with a tolerance of  $\pm 1$  psi at a set pressure of 37.0 psia. The absolute pressure switches prevent damaging pressure extremes within the FCSS in order to maintain reactant purity within the cell. The pressure transducers monitor hydrogen and oxygen pressures to the FCSS. The Inlet solenoid valve is equipped with supervisory contacts from which valve position can be monitored. The reactant pre-heaters are used to bring both the oxygen and hydrogen up to operating temperature.

### 4.3 THERMAL CONTROL AND CONDITIONING SUBSYSTEM (TCCS)

The TCCS is designed to perform the following function:

- a. Bring the fuel cell from 0°F up to 185°F within one hour,
- b. Maintain fuel cell operating temperatures at  $195 \pm 5$ °F throughout the operating period,
- c. Monitor the canister pressure and provide canister over pressure protection.

#### 4.3.1 Description

A schematic representation of the TCCS is shown in Figure 4-4, and Figure 4-5. Functions of the major TCCS components are as follows:

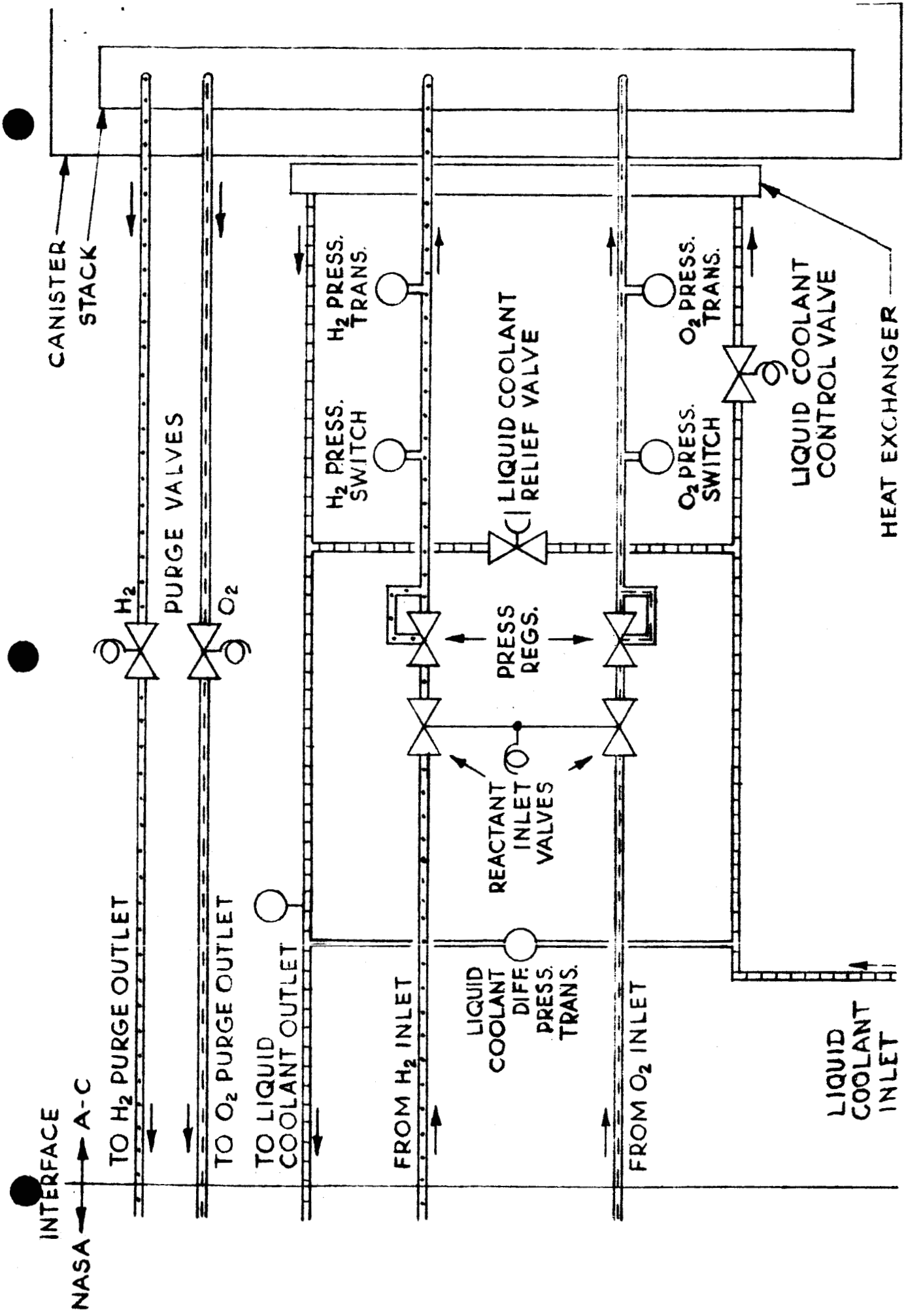


FIG. 4-3 REACTANT CONDITIONING CONTROL SUBSYSTEM(RCCS)

# THERMAL CONDITIONING AND CONTROL SUBSYSTEM (TCCS)

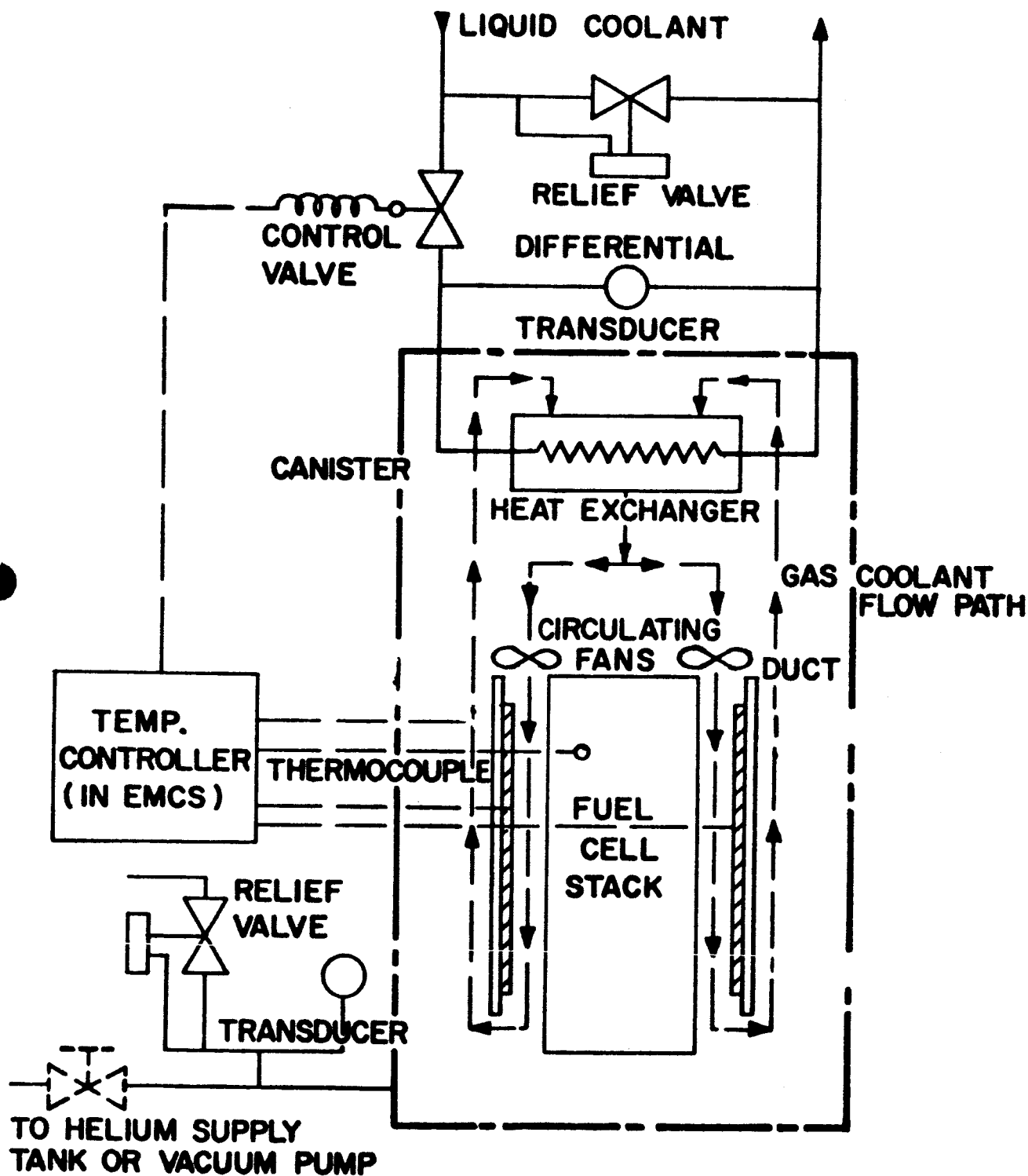
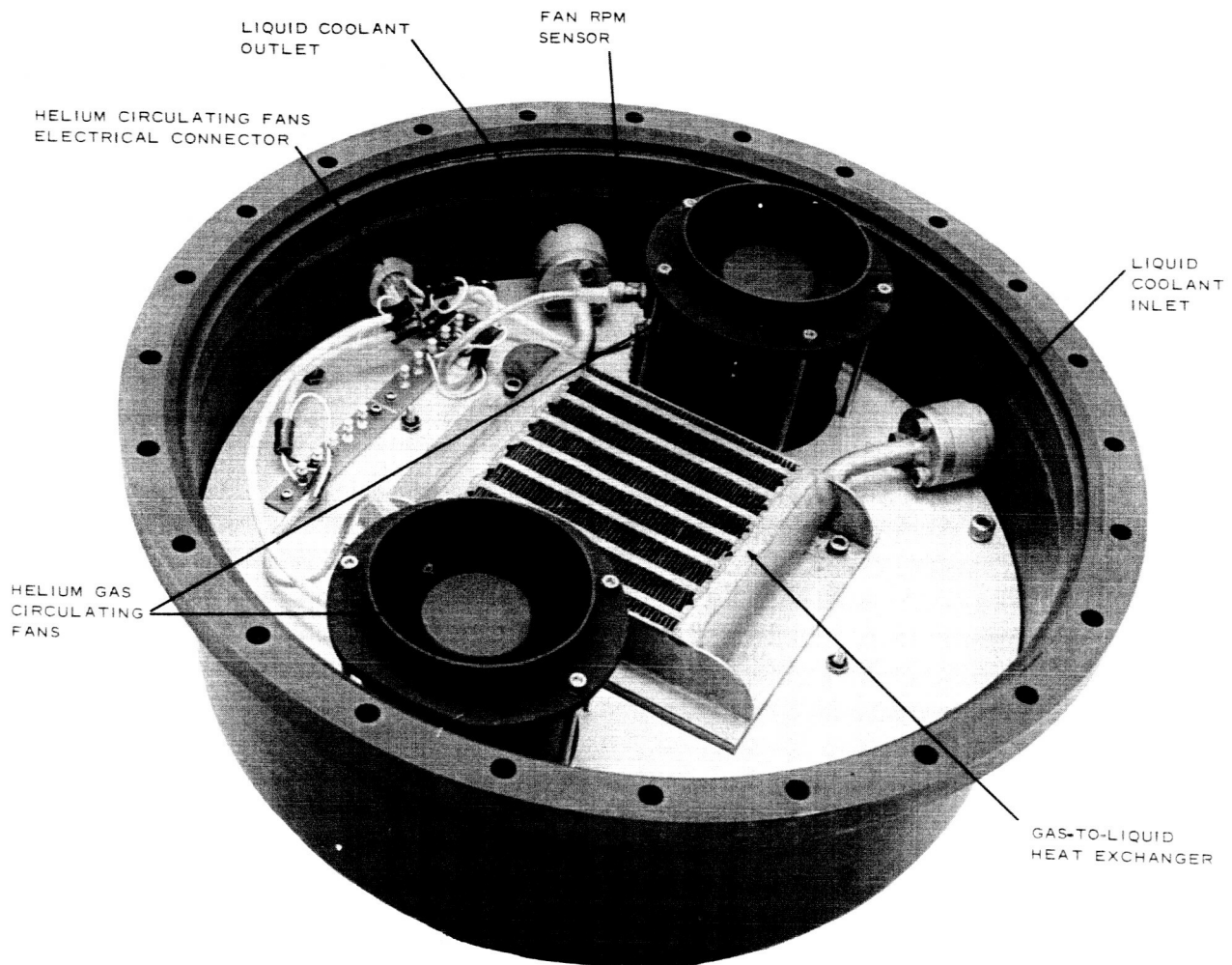


FIGURE 4-4



**THERMAL CONTROL AND CONDITIONING.** Circulating fans and a gas-to-liquid heat exchanger, mounted in the canister dome, provide thermal conditioning for the stack. The fans circulate helium gas through distribution ducts, over protruding cell fins, and then through the heat exchanger. Stack temperature is maintained by controlling the flow of liquid coolant through the heat exchanger.

Figure 4-5

- a. Warmup — The warm-up heaters (two 800-watt heaters controlled by the EMCS, powered by NASA) and the stand-by heaters (two 50-watt heaters controlled by the EMCS, powered by the EMCS) heat the helium gas in the canister, and the helium gas is circulated around the FCSS by the fans. This operation transfers thermal energy from the warm-up heaters and stand-by heaters to the FCSS and brings it up to operating temperature. A thermostat senses stand-by heater surface temperature and controls the power input to these heaters, through a solid state switch, such that the surface temperature of the stand-by heaters never exceeds 250°F. The temperature controller senses FCSS temperature when that temperature reaches 185°F, power to the warm-up heaters stand-by heaters is shut off. NOTE: Both type heaters are parallel controlled by the EMCS. However, the 800-watt heater surface temperature can be controlled by NASA through the 800-watt thermostats.
- b. Operating — After the FCSS reaches a temperature of 195°F, the TCCS must remove the excess heat generated by fuel cell operation. To accomplish this, the temperature controller senses FCSS temperature through a thermistor and controls the flow of liquid coolant through the heat exchanger, shown in Figure 4-4 by opening or closing the liquid coolant control valve. The liquid coolant control valve opens when FCSS temperature is maintained at 195<sup>±</sup> 5°F. The helium gas in the canister is circulated around the FCSS and through the heat exchanger by continuous operating fans. The helium gas picks up heat from the FCSS and transfers it to the liquid coolant in the heat exchanger. The liquid coolant, in turn, transports the heat out of the canister to the liquid coolant sink where it is dissipated.

If the level of fuel cell operation is such that FCSS temperature drops below 185°F, the TCCS will go into the warm-up operation to maintain operating temperature. If one desires, the 800-watt heaters can be shut off and additional heating will be accomplished by the 50-watt heaters.

- c. Canister Pressurization — The fuel cell canister is evacuated and filled with helium gas to a pressure of  $40 \pm 2$  psia. The operation capacity of the TCCS is dependent upon the characteristics of this helium gas, which is used as a secondary coolant in the TCCS.

Therefore, a method is provided for maintaining canister pressurization as shown in Figures 4-4 and 4-5. A relief valve provides over-pressure protection and a pressure transducer provides a method of monitoring. A manually operated valve, located on the opposite side of the system interface, is used for evacuation and fill.

#### 4.4 MOISTURE REMOVAL SUBSYSTEM (MRS)

The Moisture Removal Subsystem (MRS) maintains water removal cavity pressure (utilized in the Static Moisture Removal Concept) equivalent to the water vapor pressure exerted by the most concentrated KOH solution desired in the FCSS water cavity. The regulator opens the valve and vents water vapor from the fuel cell if the cavity pressure exceeds this value.

##### 4.4.1 Description

The design uses a solenoid operated water cavity valve actuated by an electronic temperature compensated controller that receives its signal from a pressure transducer. A schematic diagram of the subsystem is shown in Figure 4-6.



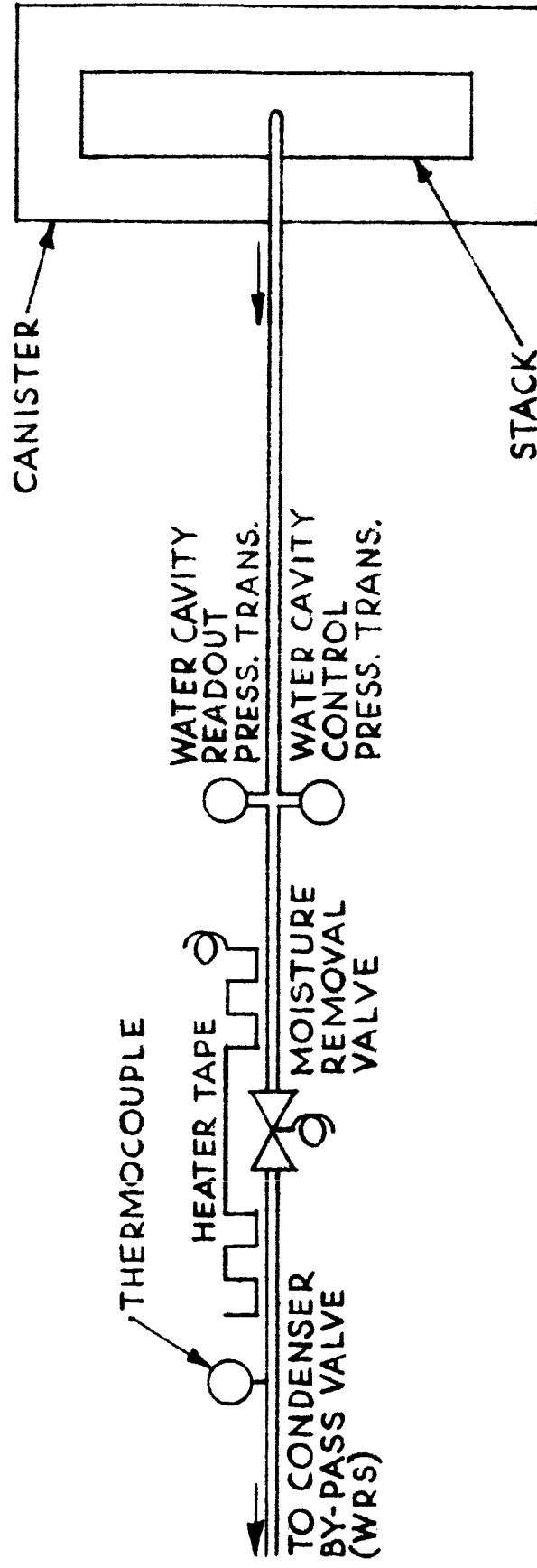


FIG.4-6 MOISTURE REMOVAL SUBSYSTEM (MRS)

#### 4.5

### ELECTRICAL MONITORING AND CONTROL SUBSYSTEM (EMCS)

The EMCS consists of a power supply, a temperature controller, a purge controller, a moisture removal controller, and a master controller. These controllers are built-up on a number of printed circuit boards cards, and all are mounted in a single control box. The circuits for these various controllers have been designed to use static control throughout, with the exception of the output devices which are miniature-type latching relays. This arrangement permits isolation of the valve control solenoids from their electronic circuits. A brief functional description of each of the controllers follows in the succeeding paragraphs.

#### 4.5.1 Control Power Supply —

Figure 4-7 (Drawing No. 49-200-493-501),  
Figure 4-8 (Drawing No. 49-300-542-401),  
Figure 4-9 (Drawing No. 49-300-496-401),  
Figure 4-10 (Drawing No. 49-200-492-501),  
Figure 4-11 (Drawing No. 49-200-484-401),  
Figure 4-12 (Drawing No. 49-400-300-401),

The control power supply provides up to 10 watts of dc. power at the following voltages:  $+11.2 \text{ v} \pm 0.5\%$  and  $-5.62 \text{ v} \pm 0.5\%$ . These voltages are

## CIRCUIT COMPONENT LIST FOR DRAWING

49-300-501

PIX

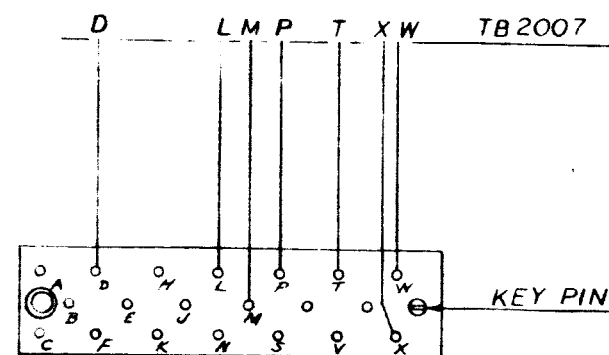
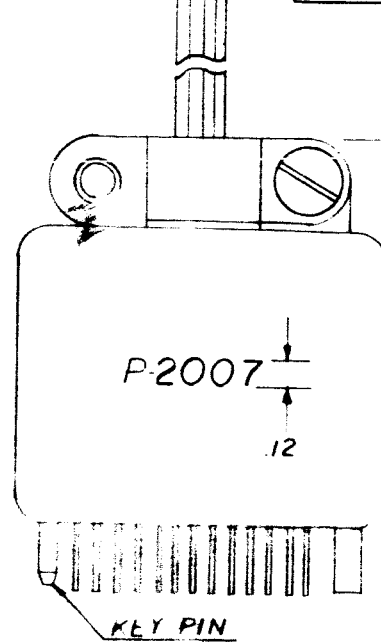
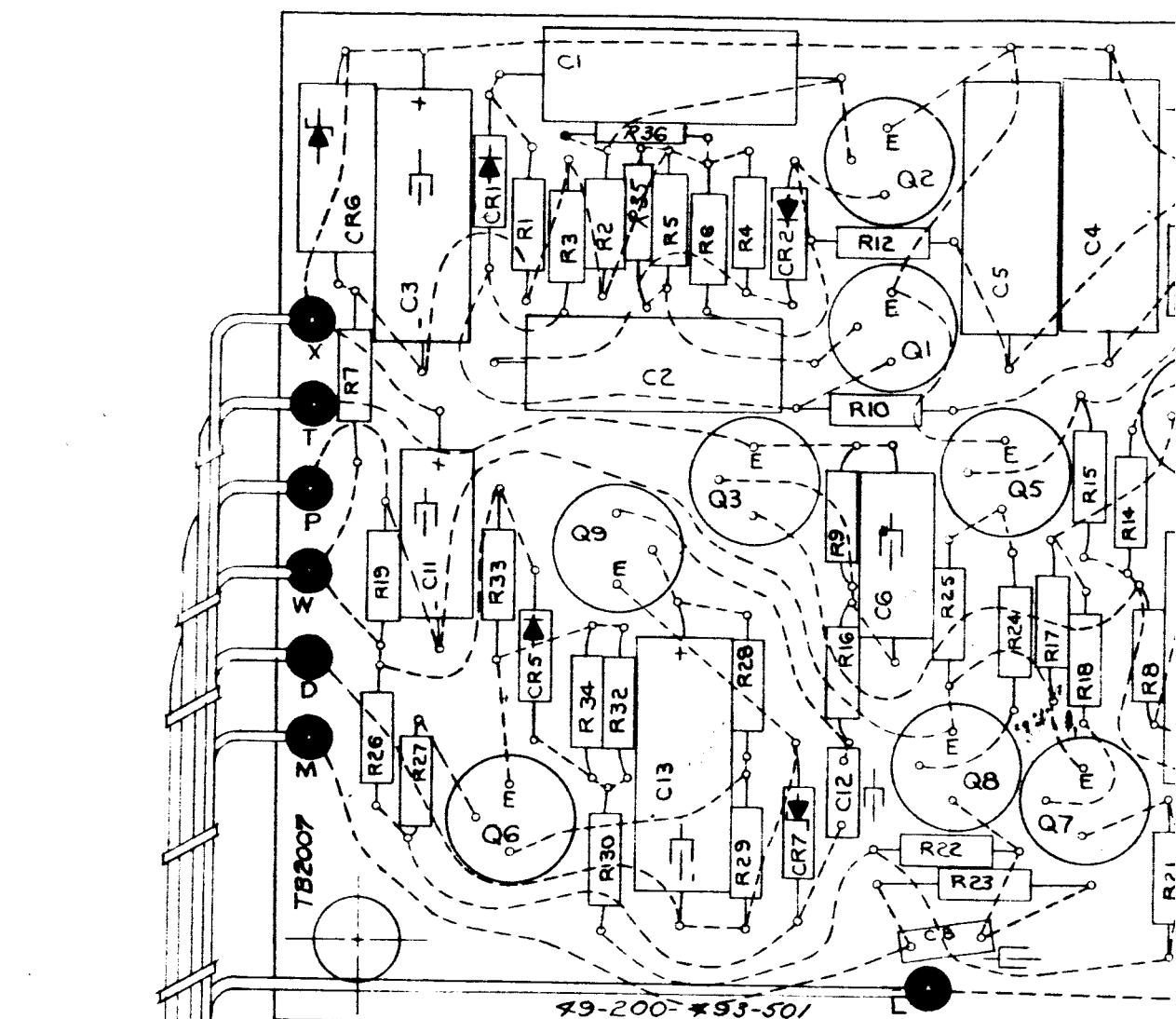
IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
1			Bill of Material Mod. 10 Control		
			Interposer (Regulator Board)		
2	R1, 3, 4, 6, 26	5	Resistor, Fixed Film, 2.2K $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD222J
3	R7, 11, 13, 17, 18, 21, 22, 24, 25, 14, 28, 15,	12	Resistor, Fixed Film, 1K $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD102J
4	R9	1	Resistor, Fixed Film, 4.7K $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD472J
5	R32, 35, 34, 36	4	Resistor, Fixed Film, * $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD
6	R10, 12, 16	3	Resistor, Fixed Film, 3.3K $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD332J
7	R20, 23	2	Resistor, FIXED FILM, 22 $\Omega$ $\pm$ 5%, 0.50W.	MIL-R-22684	RL20AD220J
8	R19, 30	2	Resistor, Fixed Film, 10K $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD103J
9	R29	1	Resistor, Fixed Film 2K $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD202J
10	R8	1	Resistor, Fixed Film, 77 $\Omega$ $\pm$ 5%, 0.25W.	MIL-R-22684	RL07AD771J
11	C10	1	Capacitor, Tantalum Polar 330uf, 6V.	MIL-C-26655B	CS13B337K
12	C12	2	Capacitor, MYLAR 0.022uf	CORNELL DUBILIER	WMF 1S22
13	C3	1	Capacitor, Tantalum Polar, 10mf, 35V	MIL-C-26655B	CS13BF106K

14	C4,5	2	Capacitor, MYLAR	0.033 mfd	CORNELL DUBILIER	WMF 1533
15	C6	1	Capacitor, Tantalum	6.8uf 35V	MIL-C-26655B	CS13BF685K
16	C7,8	2	Capacitor, Ceramic	10000 pf		CK06CW103K
17						
18	CR1,2,5	3	Diode		MSFC-338/3	SIN626
19	CR6	1	Zener Diode	11.2V	MSFC-338/7	SIN3021B
20	CR7	1	Zener Diode	5.6 V	MSFC-338/8	SIN752A
21	Q1,2,3,4,5,6	6	Transistor		MSFC-338/108	S2N1132
22	Q7,8,9	3	Transistor		MSFC-388/105	S2N697
23	G11	1	Capacitor, Tantalum Polar	2.7uf 35V	MIL-C-26655B	CS13BF275K
24	C13	1	Capacitor, Tantalum Polar	100uf 35V	MIL-C-26655B	CS13BF100K
25	C12	1	Capacitor, Ceramic	4700pf		CK06CW432K
26	R27	1	Resistor, Fixed Film	470uf 5% 0.25W	MIL-R-22684	RL07AD471J
27	R2, R5	2	Resistor, Fixed Film	22K 5% 0.25W	MIL-R-22684	RL07AD22J
28	P2007	1	CONNECTOR		WINCHESTER	WMF-20-P6H
29	R33	1	RESISTOR, FIXED FILM	15K 5% 0.25W	MIL-R-22684	RL07AD152J
30		1	PRINTED CIRCUIT BOARD			X9-300-5X3-001

31	9	PAD, TRANSISTOR MOUNTING	TRANS/PAD PRODUCTS	10012 DAT-BLANK
32	7	TERMINAL, TURRET	USECO	2030B
33	AR	WIRE AWG #26 TYPE B	MIL-W-16878D	ALPHA #7853
34	AR	SOLDER SNGO	QQ-S-571	TYPE RA
35	AR	FLUX	MIL-F-18256	TYPE A
36	AR	CONDUCTOR SOLID AWG #28	QQ-W-343	BELDEN B022
37	AR	SHRINK TUBING RNF-100	MSFC-SPEC-276	276-11TOD05DVA
REF		ELECTRICAL SCHEMATIC		29-300-886-701
38	AR	LACING TAPE, BLACK	MIL-T-713A	ALPHA #LC134
		ASBY. USED ON: 49-100-685-501 & -502		* VALUES TO BE DETERMINED ON TEST

01 2-4-66		17.11 WBS CS18BF357K		Bill of Material	
DELETED CAPACITORS		17.12 WBS METALIZED PAPER,		REGULATOR CONTROLLER	
C4 (17.10) C14 & 15 (17.12)		MIL-C-25, 180A254E52		INVERTER MOD 10	
CAPACITORS C4 & 5 (17.10)		17.18 WBS SING 626A		FIGURE 4-7	
WBS 9.0145		17.24 WBS 100 uF			
ADDED 17.25		CS1321FA78K			
02 3-25-66		17.27 WBS R23 1K20D			
WBS 25.000		18K, R107A1313			
R24, R25, R180 R192		ADDED 17.29 THRU 17.38			
ENC No 044 WBS 4300-16		TITLE WBS 1800-10			
17.35 WBS R32, 33, 34, 5		CONTR. INSTRUCTIONS			
17.36 WBS R107A0220J		OR 15-16-66			
17.40 WBS 2000-53		DATE 1-27-66			
17.41 WBS 17.42		BY D. C. S.			
		3341			
		DEPT.			
		3341			
		SIMILAR TO			
		SCALE			
		SHEET 1			
		OF			
		49-200-493-501			

JUMPER



PLUG PIN DIAGRAM  
FOR P2007

4-8-1

01 5-3-1966

REL. ED. 0170

NOTES:

1. PRINTED CIRCUIT ASSEMBLY TO BE CONSTRUCTED AND INSPECTED PER NPC 200-4
2. ALL TERMINALS SHALL BE MOUNTED ON THE PRINTED CIRCUIT BOARD PRIOR TO INSERTING ELECTRICAL COMPONENTS
3. TERMINAL IDENTIFICATION, PART, & PLUG NUMBERS ARE TO BE APPLIED IN BLACK INK .12 INCH HIGH IN POSITION SHOWN
4. COAT ALL TERMINAL IDENTIFICATION, PART, & PLUG NUMBERS WITH CLEAR LACQUER
5. AFTER QUALITY CONTROL HAS TESTED AND ACCEPTED THE PRINTED CIRCUIT ASSEMBLY APPLY A CONFORMAL COATING TO BOTH SIDES OF BOARD COATING MUST COMPLY WITH MSFC-PROC-257
6. REFERENCE DWG.  
PRINTED CIRCUIT BOARD 49-300-543-001  
PRINTED CIRCUIT MASTER 49-300-543-091  
ELECTRICAL SCHEMATIC 49-200-486-401



CONFIDENTIAL - PROPERTY OF  
**ALLIS-CHALMERS MFG. CO.**  
 3341 MI WORKS

UNLESS OTHERWISE SPECIFIED:

1 - PLACE DEC ±  
 2 - PLACE DEC ± .01  
 3 - PLACE DEC ±

ANGULAR ±

✓  
 MACHINED  
 SURFACE  
 TEXTURE

DR 002 (UDC)

CH 8/11/66

AP

SIMILAR TO

SCALE

3:1

SHEET

FINAL

49-200-493-501

NEXT ASSEMBLY USED ON

NAME  
 PRINTED CIRCUIT ASSEMBLY  
 CONTROL INVERTER

MATERIAL

FIGURE 4-8

WT.

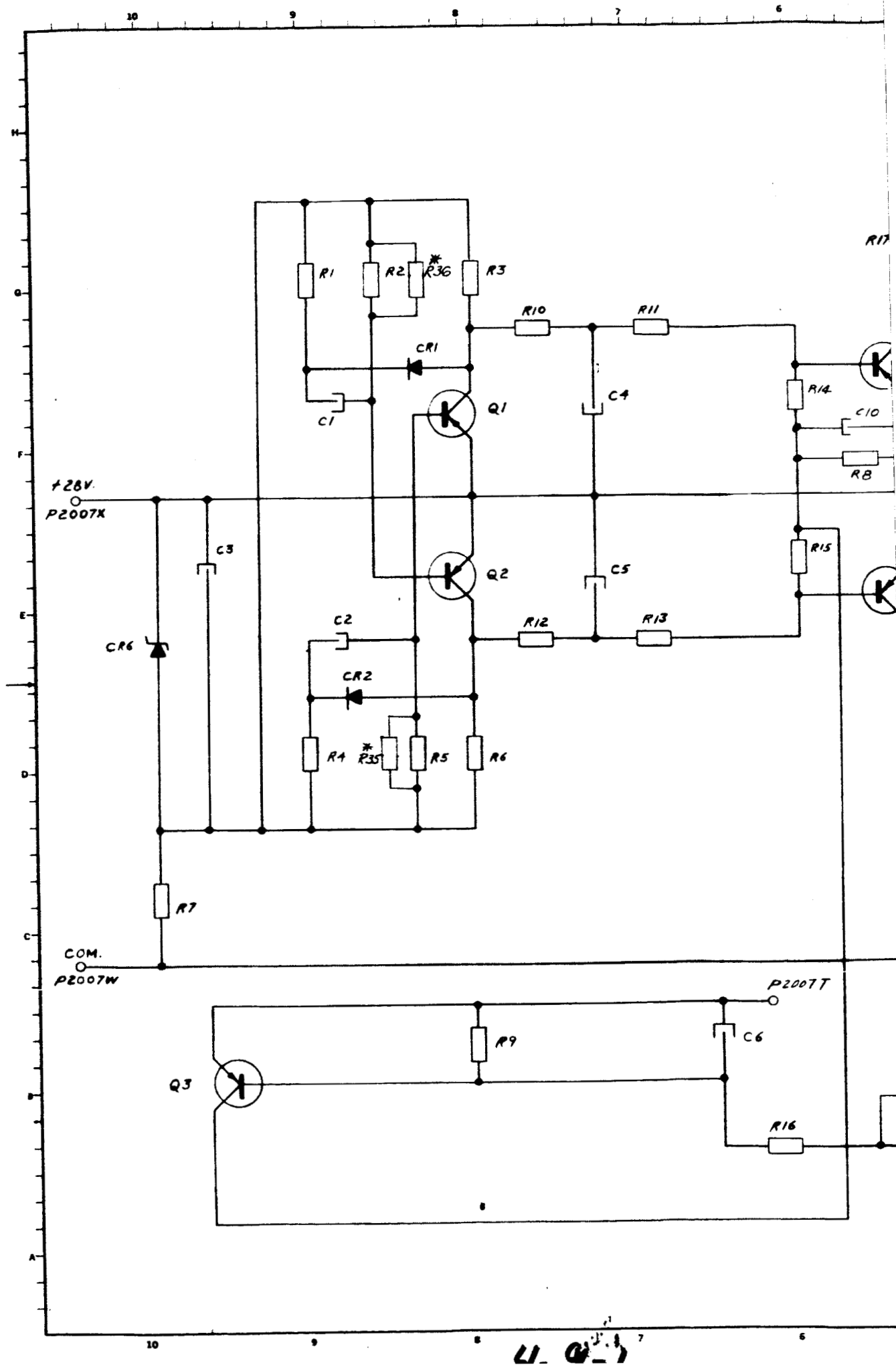
PART NO

49-300-542-401

2 #1058 (UDC)

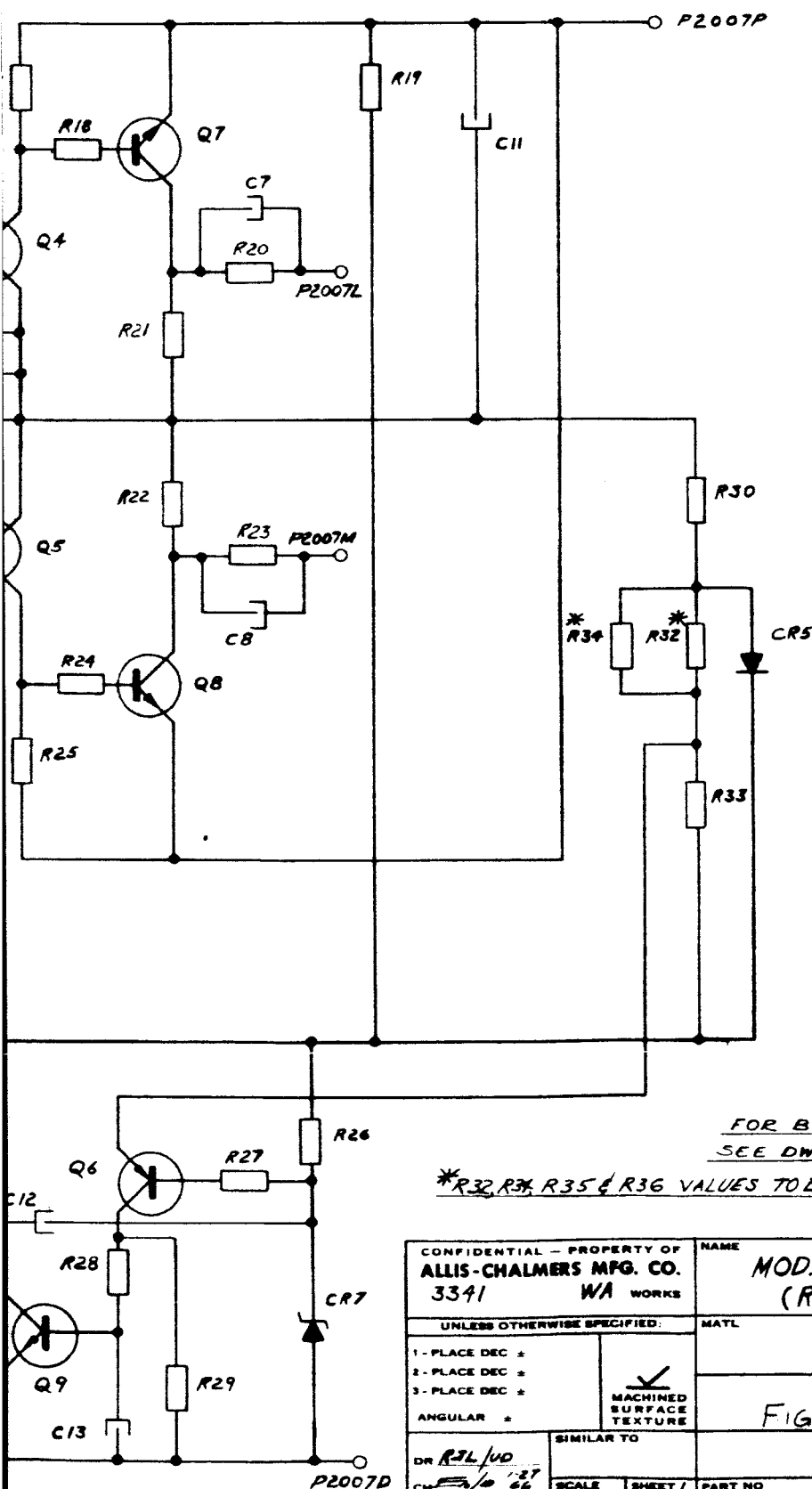
PRINTED IN U.S.A. FORM No. 0132-1

49-300-542-401 01





01	2-4-66	JD.
DELETED CAPACITORS C9, C14 & C15		
02	3-25-66	LA 10
DELETED REL ED. D. 30 1038		
ADDED R35 & R36		
03	5-2-1966	
E.O. D170		



FOR BILL OF MATERIAL  
SEE DWG. 49-200-493-501

\*R32, R33, R35 & R36 VALUES TO BE SELECTED DURING TEST

CONFIDENTIAL - PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b> 3341 WA WORKS		NAME <b>MOD. 10 CONTROL INVERTER (REGULATOR)</b>	
UNLESS OTHERWISE SPECIFIED:		MATERIAL	
1 - PLACE DEC *	<input checked="" type="checkbox"/> MACHINED SURFACE TEXTURE	FIGURE 4-9 <div style="float: right;">WT <input type="checkbox"/></div>	
2 - PLACE DEC *			
3 - PLACE DEC *			
ANGULAR *			
DR <i>R3L/UD</i>	SIMILAR TO		
CH <i>27</i>	SCALE <i>1/2"</i>	SHEET <i>1</i>	PART NO <b>49-300-486-401</b>
AP <i>3/4/66</i>		OF <i>1</i>	

49-300-486-401 03

## CIRCUIT COMPONENT LIST FOR DRAWING 49-300-503-401

IT. REFERENCE NO. DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
1	1	Circuit Board		49-300-501-001
2	22	Terminal-Feed Thru	USECO	2500C-1
3	23	Terminal	USECO	2030C-1
4	CR1 2, 4, 5, 6, 7, 8, 9, 10, 11, 12, 13, 14, 15	Diode		IN4003
5	1	Zener Diode 75V.		IN3041B
6	1	Transformer		49-200-491-401
7	1	Input Choke		49-200-489-401
8	1	Output Choke		49-200-490-401
9	1	Capacitor, Tantalex 110uf 50V.		130D117C2050W2
10	1	Capacitor, Tantalex Polar 100mf 15V.	MIL-C-28655A	CS13AE101K
11	1	Capacitor, Tantalex 350mf 15V.		130D357C2015W2
12	4	Washer-0.75 O.D.x0.125 I.D.x.093 Thk.	Make From Glass Epoxy	49-200-492-012
13	2	Washer-0.50 O.D.x0.125 I.D.x.093 Thk.	Make From Glass Epoxy	49-200-492-013
14	2	No.6-32x1.25 Lg. Mach. Screw		49-200-492-014
15	1	No.6-32x1.00 Lg. Mach. Screw		00-676-223-132

4-10-1

16	6	No. 6 Rd. Washer	00-651-027-050
17	3	No. 6 Lockwasher	00-655-017-018
18	3	No. 6-32 Hex. Nut	00-631-123-106
19	2'8"	#22 AWG TYPE Teflon Insul. Stranded Silver Plated Copper Type 3 - White MIL-W-16878D	Alpha 2855 or Equivalent
20	YES	WIRING SCHEMATIC	49-200-484-401

4-10-2

01 2-4-66

REVISED &  
RETYPE

02 3-2-66 PM

WTS 3/24/66

49-200-531-502

49-200-531-501

NEXT ASSEMBLY USED ON

UNLESS OTHERWISE NOTED  
DIMENSIONS ARE IN INCHES &  
MACHINING TOLERANCES ARE

DIMENSIONS	UP TO 1/8 INCH	OVER 1/8 TO 1 INCH	OVER 1 INCH
FRACTIONAL	± .005	± .005	± .005
DECIMAL	± .005	± .005	± .005

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RESEARCH #3341

DRN RAR 2-18-66

CHD 2-29-66

APR 3-19-66

BILL OF MATERIAL

CONTROLLER INVERTER  
OUTPUT BOARD  
MOD. 10

FIGURE 4-10

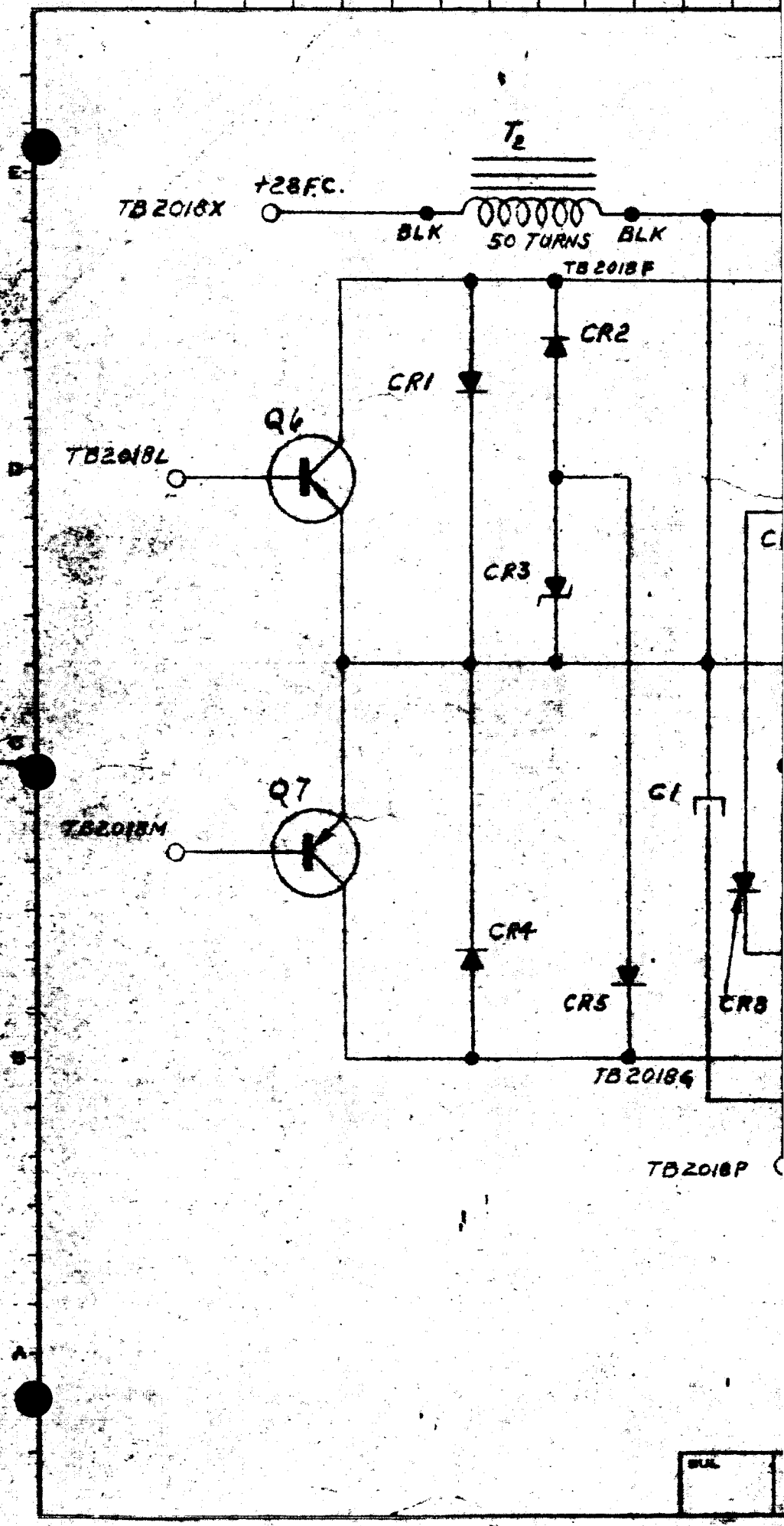
SIMILAR TO

SHEET

49-200-492-501

8

7

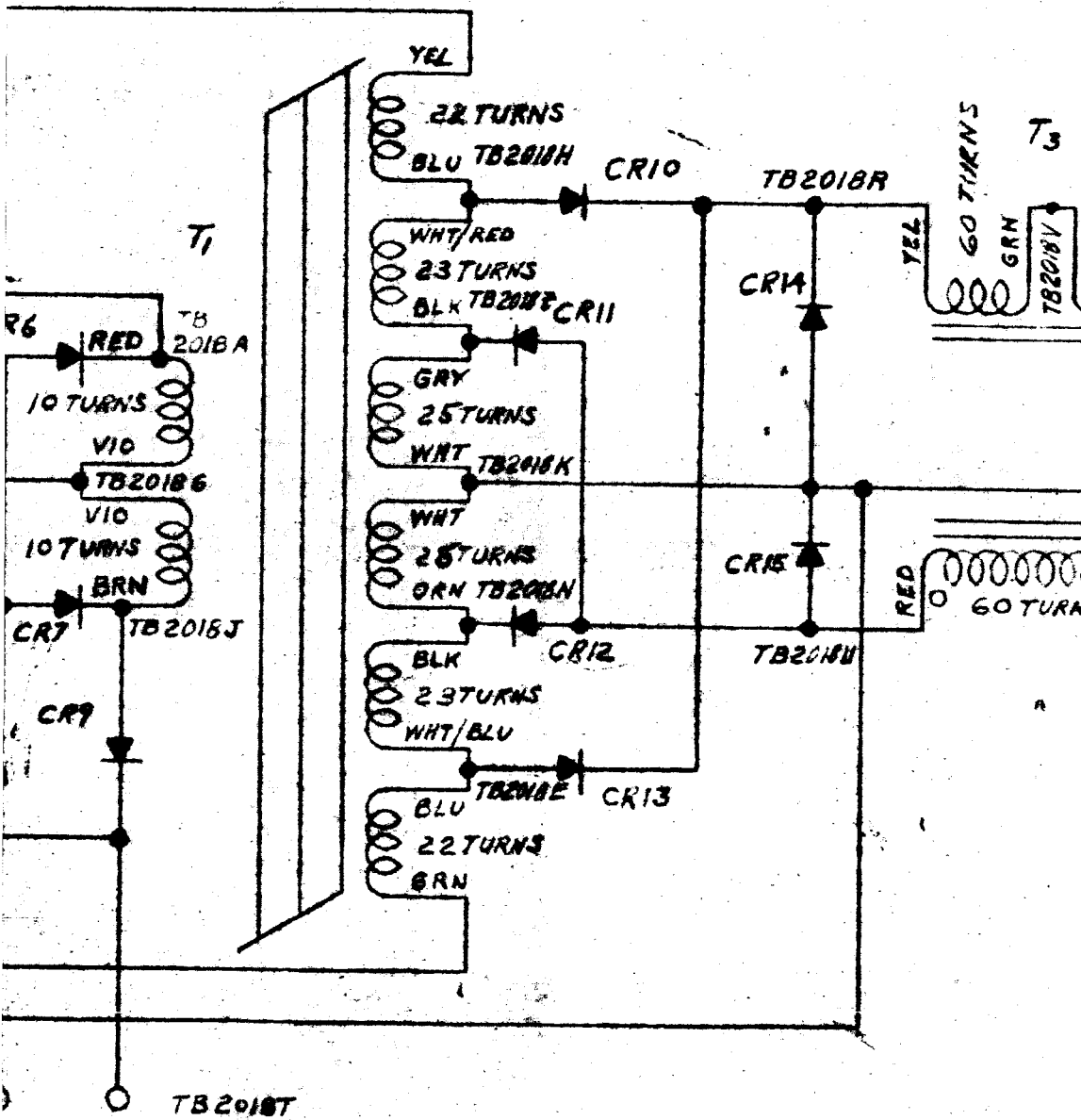


BLK

4-11-1

8

○ TB2018S



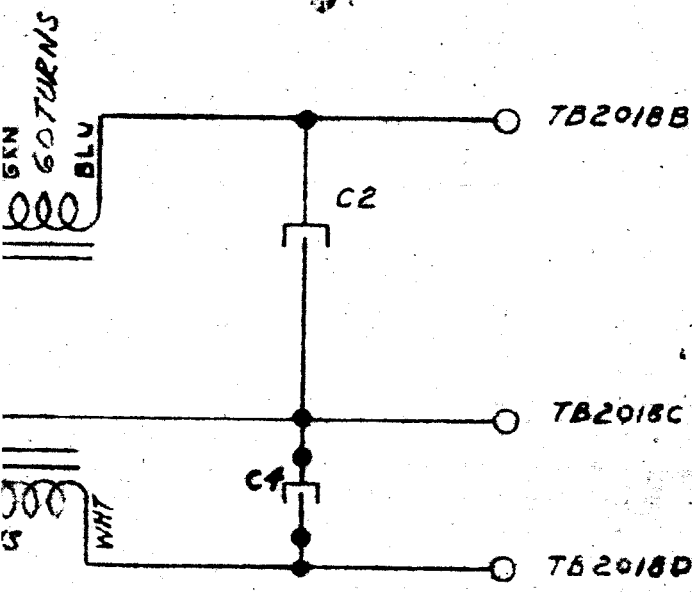
CON
ALL
33
UNLE
1-PLACE
2-PLACE
3-PLACE
ANGULA
DR RJ
CH
APL

49-200-484

02

4-11-2

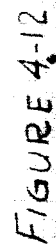
01	2-4-66	REV. 1A
NUMBER OF TURNS REVISED		
02	2-25-66	REV. 1A
REV. 3/26/66 RCL. ED. 0150 2754		



FOR BILL OF MATERIAL SEE  
DWG. 49-200-492-501

CONFIDENTIAL—PROPERTY OF <b>CHALMERS MFG. CO.</b> 41 WA PLANT		NAME <b>MOD. 10 CONTROL INVERTER          (POWER OUTPUT)</b>	
OTHERWISE SPECIFIED: DEC ± DEC ± DEC ± OR ±		MATL  	
<input checked="" type="checkbox"/> <b>MACHINED          SURFACE          TEXTURE</b>		<b>FIGURE 4-11</b>	
SIMILAR TO  		R WT P	
SCALE —		PART NO. <b>49-200-484-401</b>	

4-11-3





supplied by a dc. to dc. inverter from the auxiliary 28 volts dc. power source which may be either the fuel cell output or an external source. The control power supply is designed to operate over an input voltage of 18 to 35 vdc with an overall efficiency of not less than 70% at a 28 volt input. The power supply operates as follows. The output of an 800 cycle, free running, square wave oscillator, is modified to a triangular wave. The triangular wave drives an amplifier into saturation over approximately 165 electrical degrees. The output of this amplifier drives a transformer which operates similar to an AUTO transformer being alternately switched across the dc. power source. One secondary winding provides power for the 11.2 vdc filter network. Another secondary winding provides the power for the -5.6 vdc filter network, which in turn feeds the main voltage regulator circuit. A reactor forms part of the filtering network of both the +11.2 and the -5.6 vdc outputs. It acts as a load equalizer for any unbalanced load currents between these two outputs. The main voltage regulator circuit controls the bias applied to amplifier, thereby providing pulse width voltage regulation.

4.5.2 Temperature Controller See Figure 4-13A and 14-13B (Dwg. No. 49-200-413-501), Fig. 4-14 (Dwg. No. 49-400-337-401), Fig. No. 4-15 (Dwg. No. 49-400-293-401).

The temperature controller consists of 3 distinct circuits: the sensing circuit, the coolant circuit, and the heating circuit. The sensing circuit determines whether cooling or heating is required by sensing the fuel cell stack temperature. It then provides a signal to either the coolant control circuit or the heating circuit, which control respectively, the liquid coolant control valve or the fuel cell stack standby heaters.

The temperature of the fuel cell stack is monitored by a thermistor located on the stack. This signal is fed to an operational amplifier whose output is

# CIRCUIT COMPONENT LIST FOR DRAWING 49-400-337-401

IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
1	R34, 36 R1, 2, 9, 23,	6	Resistor Fixed Film, 21.2K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD222J
2	R3, 11, 55	3	Resistor Fixed Film, 5.6K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD562J
3	R4, 12, 16, 17, 18, 28, 33, 37, 44, 45	9	Resistor Fixed Film, 10K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD103J
4	R5, 26	2	Resistor Fixed Film, 1.8K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD182J
5	R6, 27	2	Resistor Fixed Film, 220 ohms <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD221J
6	R7, 29, 32, 35	4	Resistor Fixed Film 100K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD104J
7	R8, 24	2	Resistor Fixed Film 8.2K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD822J
8	R10, 25	2	Resistor Fixed Film 6.2K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD622J
9	R30, 22, 53, 51	4	Resistor Fixed Film 2K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD202J
10	R31, 18, 21, 54, 42	5	Resistor Fixed Film 1K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD102J
11	R45	1	Resistor Fixed Film 4.7K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD472J
12	R15, 52	2	Resistor Fixed Film 3.3K <sup>+</sup> 5% 0.25W	MIL-R-22684	RL07AD332J

4-13A-1

13	R41	1	Resistor Fixed Film 5.1K - 5% 0.25W	MIL-R-22684	RL07AD512J
14	R40	1	Resistor Fixed Film 20K - 5% 0.25W	MIL-R-22684	RL07AD203J
15	R38, 14	2	Resistor Fixed Film 12K - 5% 0.25W	MIL-R-22684	RL07AD123J
16	K1	1	Relay Permanent Magnet 24VDC	P & B	SCI1DB
17	C2	1	Capacitor Ceramic 2200pf	MIL-C-11015C	CK06CW222K
18	C4, 5	2	Capacitor-Ceramic 470pf	MIL-C-11015C	CK05CW471K
19	C6	1	Capacitor Ceramic 1000pf 200V	MIL-C-11015C	CK05CW102K
20	C1, 3	2	Capacitor Tantalum Polar 10mf 35V	MIL-C-26655	CS13AF100K
21	C7, 10	2	Capacitor Tantalum Polar 47mf 35V	MIL-C-26655	CS13AF470K
22	CR1	1	Diode Zener 39V		IN3034B
23	CR3,	1	Diode	MSFC-338/3	SIN626
24	CR6	1	Diode		IN4003
25	CR5	1	Diode Zener 3.6V	MSFC-338/8	SIN747A
26	CR8	1	Diode Zener 6.8V	MSFC-338/8	SIN754A
27	Q1	1	Operational Amplifier	Fairchild	MA702A

4-13A-2

28	12, 13, 17, 18	10	Transistor	MSFC-338/105	S2N697
29	Q3, 9, 10, 14	4	Transistor	MSFC-338/108	S2N1132
30	R19*, R13*	2	RESISTOR FIXED FILM * $\pm 5\% 0.25W$	MIL-R-22684	*
			*VALUE SELECTED DURING TEST		
32	R20	1	RESISTOR FIXED FILM $6.8K \pm 5\% 0.25W$	MIL-R-22684	R107AD 682J
35	R50	1	RESISTOR FIXED FILM $470\Omega \pm 5\% 0.25W$	MIL-R-22684	R107AD 471J
34	C8	1	CAPACITOR CERAMIC 1500pF	MIL-C-11015C	CK06CW152K
35	C11	1	CAPACITOR CERAMIC 6800pF	MIL-C-11015C	CK06CW682K
36	C9	1	CAPACITOR TANTALUM 27mF 35V	MIL-C-26655	CS13AF270K
37	Q19	1	TRANSISTOR	MSFC-338/119	S2N1486
38	CR10	1	DIODE ZENER		IN3034B
39	P2011	1	CONNECTOR	WINCHESTER	SMRE-20-PGH

01

9-14-65

RETYPE

02

2-7-66

R34, 36 WERE 1K (1T.10)

R12 WAS 10K (1T.8)

R14, 20 WERE 12K (1T.15)

R42 WAS 2K (1T.9)

R19 WAS 4.7K (1T.11)

ADDED R50 (1T.33), R51 (1T.11),

R52 (1T.12), R53 (1T.9) & R54

(1T.10) & R18 (1T.28)

ADDED (ITEMS 34, 35, 36,

37 & 38)

1T.17 WAS 10000P

CK06CW152K

ADDED 10.34

03 3-25-66

THE 3/24/66

RELE. 01/50 10/2

IT.1-DELETED R51 & R45

WAS (3) REQ'D.

IT.2-DELETED R45

ADDED R55.

IT.3-DELETED R21

ADDED R43

IT.4 ADDED R51

WAS (3) REQ'D.

IT.23-DELETED CR9

WAS (2) REQ'D.

IT.25-DELETED CR2 & 4

WAS (3) REQ'D.

IT.28 ADDED CR8 WAS

9 R50 & 2 (CONFIDANT 5/2)

04 5-18-66 54190

UNLESS OTHERWISE NOTED

DIMENSIONS ARE IN INCHES &

MACHINING TOLERANCES ARE:

DIMENSIONS

UP TO 6 INCL

OVER 6 TO 24 INCL

OVER 24

FRACTIONAL

$\pm$

$\pm$

$\pm$

DECIMAL

$\pm$

$\pm$

$\pm$

ANGLES  $\pm$

TEMPERATURE

CONTROLLER MOD.10

FIGURE 4-73A

ALLIS-CHALMERS MFG. CO., Milwaukee, Wis.

RESEARCH #3341

DEPT.

PROPERTY OF

DRN R.V.L. 2/1/66

CHD 2/1/66

APD 2/1/66

TRD

SIMILAR TO

SHEET 1

OF 2

SCALE

49-200-413-501

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REQ. ITEM	DESCRIPTION	MIL. SPEC. OR MFG. NAME	PART NUMBER
14	40 Pad. Transistor Mounting	Transload Products	100-12DAT-Black
1	41 Pad. Transistor Dual	Milton Ross	10229
23	42 Terminal Turret	USECO	2030B
AR	43 Wire AWG #26 Type B	MIL-W-16878/1	Alpha #1853
AR	44 Conductor, Solid AWG #24	QQ-W-343	Balden #8022
AR	45 Shrink Tubing RNF-100	MSFC-SPEC-276	276-11TOP05 PIA
AR	46 Solder SN60	QQ-S-571	Type RA
AR	47 Flux	MIL-F-14256	Type A
1	48 Printed Circuit Board		49-400-338-001
2	49 Screw	MS-51957	MS51955-14
2	50 Nut, Self Locking	NAS1021	NAS1021004
6	51 Washer, Plain	MS-15795	MS15795-804
AR	52. LACING TAPE, BLACK	MIL-T-713A	ALPHA #LC134

4-19-B-1

U-13-B-E

## Ref. Electrical Schematic

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
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— PROPERTY OF —  
ALLIS-CHALMERS MFG. CO.

Temperature Controller MOD.10

# PLANT

Research 3341

**CONFIDENTIAL**

FIGURE 4-13A

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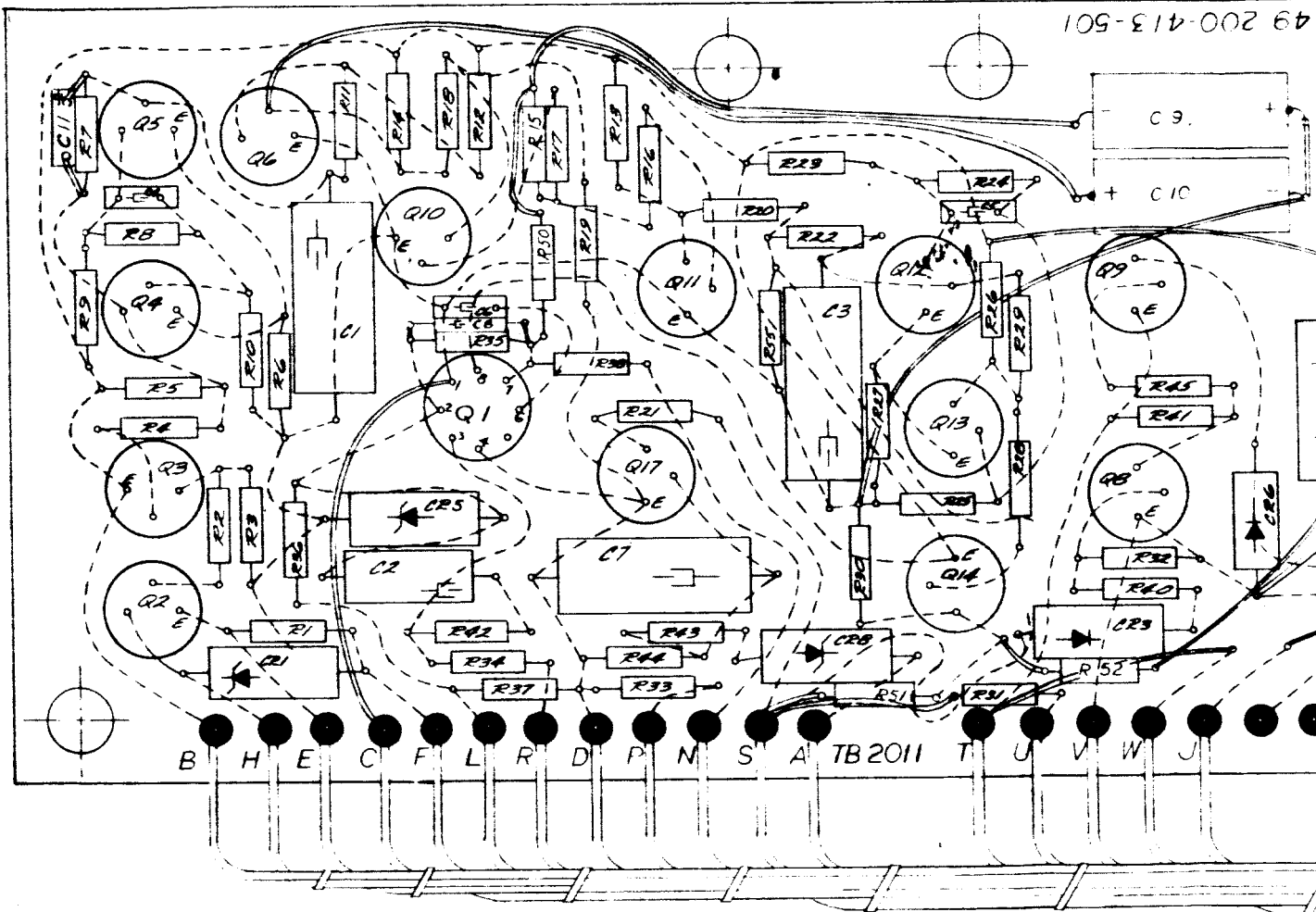
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19-300-413-501

THE UNIVERSITY OF CHICAGO PRESS

49 200-413-501



4-14-1

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NOTES:

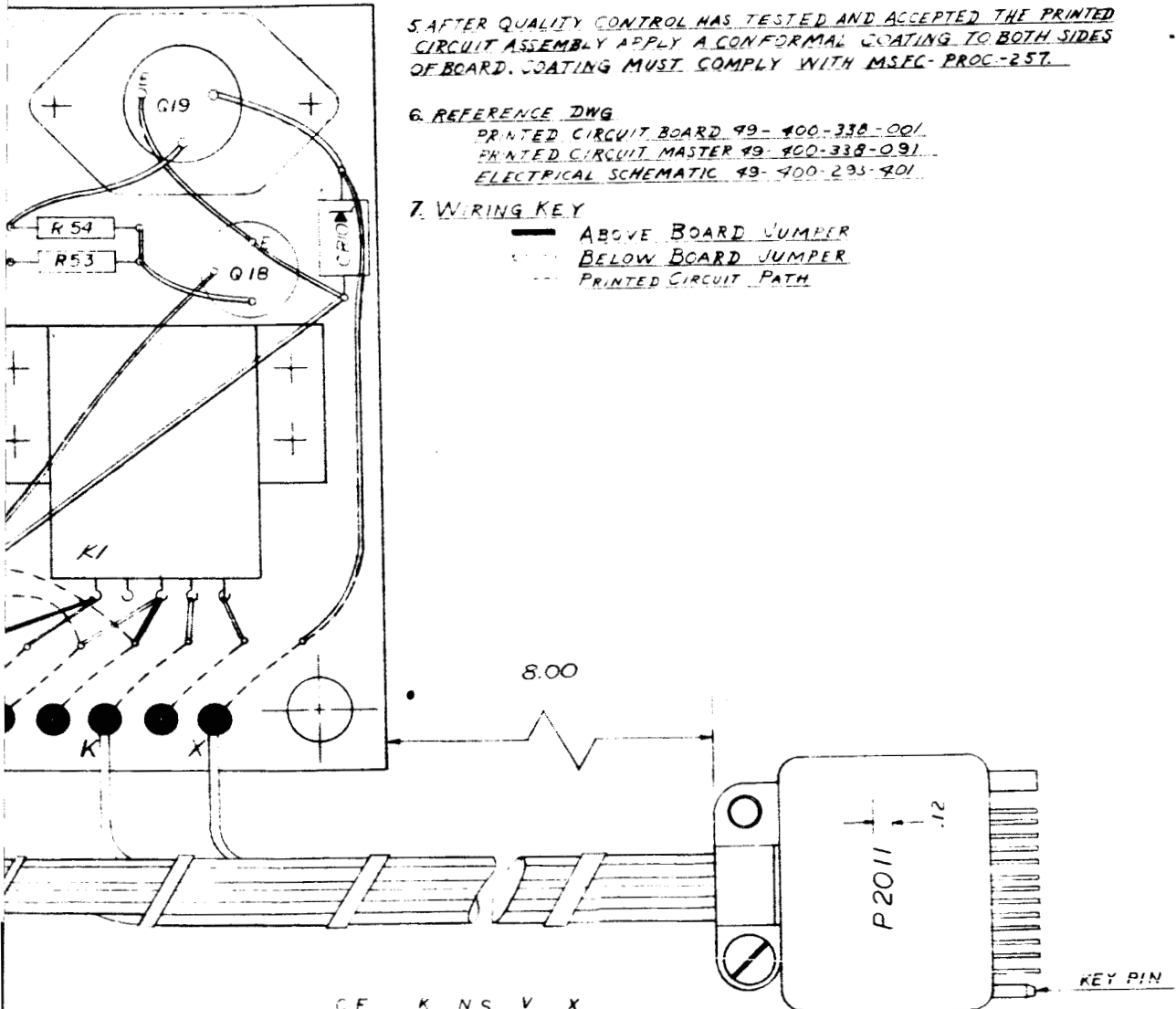
1. PRINTED CIRCUIT ASSEMBLY TO BE CONSTRUCTED & INSPECTED PER NPC 200-4.
2. ALL TERMINALS SHALL BE MOUNTED ON THE PRINTED CIRCUIT BOARD PRIOR TO INSERTING ELECTRICAL COMPONENTS.
3. TERMINAL IDENTIFICATION, PART & PLUG NUMBERS ARE TO BE APPLIED IN BLACK INK .12 INCH HIGH IN POSITION SHOWN.
4. COAT ALL TERMINAL IDENTIFICATION, PART & PLUG NUMBERS WITH CLEAR LACQUER.
5. AFTER QUALITY CONTROL HAS TESTED AND ACCEPTED THE PRINTED CIRCUIT ASSEMBLY APPLY A CONFORMAL COATING TO BOTH SIDES OF BOARD. COATING MUST COMPLY WITH MSEC-PROC-257.

6. REFERENCE DWG

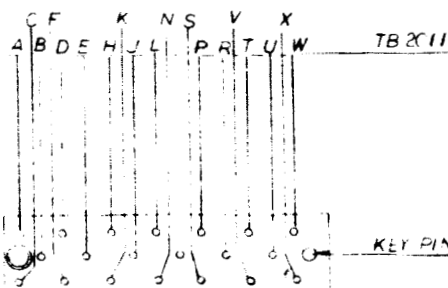
PRINTED CIRCUIT BOARD 49-400-338-001  
PRINTED CIRCUIT MASTER 49-400-338-091  
ELECTRICAL SCHEMATIC 49-400-293-401

7. WIRING KEY

- ABOVE BOARD JUMPER
- - - BELOW BOARD JUMPER
- PRINTED CIRCUIT PATH



WIRING KEY  
 \* TOP DECK  
 □ LOWER DECK



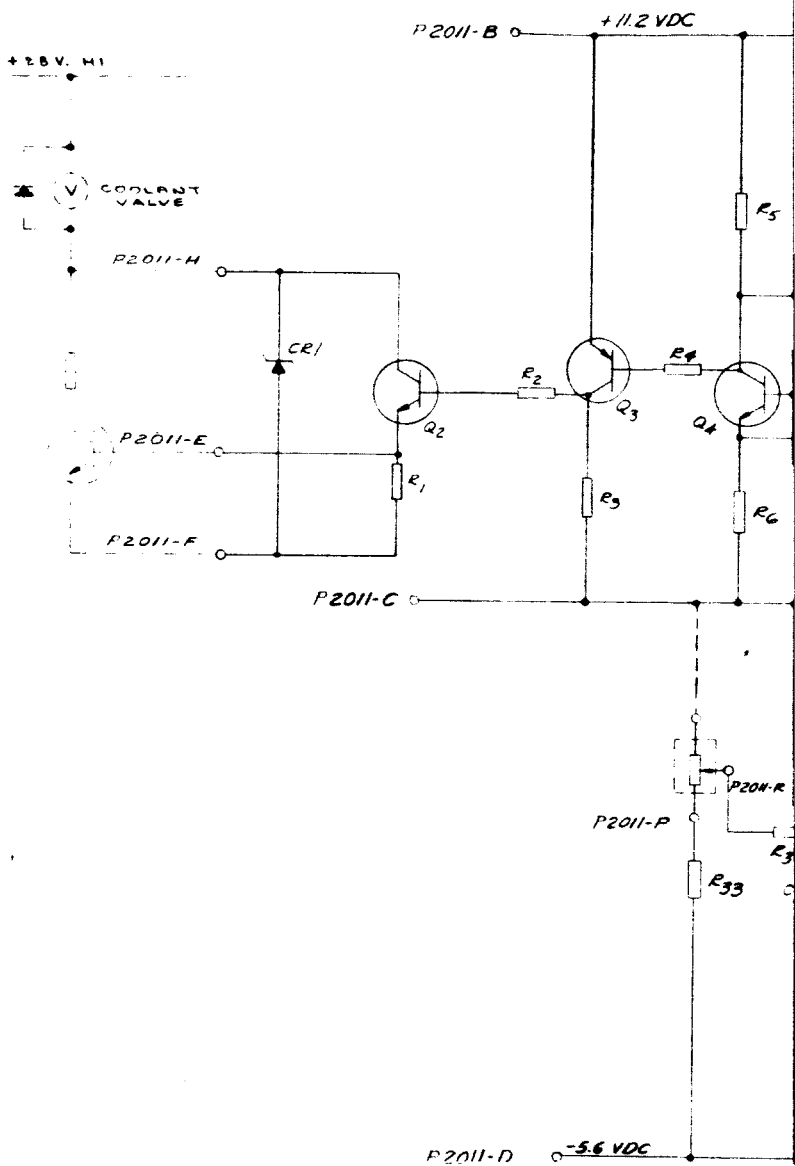
PLUG PIN DIAGRAM FOR P2011

CONFIDENTIAL - PROPERTY OF ALLIS-CHALMERS MFG. CO.	
49-400-413-501	PLANT
UNLESS OTHERWISE SPECIFIED:	
1. PLACE DEC 8	✓ MACHINED SURFACE TEXTURE
2. PLACE DEC 8 .01	
3. PLACE DEC 8	
ANGULAR 8	
69/100 3/4/66	SIMILAR TO
CH 2/2/74 9/8/66	SCALE 5:1
AP	SHEET 1 OF 1

49-400-413-501	
NEXT ASSEMBLY USED ON	
NAME PRINTED CIRCUIT ASSEMBLY TEMPERATURE CONTROLLER II	
MATERIAL	
FIGURE 4-14	
PART NO. 49-400-413-501	

49-400-413-501

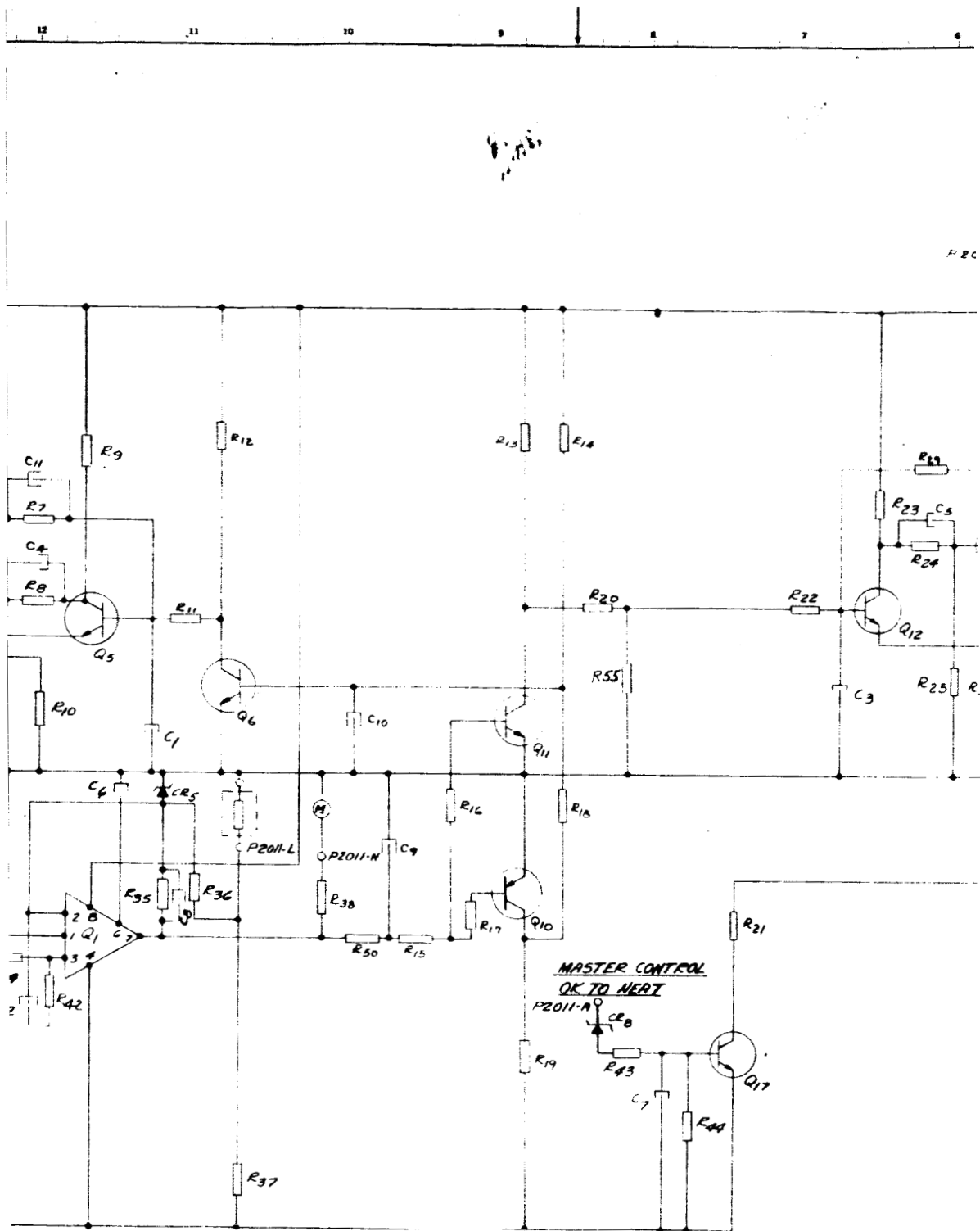




EXTERNAL COMPONENT

Ⓜ EITHER AN EXTERNAL 500-0-500 OHM POTENTIOMETER OR A JUMPER TO BE INSTALLED BETWEEN P2011-P AND P2011-R

4-15-1

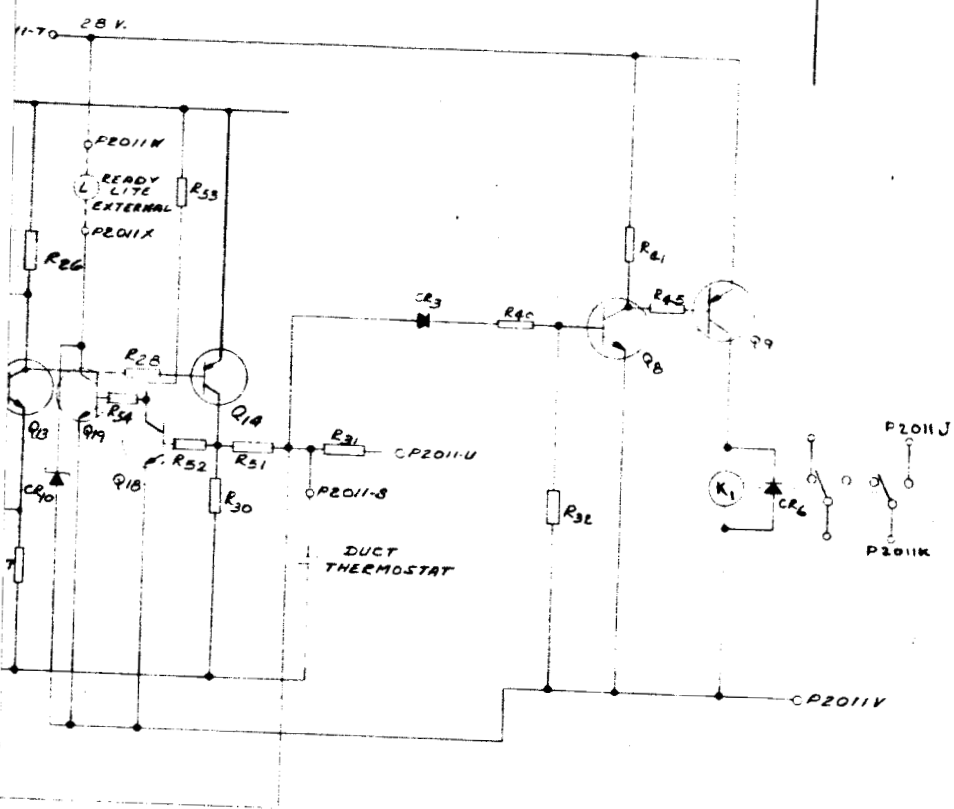


P. 20

NOTE  
FC  
SE

0.4 AMP. METER OR  
BETWEEN TERMINAL "N" & GROUND

01	2-7-66	10
DWA REVISED BROUGHT UP TO DATE		
02	3-25-66	14
BY SUB. BY RA. ED. D-5 21-66		
PROJECT NO. P2011L		
03	4-21-66	



OR BILL OF MATERIALS  
E DWG. 49-200-413-501

CONFIDENTIAL - PROPERTY OF ALLIS-CHALMERS MFG. CO. RSCH. #3541 M.I. WORKS		NAME <u>TEMPERATURE CONTROLLER II</u>	
UNLESS OTHERWISE SPECIFIED		MATERIAL	
1. PLACE DEC 2	<input checked="" type="checkbox"/> MACHINED SURFACE TEXTURE	FIGURE 4-15	
2. PLACE DEC 2			
3. PLACE DEC 2			
ANGULAR		WT.	
OR <u>12-24-66</u>	SIMILAR TO		
CHG. BY <u>5/2/66</u>	SCALE <u>N.A.</u>	SHEET <u>CF1</u>	PART NO. <u>49-400-293-401</u>
APPROVED BY <u>4/16</u>			

49-400-293-401 03

4-15-3

a negative signal if cooling is required, or a positive signal if heating is required. The positive signal from the operational amplifier is fed to an inhibited AND gate. To satisfy the inhibited AND gate, the "OK to Heat Signal" from the master controller must be present, the "Stand by heater thermostat signal" cannot be present. Completing the input requirements of this gate, the output feeds a Schmitt trigger with dual outputs: one turns on the 100-watt current driver (100 watt supply for the standby heaters), the second output is the command signal for the 1,600 watt warm-up heaters. The negative signal from the operational amplifier feeds the coolant Schmitt trigger circuit which in turn triggers the relay driver for the coolant control valve.

4.5.3 Purge Controller - See Figure 4-16A and 4-16B (Dwg. No. 49-200-465-501), Fig. 4-17 (Dwg. No. 49-500-183-401), Fig. 4-18 (Dwg. No. 49-400-291-401).

Purging is accomplished by a brief and accurately timed opening of the purge valves which permit a rapid movement of hydrogen and oxygen through the respective reactant cavities of the fuel cell. This action cleans the cavities of inert gases which are an accumulation of impurities contained in the reactant gases. It is therefore evident that the frequencies of purging are related to the amount of reactant gases used and the amount of impurities in the gases. The amount of reactant gases used is directly related to the ampere-hours of current produced by the fuel cell.

The purge controller provides three modes of purging.

# CIRCUIT COMPONENT LIST FOR DRAWING 49-500/83-401 P/A

IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
1	R1	1	Resistor Fixed Film 51K $\pm 5\%$ 0.25W	MIL-R-22684/1	RL07ADSI03
2	R2, 3, 141, 6	4	Resistor Fixed Film 2.2K $\pm 5\%$ 0.25W	MIL-R-22684/1	RL07AD322J
3	R5	1	Resistor Fixed Film 82K $\pm 5\%$ 0.25W	MIL-R-22684/1	PL07AT823J
4	-- R10, 140, 121	3	Resistor Fixed Film 1K $\pm 5\%$ 0.25W	MIL-R-22684/1	RL07AD102J
5	R <sup>9</sup> , 14	2	Resistor Fixed Film 12K $\pm 5\%$ 0.25W	MIL-R-22684/1	PL07AD123J
6	R11, 14	1	Resistor Fixed Film 100K $\pm 5\%$ 0.25W	MIL-R-22684/1	PL07AD101J
7	R12, 13, 15, 17, 18, 21, 22, 23, 24, 30, 144, 40, 41, 43, 145, 47, 48, 52, 53	60	Resistor Fixed Film 100K $\pm 5\%$ 0.25W	MIL-R-22684/1	PL07AD102J

4-16 R-1



17	64, 7, 12, 16, 19, 24, 28, 32, 36, 40, 44, 48, 52,	13	Capacitor Ceramic 470 pf	MIL-C-11015/18	CK05CW470K
18					
19	65, 10, 11, 14, 15, 18, 20, 22, 24, 26, 27, 30, 31, 34, 36, 37, 39, 42, 43, 47, 50, 51, 54, 55	26	Capacitor Ceramic 5000 pf	MIL-C-11015/19	CK06CW5000K
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# CIRCUIT COMPONENT LIST FOR DRAWING 49-500-163-401 P/A

IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
20	C10	1	Capacitor, Tantalum 68 pf	MIL-C-26655 / 2	CS125P2001
21	C01, 2, 4, 5, 6, 7, 8, 9, 10, 11, 12, 13, 14, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52	36	Diode	MSFC - 338 / 3	31 M2C
22	CX3, 4, 5, 6, 7, 8, 9, 10, 11, 12, 13, 14, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52	12	Resistor	MSFC - 338 / 8	SIN 747A
23	A1	1	Operational Amplifier	Intersil	uA702A
24	Q2, 3, 4	2	Transistor	MSFC - 338 / 100	2N4119
25	Q5, 6, 7, 8, 9, 10, 11, 12, 13, 14, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52	5	Transistor	MSFC - 338 / 105	2N4119
26	Q37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52	1	Dual Transistor	Fairchild or Motorola	2N3612

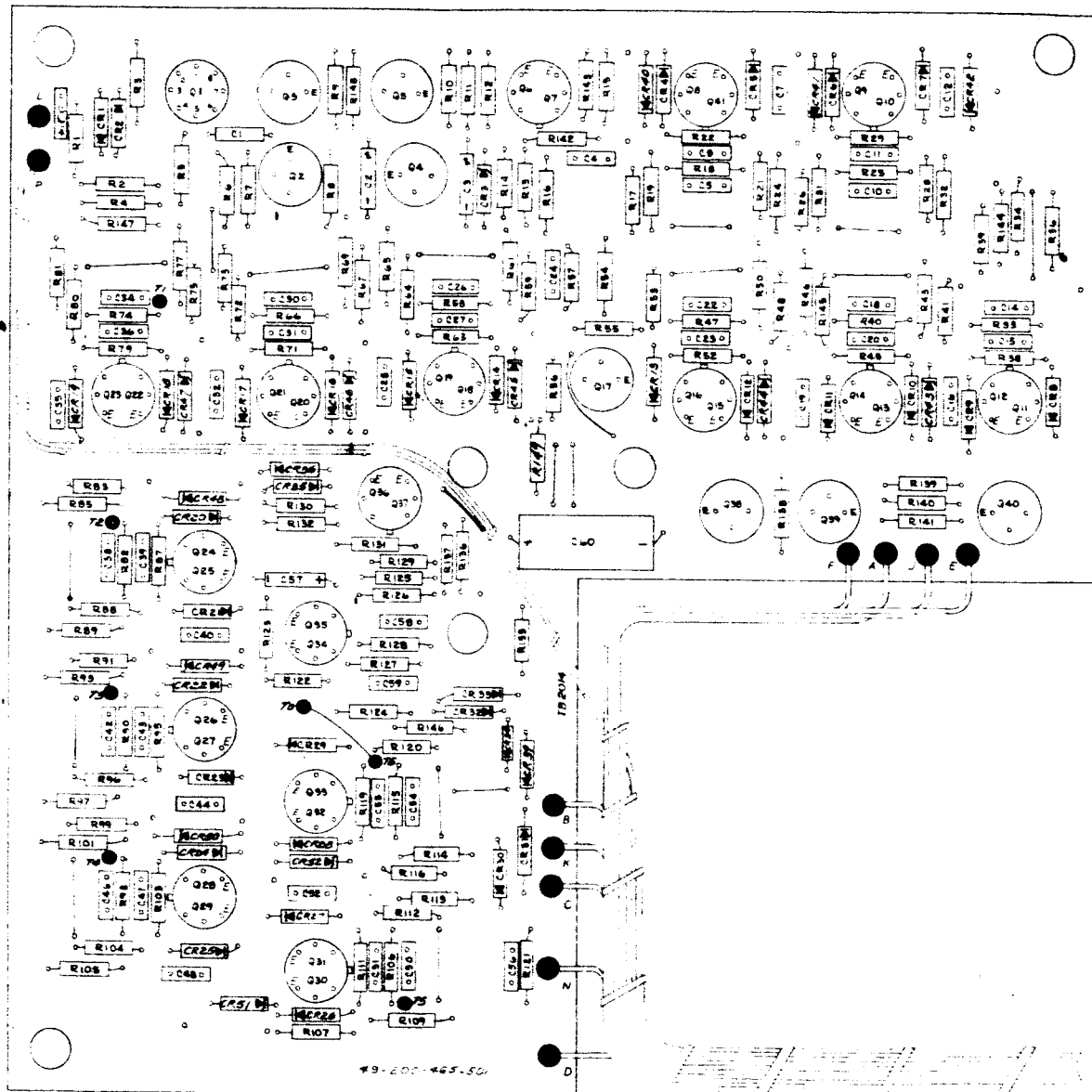
4-16B-1



27	R7	1	RESISTOR FIXED FILM 3.3K $\pm 5\%$ 0.25W	MIL-R-22684/1	RL07AD 332J
28	R143	1	RESISTOR FIXED FILM 56K $\pm 5\%$ 0.25W	MIL-R-22684/1	RL07AD 562J
29	C56, C61	2	CAPACITOR TANTALUM 1uF	MIL-C-26655/2	CS13BF105K
30		1	CONNECTOR	WINCHESTER	SMP-E-20-F9H
31		AR.	CONDUCTOR, SOLID AWG 24	QQ-W-343	FEEDEN 2022
32		AR.	SHRINK TUBING RNF-100	MS-FC-SPEC-246	276-117 0L0360HA
33		8	PAD, TRANSISTOR MOUNTING	TRANSIPAD PRODUCTS	10012 DAT-BLACK
34		18	TERMINAL, TURRET	USECO	2030B
35		17	PAD, DUAL TRANSISTOR	MILTON ROSS	10229
36		AR	WIRE #20 AWG TYPE B	MIL-W-16878D	ALPHA * 1953
37		AR	SOLDER SN60	QQ-S-571	TYPE RA

4-1682

[illegible]



4-17-1

ADDRES: PANTEL CRISTINA, C/ RIV. DE GUAL, LT. 2, ENCEPES, PL. MEN. 48.300-6

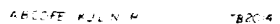
3. THINNING LOCATION, PART OF AREA AND ARE TO BE APPLIED TO THE AREA OF THE THINNING POSITION. NONE

4. AFTER G.O.A. CONTROL HAS TESTED AND ADVERTISED THE PRINTED CIRCUIT ASSEMBLY, APPLY A CONFORMAL COAT AS  
TO ALL SURFACES OF BOARD. EXCEPT TERMINALS TO INK. TO BE KEPT CLEAR. COAT IS MUST COMPLY  
WITH MIL-PRC-13000.

5. VEREKENINGEN 2.3.3

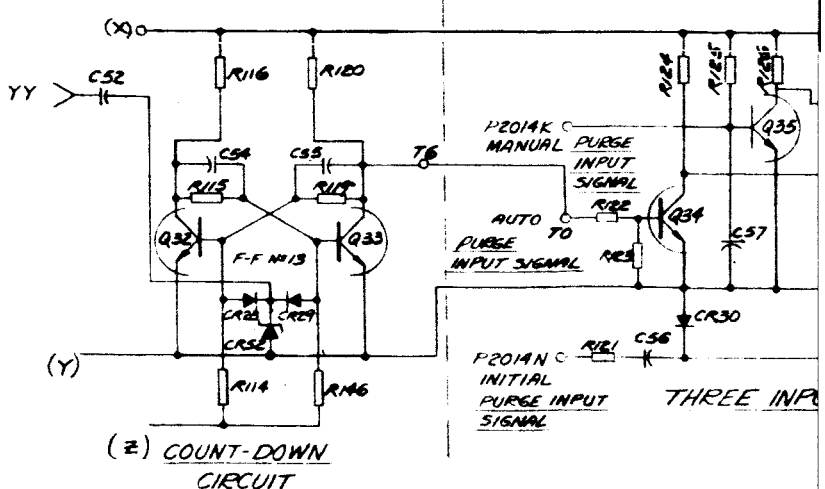
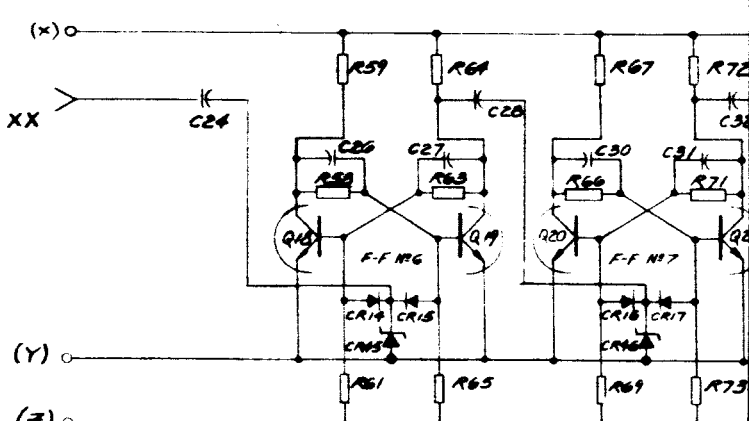
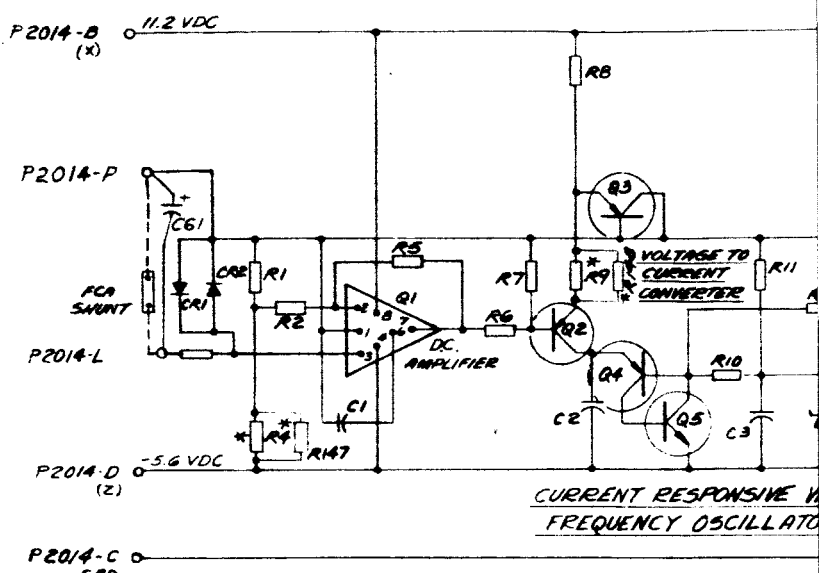
PRINTED CIRCUIT BOARD 43-400-140 107  
PRINTED CIRCUIT MASTER 43-400-141 131  
ELECTRICAL SCHEMATIC 43-400-143 401

SUBSTITUTION REACTION OF ALKYL HALIDES WITH NUCLEOPHILIC

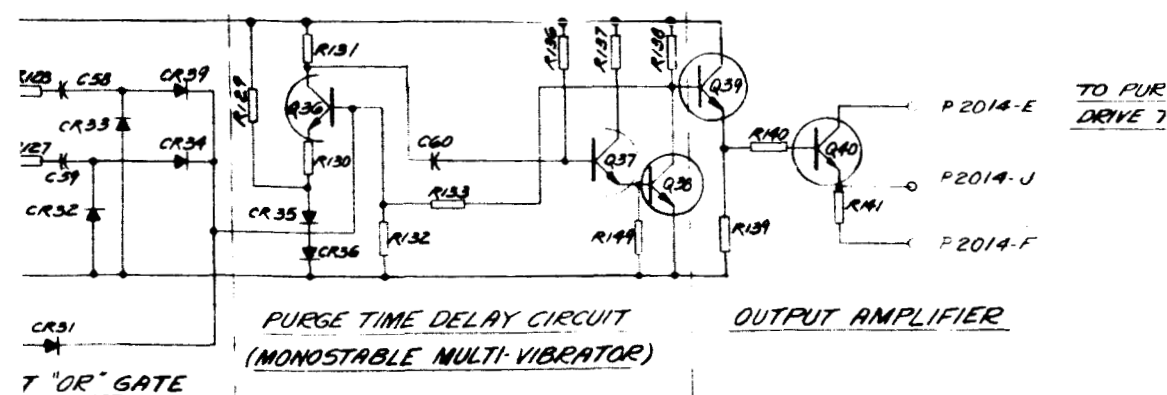
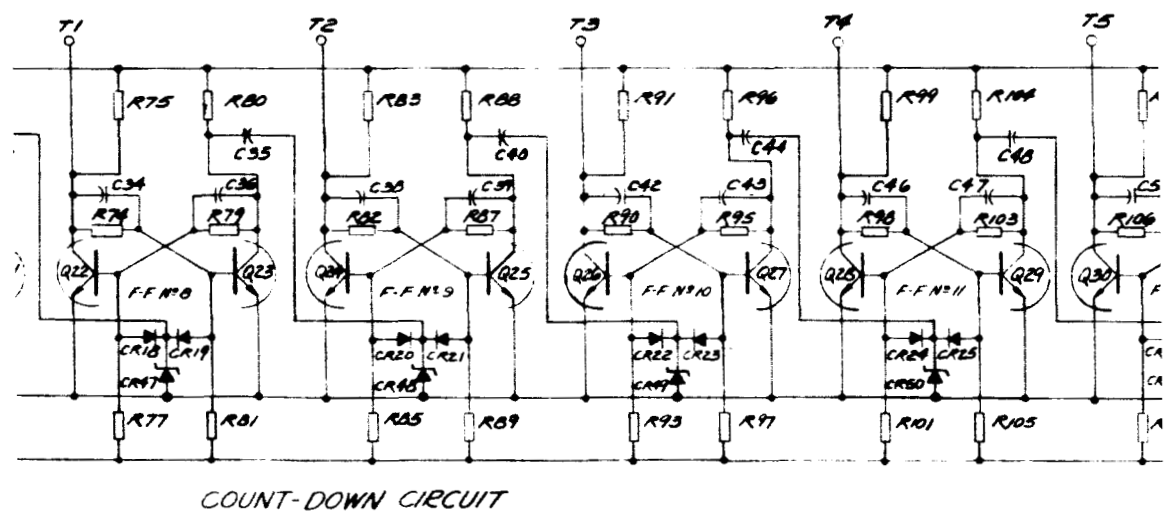
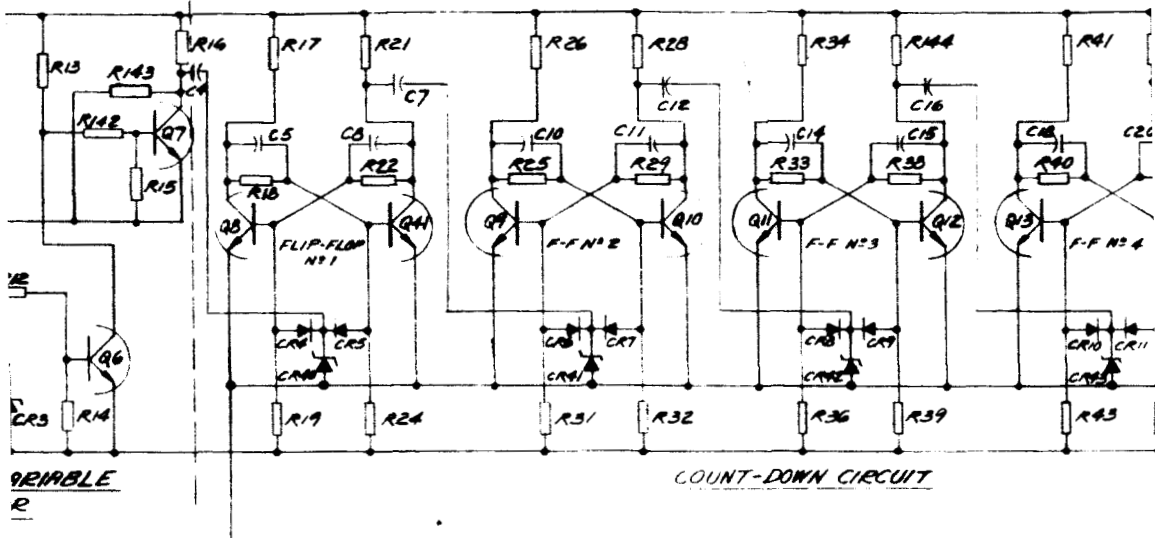
[illegible]

447-2

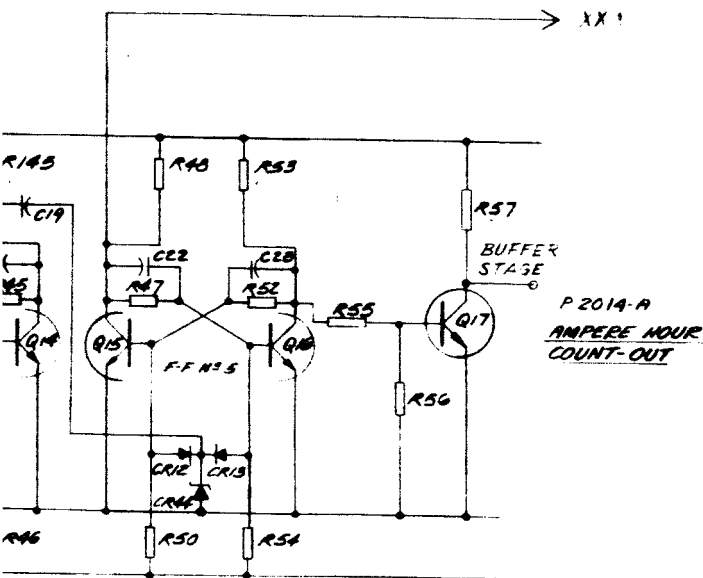
### CURRENT SENSING CIRCUIT



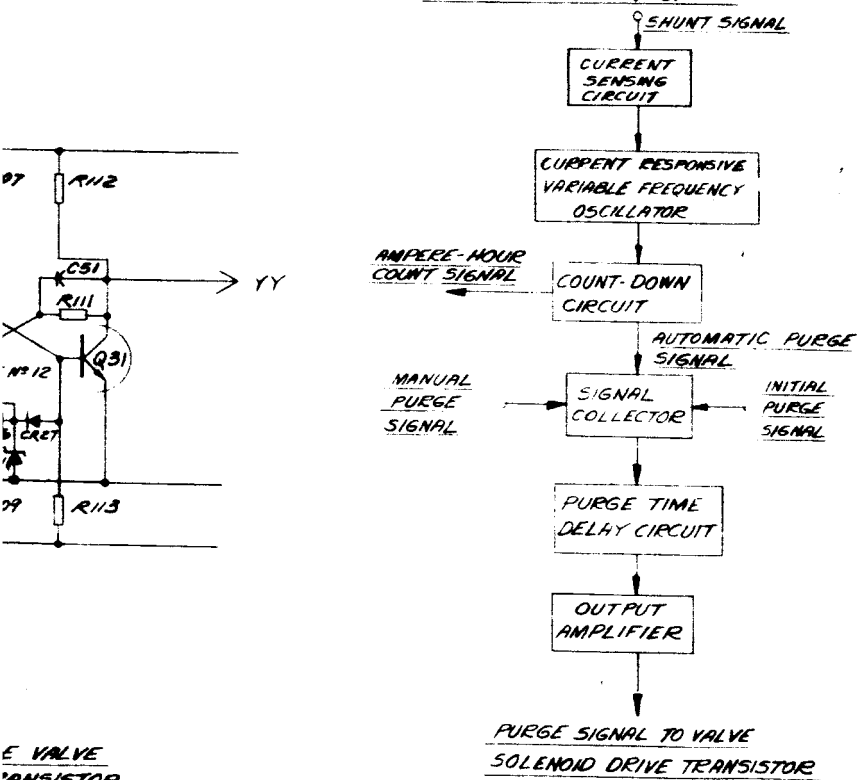
4-18-1



4-18-2



### SYSTEM BLOCK DIAGRAM



E VALVE  
TRANSISTOR

FOR BILL OF MATERIAL SEE  
DWG 49-200-465-501

CONFIDENTIAL PROPERTY OF ALLIS-CHALMERS MFG. CO. RSCH. #3341 M.I. WORKS		NAME <b>MODIFICATION #10 PURGE CONTROLLER SCHEMATIC</b>	
UNLESS OTHERWISE SPECIFIED		MATERIAL	
1- PLACE DEC #	<input checked="" type="checkbox"/> MACHINED SURFACE TEXTURE	<b>FIGURE 4-18</b>	
2- PLACE DEC #			
3- PLACE DEC #	SIMILAR TO	WT.	
ANGULAR	DR 12-22-63	PART NO.	
SCALE	SCALE	49-400-291-401	
SHEET	SHEET	PRINTED IN U.S.A. FORM NO. 81231	

49-400-291-401 03

Normal Operation - Purge cycling during normal operation is controlled by the ampere-hours (AH) of current delivered by the fuel cell. The load current signal from the fuel cell shunt is processed in a differential amplifier which provides a linear voltage to a relaxation oscillator and pulse circuit whose output pulse rate is directly proportional to the fuel cell output current. However, this pulse rate is too high to be of practical value. Therefore, this output is processed through a binary counter string which consists of a series of flip-flops until a predetermined output rate is obtained. The pulse circuit is normally adjusted for approximately four pulses per AH at the input to the binary counter string. Outputs obtained from successive flip-flops are two pulses per AH, one pulse per AH, one pulse per 2 AH, etc. until one pulse per 256 AH is obtained at the end. The purging rate is preset by selecting the appropriate tap of the binary counter string. This selected output is fed to an OR gate which feeds the delay monostable multivibrator energizing the purge valves through the valve driver. The duration of the purge is dependent upon the circuit parameters of the monostable multivibrator.

Manual Operation - If desired, a manual purge can be initiated by depressing the purge push button which supplies a trigger pulse to the OR gate circuit.

Automatic Purge at "APPLY LOAD" - When an "apply load" signal is applied to the EMCS, an initial purge signal is sent from the master controller to the OR gate for accomplishing an initial purge whenever load is first applied to the fuel cell.

- 4.5.4 Moisture Removal Controller - See Figure 4-19A & 19B (Drawing No. 49-200-454-401), Figure 4-20 (Drawing No. 49-300-545-401), Figure 4-21 (Drawing No. 49-200-421-401).

The moisture removal controller maintains the desired KOH concentration in the fuel cell stack by removing the excess by-product water produced by the reaction. Because the percentage of KOH concentration is a function of the moisture removal cavity pressure and the stack temperature, these two parameters are used as the controlling functions.



# CIRCUIT COMPONENT LIST FOR DRAWING 49-300-585-801 P1X

IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
1	R20, R29 R21, R24	4	Resistor Fixed Film $\pm 5\%$ 0.25 W	MIL-R-22684	
2	R3	1	Resistor Fixed Film 2K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD202J
3	R4, R12, R13, R16	4	Resistor Fixed Film 2.2K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD222J
4	R5	1	Resistor Fixed Film 100K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD104J
5	R6, R1	2	Resistor Fixed Film 8.2K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD822J
6	R7	1	Resistor Fixed Film 6.2K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD622J
7	R8	1	Resistor Fixed Film 220 $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD221J
8	R9	1	Resistor Fixed Film 1.8K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD182J
9	R10, R19	2	Resistor Fixed Film 12K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD123J
10	R11	1	Resistor Fixed Film 5.6K $\pm 5\%$ 0.25 W	MIL-R-22684	RL07AD562J

4-19-A-51

11	R15, R17	2	Resistor Fixed Film 5.1K + 5% 0.25 W	MIL-R-22684	RL07AD512J
12	R18	1	Resistor CARBON 330K + 5% 0.25 W	MIL-R-11/8	RC07GF334J
13	R20	1	Resistor Fixed Film 22K + 5% 0.25 W	MIL-R-22684	RL07AD223J
14	R22, R25, R26	3	Resistor Fixed Film 10K + 5% 0.25 W	MIL-R-22684	RL07AD103J
15	R23	1	Resistor Fixed Film 470 + 5% 0.25 W	MIL-R-22684	RL07AD471J
16	R27	1	Resistor Fixed Film 820 + 5% 0.25 W	MIL-R-22684	RL07AD821J
17	AR	CONDUCTOR, SOLID AWG 28	QQ-W-343	BELDEN 8022	
18	C1	1	Capacitor, Tantalum Polar 47mf	MIL-C-26655B	CS13BF476K
19	C2	1	Capacitor, Ceramic 1000 pf	MIL-C-11015	CK05CW102K
20	CR1	1	Diode, Zener 33 V	IN3032	
21	CR2	2	Diode, Zener 3.6 V	MSFC-336/B	SIN747A
22	CR3	1	Diode, Zener 6.8 V	MSFC-336/B	SIN754A
23	Q1	1	Operational Amplifier	Fairchild	uA-702
24	Q2, Q3, Q5, Q7	4	Transistor	MSFC-338/105	S2N697
25	Q4	1	Transistor	MSFC-338/108	S2N1132
26		1	CONNECTOR	WINCHESTER	SMCE-20-PGH



# CIRCUIT COMPONENT LIST FOR DRAWING 49-300-545-401P1X

IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
37	R2	1	RESISTOR FIXED FILM 1K $\pm 5\%$ 0.25W	MIL-R-22684	R207AD102J
38		AR	LAGGING TAPE, BLACK	MIL-T-713A	ALPHA #20134

4-19B-1

4-19-B-2

ADDED SHT 2 OF 2  
Q3 5-13-66 RDM  
RAC. REC. 69 065 N/2

UNLESS OTHERWISE NOTED DIMENSIONS ARE IN INCHES & MACHINING TOLERANCES ARE				ANGLES ±	
DIMENSIONS	UP TO 6 INCL	OVER 6 TO 24 INCL	OVER 24		
FRACTIONAL	±	±	±		
DECIMAL	±	±	±		

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ALLIS-CHALMERS MFG. CO., Milwaukee, Wis.  
RESEARCH - 3341

DRN RAR 12-2-65  
TRD

CHD FJ.M. 4-27-66  
APPROL 3-5-66

BILL OF MATERIAL

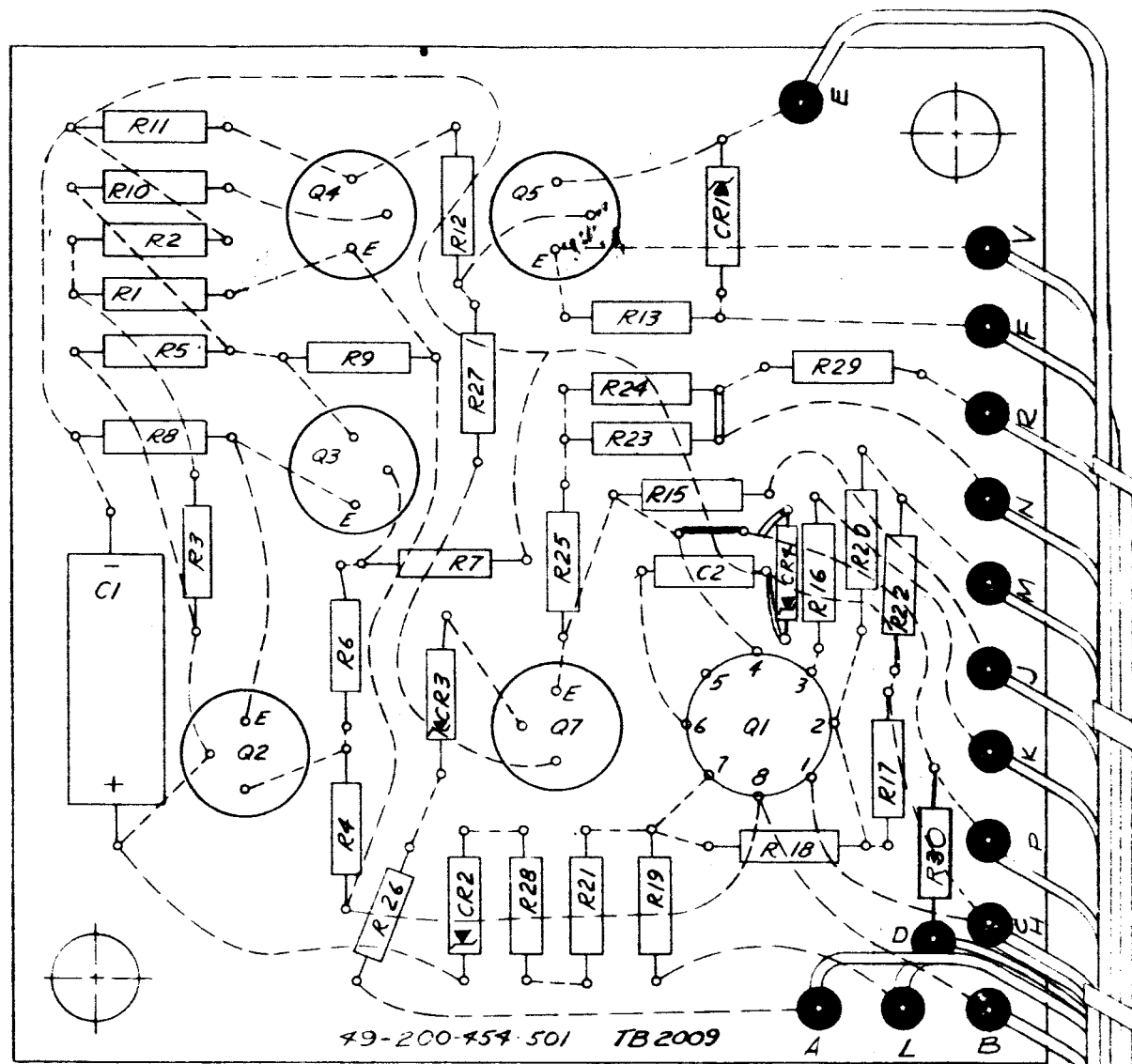
H<sub>2</sub>O CAVITY PRESSURE  
CONTROL MOD. 10

FIGURE 4-19B

SIMILAR TO SCALE 1:12

SHEET 2  
OF 2

49-200-454-501



P20

KEY PIN

4-20-1

01 7-27-66

REL. 20, 0166

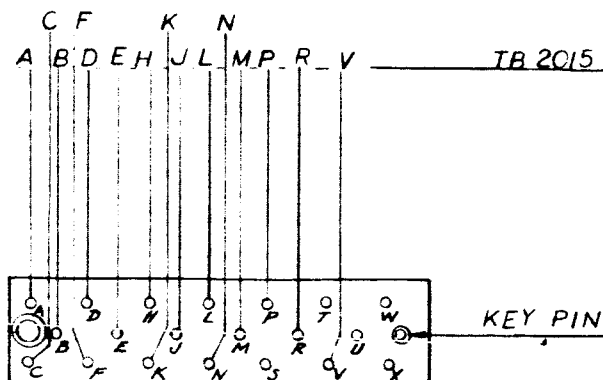
NOTES:

1. PRINTED CIRCUIT ASSEMBLY TO BE CONSTRUCTED AND  
INSPE 1FL PER NPC-200-4
2. ALL TERMINALS SHALL BE MOUNTED ON THE PRINTED  
CIRCUIT BOARD PRIOR TO INSERTING ELECTRICAL COMPONENTS
3. TERMINAL IDENTIFICATION, PART & PLUG NUMBERS ARE TO BE APPLIED  
IN BLACK INK .12 INCH HIGH IN POSITION SHOWN
4. COAT ALL TERMINAL IDENTIFICATION, PART & PLUG NUMBERS  
WITH CLEAR LACQUER
5. AFTER QUALITY CONTROL HAS TESTED AND ACCEPTED THE  
PRINTED CIRCUIT ASSEMBLY APPLY A CONFORMAL  
COATING TO BOTH SIDES OF BOARD. COATING MUST  
COMPLY WITH MSFC-PROC-257
6. REFERENCE DWG.

PRINTED CIRCUIT BOARD 49-300-546-001  
PRINTED CIRCUIT MASTER 49-300-546-091  
ELECTRICAL SCHEMATIC 49-200-421-401

7. WIRING KEY

- ABOVE BOARD JUMPER  
 — BELOW BOARD JUMPER  
 --- PRINTED CIRCUIT PATH



PLUG PIN DIAGRAM  
FOR P2009

CONFIDENTIAL - PROPERTY OF ALLIS-CHALMERS MFG. CO. 3341 MI WORKS		NAME PRINTED CIRCUIT ASSEMBLY H20 CAVITY PRESSURE CONTROL II	
UNLESS OTHERWISE SPECIFIED:		MATERIAL	
1. PLACE DEC *		<div style="text-align: center;">  MACHINED SURFACE TEXTURE         </div>	
2. PLACE DEC * .01			
3. PLACE DEC *			
ANGULAR *		FIGURE 4-20	
SIMILAR TO		WT	
DR. 4566		PART NO.	
CH. 7/5/66		49-300-545-401	
AP		SCALE 3:1 SHEET 1- FINAL	

49-200-454-501  
 NEXT ASSEMBLY USED ON

FIGURE 4-20

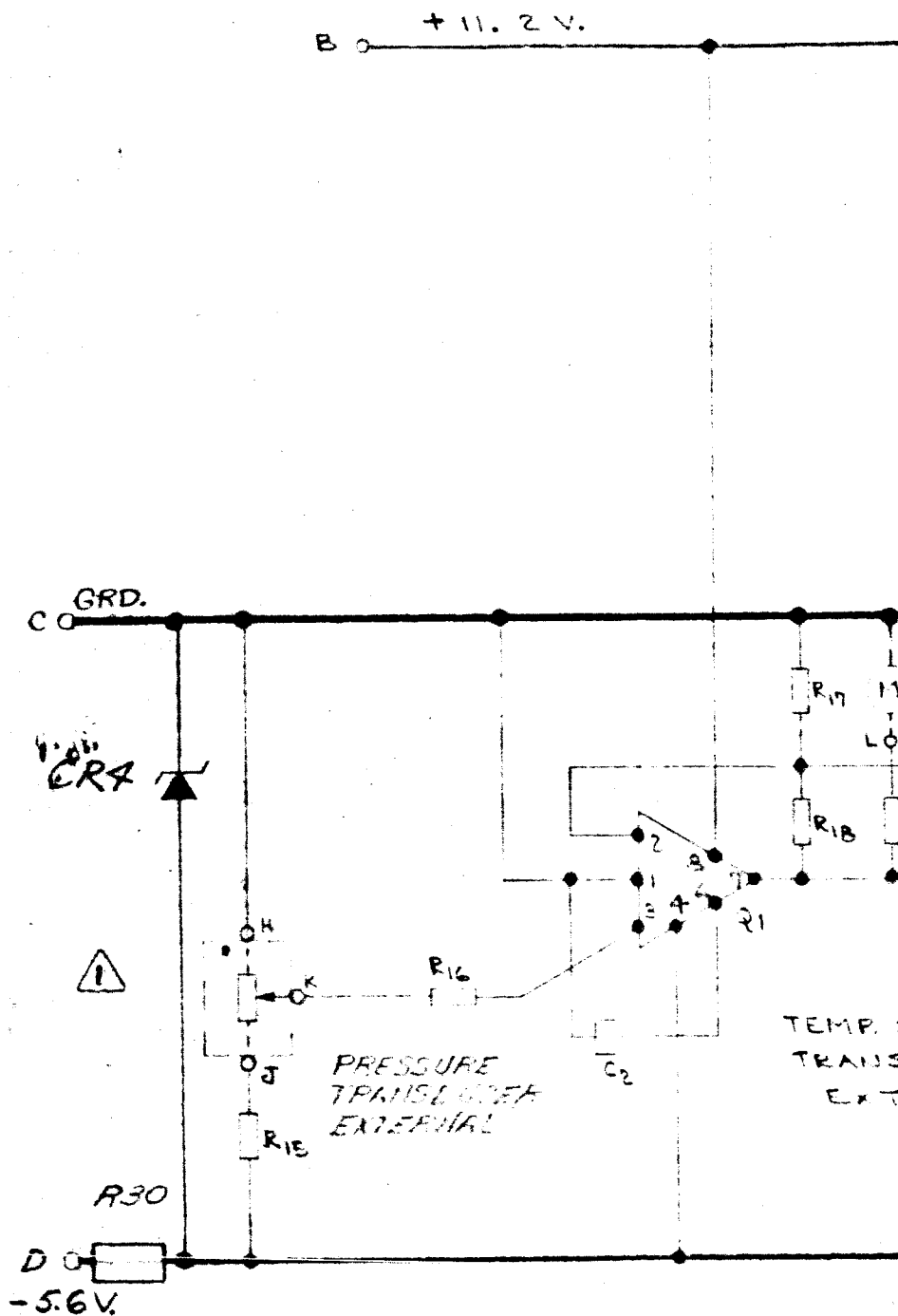
PART NO. 49-300-545-401

2 1054-UDC

PRINTED IN U.S.A.

FORM No. 8122-1

49-300-545-401



\* NOTE:-

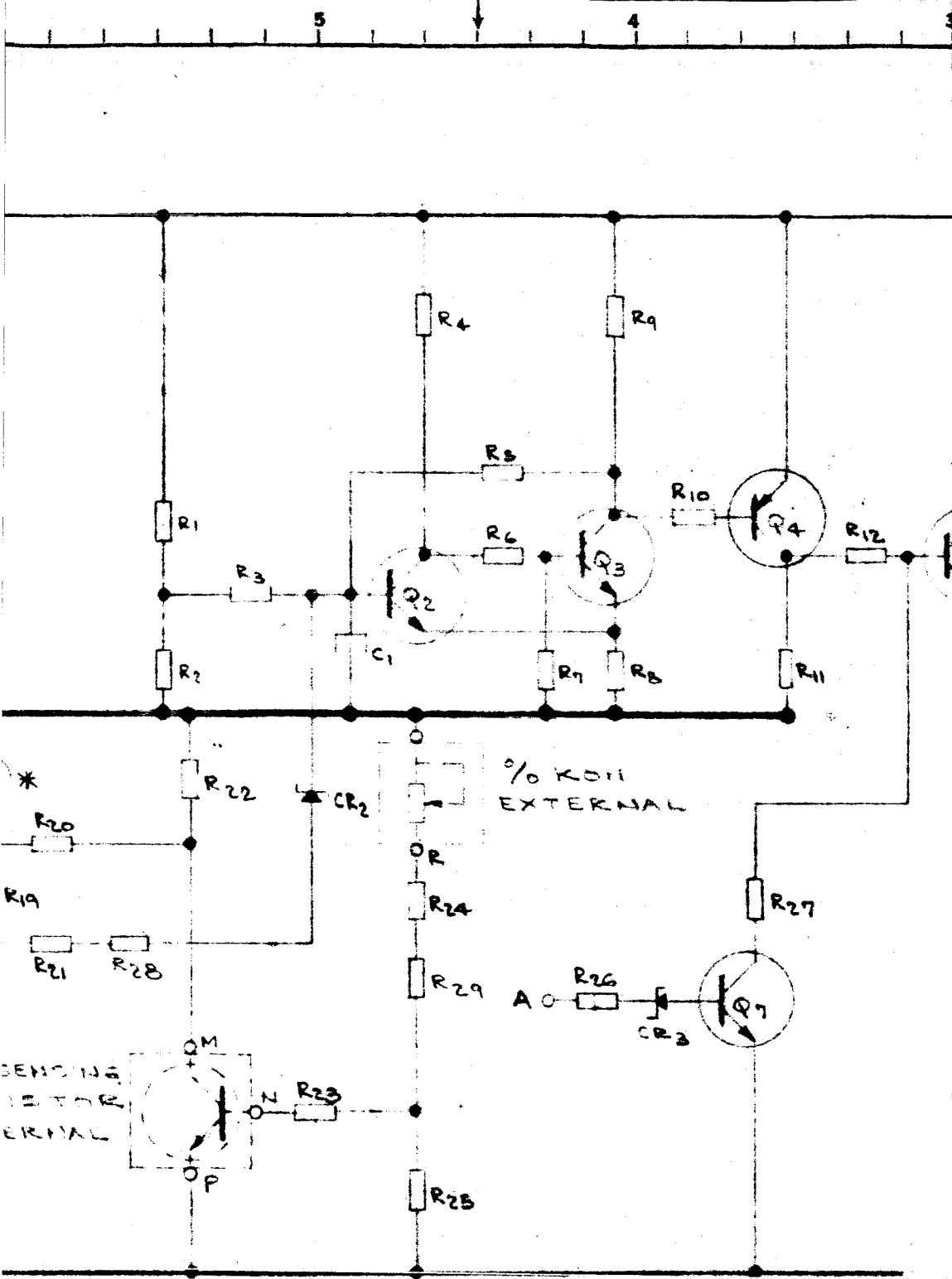
EITHER AN EXTERNAL 500-O.-S.  
METER OR A JUMPER TO BE  
BETWEEN TERMINAL 'L' & C

CONNECTOR REFERENCE DESIGNA

4-21-1

BUL





100 WAMP  
INSTALLED  
ROUND.

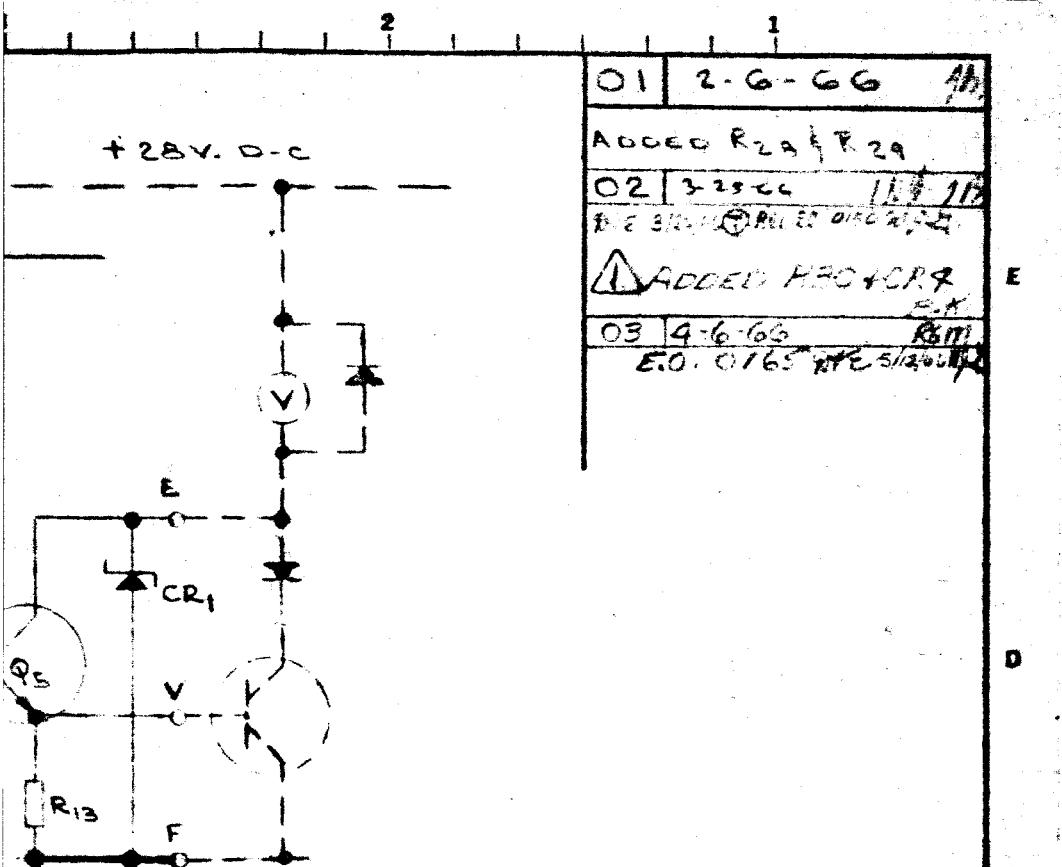
TION J2009

49-200-421-401

03

4-21-2

COMP	ALL
RESEA	32
TYPE	1-PLAC
	2-PLAC
	3-PLAC
	ANGUL
DR	RAT
ON	Q
AT	PA



IDENTICAL - PROPERTY OF <b>CHALMERS MFG. CO.</b> RCH - W.A. WORKS AS OTHERWISE SPECIFIED:		NAME <b>H<sub>2</sub>O CAVITY PRESSURE CONTROL MOD. 10</b>	
<div style="display: flex; align-items: center;"> <div style="border: 1px solid black; padding: 2px; margin-right: 10px;">           DEC ± DEC ± DEC ± AN ±         </div> <div style="text-align: center;">   <b>MACHINED SURFACE TEXTURE</b> </div> </div>		MATL	
10-11-65 <i>2/6/66</i> <i>2/5/66</i>		<div style="display: flex; justify-content: space-between;"> <span><b>FIGURE 4-21-3</b></span> <span>WT <sup>R</sup> F</span> </div>	
SIMILAR TO		PART NO <b>49-200-421-401</b>	
SCALE <i>1/2" = 1"</i>		SHEET <b>OF 1</b>	

Stack temperature is sensed by a transistor embedded in the fuel cell stack. The temperature-emitter current characteristic curve of this transistor closely matches the KOH concentration temperature - vapor pressure characteristic curve. The output of this transistor provides a reference voltage signal of the stack temperature. The moisture removal cavity pressure is sensed by a variable resistance type transducer mounted ahead of the moisture removal solenoid valve. The transducer provides a voltage signal proportional to the actual moisture removal cavity pressure.

These two signal voltages are applied to a differential amplifier. If a negative output signal is present, the pull-in coil of the moisture removal solenoid valve relay will be energized through the action of the control amplifier. The magnitude at which the valve operates is determined by the desired allowable pressure error band. A transistor interlock is provided to prevent actuating the valve unless the load has been applied.

- 4.5.5 Master Controller - See Fig. 4-22 A and 4-22 B (Dwg. No. 49-200-456-501), Fig. 4-23 (Dwg. No. 49-500-184-401), Fig. 4-24 (Dwg. No. 49-300-474-401). The master controller distributes the input control signals to the proper circuits and functional devices, and integrates protective controls into the fuel cell systems. The master controller operates as follows:

Under normal startup conditions, the start AND gate must have present -5.6 VDC, +11.2 VDC, +28 VDC, and the start command. Having satisfied these conditions, an "OK signal to heat" is sent to the temperature controller which turns ON the stand-by heaters and commands the warm-up heaters to be turned ON.

## CIRCUIT COMPONENT LIST FOR DRAWING

49-590-184-491

P1X

IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
1	R 26, 36, 67, 79	4	Resistor Fixed Film 1K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 102J
2	R 5 & 8	2	Resistor Fixed Film 1.5K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 152J
3	R 13, 23, 31	8	Resistor Fixed Film 1.8K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 182J
	13, 59, 66, 74, 85				
4	R 16, 18, & 93	3	Resistor Fixed Film 2.0K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 202J
5	R 1, 12, 25, 31, 35	14	Resistor Fixed Film 2.2K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 222J
	38, 46, 47, 55, 58				
6	R 11, & 94	2	Resistor Fixed Film 3.3K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 332J
7	R 33, 45, 76, 97	4	Resistor Fixed Film 5.6K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 562J
8	R 29, 41, 72, 83	4	Resistor Fixed Film 6.2K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 622J
9	R 6, 2, 23, 40	7	Resistor Fixed Film 8.2K $\pm$ 5% 0.25W	MIL-R-22684	RLQ7AD 822J

4-22A-1

10	50, 71, 82	11	Resistor Fixed Film 10K $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 103J
	R 2, 3, 17, 19, 20, 48, 49, 52, 56, 57				
	62				
11	R 14, 22, 32, 44				
	60, 64, 75, 86	8	Resistor Fixed Film 12K $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 123J
12	R 24, 37, 51, 68, 78, 21	6	Resistor Fixed Film 22K $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 223J
13		1	Printed CIRCUIT Board		49-400-342-001
14	R 10, 27, 39, 53, 70, 81, 15, 61	8	Resistor Fixed Film 100K $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 104J
15		2	Transistor Pad	MILTON ROSS	10229
16	R 30, 42, 73, 84	4	Resistor Fixed Film 220 $\Omega$ $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 221J
17	R 4, 8, 7	2	Resistor Fixed Film 120 $\Omega$ $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 121J
18	R 85	1	Resistor Fixed Film 56K $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 563J
19	R 89	1	Resistor Fixed Film 5.1K $\pm$ 5% 0.25W	MIL-R-22684	RLO7AD 512J
20	C 1, 3, 4, 10, 11, 13	6	Capacitor Tantalum 0.33 mf	MIL-C-26655B	CS13BF334K
21	C 7, 9, 16, 18	4	Capacitor Tantalum 10 mf	MIL-C-26655B	CS13BF106K
22	C 6, 8, 15, 17	4	Capacitor Tantalum 68 mf	MIL-C-26655B	CS13BF686K

23	C 2, 5, 12, 14	4	Capacitor Ceramic 220 pf	MIL-C-11015	CK05CW221K
24	Q 1 & 14	2	Operational Amplifier	Fairchild	UA702A
25	Q 2, 3, 4, 5, 6, 7	20	Transistor	MSFC 338/105	S2N697
	9, 10, 11, 13, 15, 16				
	17, 18, 19, 20, 22				
	23, 24, 26				
26	Q 8, 12, 21, 25	4	Transistor	MSFC 338/108	S2N1132
27	CR 1, 2, 3, 5, 10	24	Diode	MSFC-338/3	SIN626
	11, 12, 13, 14, 17				

ASSY. USED ON: 49-200-585-501 & 502

UNLESS OTHERWISE NOTED DIMENSIONS ARE IN INCHES & MACHINING TOLERANCES ARE:			ANGLES ±
DIMENSIONS	UP TO 6 INCL.	OVER 6 TO 24 INCL.	OVER 24
FRACTIONAL	±	±	±
DECIMAL	±	±	±

- Property of

ALLIS-CHALMERS MFG. CO., Milwaukee, Wis.

BILL OF MATERIAL  
MASTER CONTROL  
MOD 10

FIGURE 4-22A  
3

SIMILAR TO: SCALE = 12

SHEET 1  
OF 2  
49-200-456-501

DRN JMT 12-9-65  
TRD  
CH'D F.T.N. 7/20/66  
APP'D P.D.B. 5/6/66  
DEPT.

RETYPE P/L  
PIX OWN. N. WAS  
49-300-474 & 401  
TYPE-ADDED R21  
WAS (5) REQ'D.  
IT IS BE ADDED  
IT IS DELETED R21  
WAS (2) REQ'D.  
IT 201 WAS IT 17  
" 21 " " 18  
" 22 " " 19  
" 23 " " 20  
" 24 " " 21  
" 25 " " 22  
" 26 " " 23  
" 27 " " 24  
0414-20-66  
E.O. 0127  
18. ME

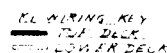
# CIRCUIT COMPONENT LIST FOR DRAWING 79-500-187-701

IT. NO.	REFERENCE DESIGNATION	NO. REQ.	DESCRIPTION	MIL SPEC. OR MFG. NAME AND CODE	MIL TYPE OR MFG. PART NO.
27	18, 19, 20, 23, 24				
28	26, 27, 29, 30, 37				
29	38, 39, 40, 41				
28	CR 28, 31, 33	3	Diode		IN4003
29	CR 4, & 6	2	Zener Diode	MSFC-338/8	SIN751
30	CR 8	1	Zener Diode	MSFC-338/11	SIN964B
31	CR 9	1	Zener Diode	MSFC-338/8	SIN758A
32	CR 15, 16, 35, 36	4	Zener Diode	MSFC-338/8	SIN747A
33	CR 21, 22, 32, 34	4	Zener Diode		IN3034B
34	CR 25, & 7	2	Zener Diode	MSFC-338/11	SIN968B
35	K1	1	Relay Latching 24VDC	Porter & Bramfield	SL11DB
36	C19	1	Capacitor Tantalum .47 mf	MIL-C-26455B	CS13RF476K

37	P2015 P2021	2	Connector		Winchester	SMRE-20-PCGH
38	C20 & C21	2	Capacitor, Ceramic 2200 PF		MIL-C-11015	CK05CW222K
39		24	Rad. Transistor MOUNTING		TRANSIPAD PRODUCTS	10012 DAT. BLACK
40		AR	Conductor, Solid AWG 24		QQ-W-343	BELDEN 8022
41		AR	Shrink Tubing, RNF-100.		276-UL T 0D05 Dia.	
42		24	Terminal Turret		USECO	2030B
43		AR	Solder SN60		QQ-S-571	Type RA
44		AR	Flux		MIL-F-14256	Type A
45		AR	WIRE AWG #26 TYPE B		MIL-W-16870/1	49-150-009.004
Ref.		-	Electrical Schematic			49-100-474-401
46		4	SCREW #2 - 5 1/2" LG.		MS 51957	MS 51957-14
47		4	WASHER, PLAIN		MS 15795	MS 15795-804
48		4	NUT, SELF-LOCKING, HEX.		NAS 1021	NAS 1021004
49		AR	LACING TAPE, BLACK		MIL-T-713A	ALPHA #CC134







4-23-

—

1. DATE: 11/15/2011

ALL INFORMATION CONTAINED HEREIN IS UNCLASSIFIED EXCEPT WHERE SHOWN OTHERWISE

[illegible]

WHAT WE MUST COMPLY WITH THE PROSECUTOR'S REQUESTS.

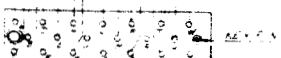
EXPENSES 2008  
PRINTED CIRCUIT BOARD 43-402-382-00  
PRINTED CIRCUIT BOARD 43-402-382-00  
ELECTRONIC BOARD 43-402-382-00

SUBSTRATE: FARMHOUSE, 100 YARDS EAST OF ROAD. LUMBER WITH CLEAR LAGGER

**VOLUME 1**

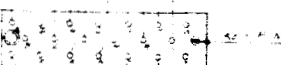
--- ABOVE BOARD JUMPER  
--- BELOW BOARD JUMPER

$\begin{matrix} & C & & K & & S \\ L & B & C & F & = & L & M & D, & : & J & N & \dots & 2 & 2 & 2 & 5 \end{matrix}$



PLUG IN CABLE  
FOR P2015

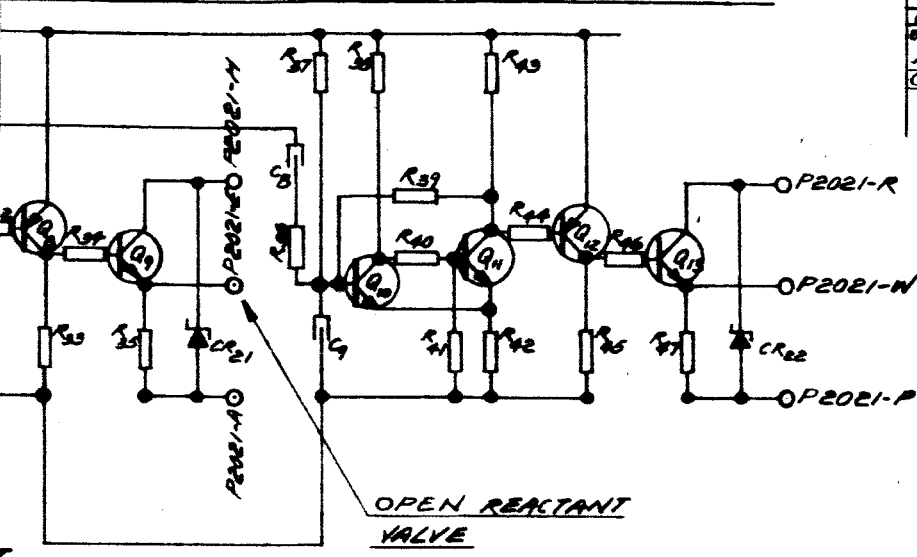
5. 574 A. P. P. A. 1822

[illegible]

PROPERTY OF <b>ALLIS-CHAMBERLAIN MFG. CO.</b>		60 20-65251	
3394 N. 10100		NEXT TO EMER. USEL ON	
1. PLACE BELL 2. PLACE BELL 3. PLACE BELL 4. PRESS <input checked="" type="checkbox"/>		PRINTED CIRCUIT BOARD TO THE CONTROLLER	
MAINTENANCE SURFACE TO BE REFINISHED		FIGURE 4-23	
ON 20-65251 COPY 11 3 1/2 10100		SCALE 1:1 DATE 4-500-184-401	



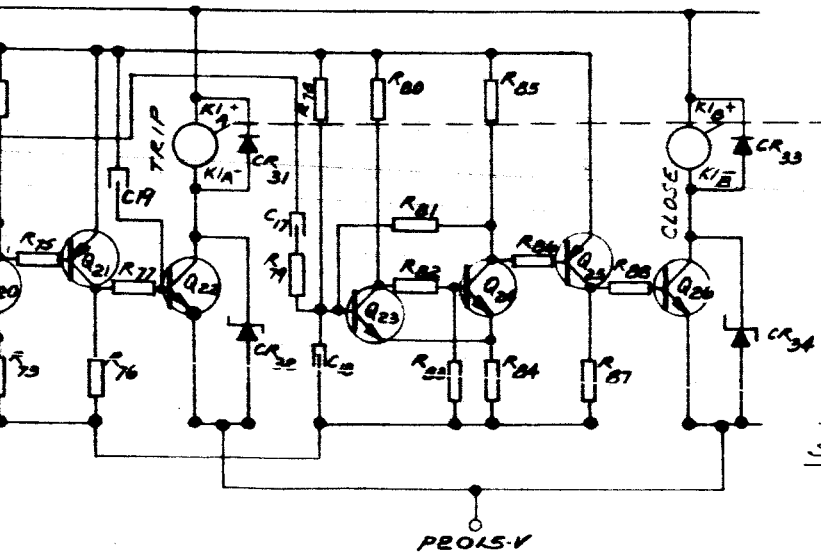
01 2-7-66 JH.  
 DWG. BROUGHT UP TO DATE  
 02 3-24-66 JH. JH.  
 THE 3RD REV. REL. ED. 0-20-66  
 RELUC. C-19 ADD. CROF  
 CEI JMT  
 03 4-24-66 JH. JH.  
 DWG. + SLD (C) REL. ED. 0-20-66  
 RELAY K1 POLARITY INDICATED.  
 04 4-24-66  
 E.O. 0787



CLOSE REACTANT VALVE

OPEN REACTANT VALVE

WATER CONTROL



CLOSE REACTANT VALVE  
 OPEN REACTANT VALVE

FOR BILL OF MAT'L.  
 SEE DWG. 49-200-456-501

WATER CONTROL

PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b> 3341 M.I. WORKS		NAME <u>MASTER CONTROLLER II</u> <u>ELECTRICAL SCHEMATIC</u>	
UNLESS OTHERWISE SPECIFIED:		MATERIAL	
1. PLACE DEC ±	MACHINED SURFACE TEXTURE	FIGURE 4-24	
2. PLACE DEC ±			
3. PLACE DEC ±			
ANGULAR ±	SIMILAR TO	WT	
DRW JMT 12/3/65	SCALE	PART NO	
CHK JMT 1/10/66	N.A.	49-300-474-401	
AP JMT 4/4/66	SHEET /	FORM No. 8127-1	
	001		

4-24-2

To apply an external load to the fuel cell, an apply load signal is applied to the load flip-flop providing the start AND gate has been satisfied. This will actuate the load relay, send a signal to the purge controller initiating a purge cycle, and complete the interlock requirements for the water removal valve (valve cannot be opened unless load has been applied).

The reactant inlet valve is opened manually through the reactant inlet valve driver.

To shut down the fuel cell, a stop signal is applied to stop OR gate. This activates the stop flip-flop which in turn triggers the valve driver to close the reactant valve. At the same time, the same flip-flop triggers the "Main Circuit Breaker OR gate" which trips the "Load Trip Relay" through a flip-flop. Also the stop OR gate deactivates the water cavity pressure controller which closes the moisture removal valve, and interrupts the "OK to heat signal" which stops all heating until the start command is initiated again. The liquid coolant valve will automatically close dependent on temperature. Other inputs such as: High reactant pressure, over-temperature, supply voltage and emergency stop will shut down the fuel cell in an identical manner.

"Trip main circuit command" will trip the load and close the moisture removal valve. Fuel cell voltage less than 21 VDC will trip the load and close the moisture removal valve.

#### 4.6 CANISTER AND SUPPORT SUBSYSTEM

The canister houses the Fuel Cell Stack and Thermal Conditioning and Control Subsystems. Acting as a pressure vessel, it contains and manifolds the helium gas coolant. Brackets bolted to the canister provide a means of rigidly mounting the assembly.

##### 4.6.1 Canister

The canister, composed of a cylindrical flanged shell and end bell, contains the gas coolant at a pressure of  $40 \pm 2$  psia and is designed for a burst pressure of 150 psia and a proof pressure of 100 psia. The canister is fabricated from a magnesium alloy. The cylindrical shell walls are 1/16 inch thick, whereas the spun end bell walls are 3/32 inches thick. Sealing between the cylindrical shell flanges, the end bell, and the fuel cell stack bottom end plate is accomplished with elastomer "O" rings.

Provisions are made on the cylindrical portion of the end bell for entry and return of the liquid coolant as well as entry of electrical blower leads and thermal instrumentation. The bottom end plate has a port provision for the initial charging or depressurization of the gas coolant in the canister. The canister pressure is continually monitored through this port by a 0-50 psia absolute - pressure transducer. A 45 psia relief valve is provided for over pressure protection.

##### 4.6.2 Support

A rigid support is provided by four brackets that mechanically fasten to the flanges of the cylindrical canister shell as shown in Figure 4-25. The

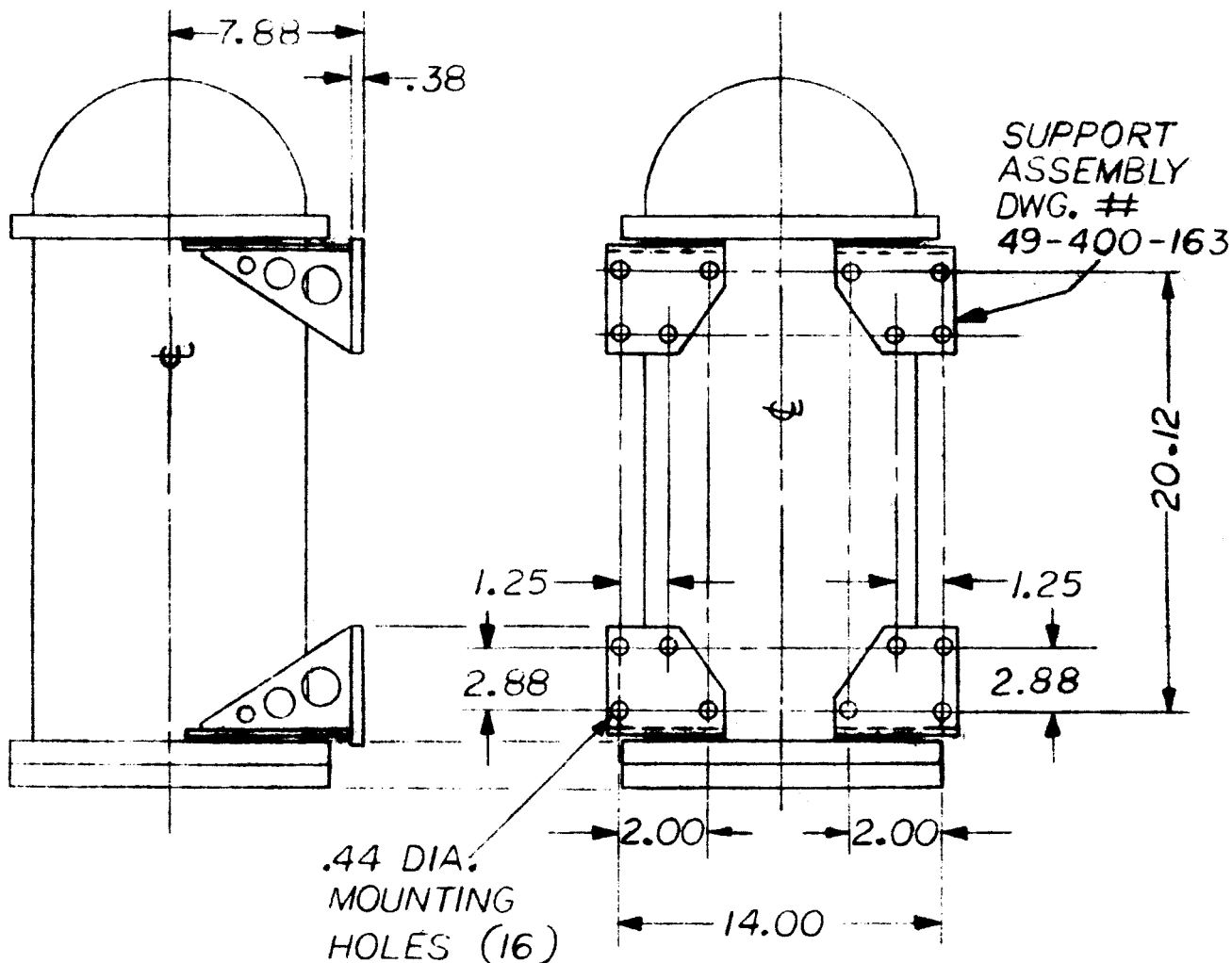


FIGURE 4-25

CONFIDENTIAL — PROPERTY OF <b>ALLIS-CHALMERS MFG. CO.</b> 3341 W.A. WORKS		NAME	
UNLESS OTHERWISE SPECIFIED:		MATL	
1-PLACE DEC ± 2-PLACE DEC ± 3-PLACE DEC ± ANGULAR ±	<input checked="" type="checkbox"/> MACHINED SURFACE TEXTURE	F.C.A. BULKHEAD MOUNTING DETAIL	
DR <u>JE</u> 8-10-65	SIMILAR TO	WT R F	
CH	SCALE	PART NO	BUL
AP	NONE	49-100-596	01



brackets attach to the canister through thermal insulating mounts to minimize losses due to thermal conduction.

The Reactant Control and Conditioning Subsystem is also attached to the canister through thermal insulation mounts, thereby isolating it from the canister, except for the tube connections.

#### 4.7 TEST INSTRUMENTATION

##### 4.7.1 Thermocouple Instrumentation

There are 12 iron-constantan thermocouples located in and outside of the Fuel Cell to permit detailed temperature analysis. (Figure 1-2A and 1-2B (Drawing No. 49-500-125-404), Fuel Cell System Interface Schematic, shows the location of the thermocouples. Figure 2-9, Sheet 5 of 5 (NASA Interface Connections) gives the location and function of each thermocouple.